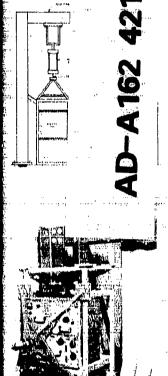
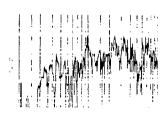
US Army Corps of Engineers





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**TECHNICAL REPORT SL-85 6** 



# FORCE-PULSE TESTS OF TACTICAL COMMUNICATION EQUIPMENT

by

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Mary Electronics R&D Command Adelphia, Maryland 20733

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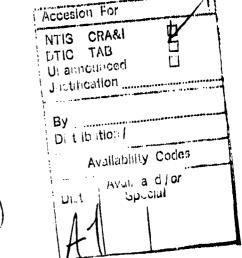
#### **PREFACE**

This data report covers developmental work on the force-pulse generator and all pulse tests on the communication equipment through December 1980. Calibration test data have been previously reported. Overall program management for the sponsors was provided by Dr. William Schuman, U. S. Army Ballistic Research Laboratory, and Mr. Robert Freiberg, U. S. Army Electronics Research and Development Command. Project Engineers were Dr. Frederick Safford, Agbabian Associates, and Mr. Roger D. Crowson, Structural Mechanics Division (SMD), Structures Laboratory (SL), U. S. Army Engineer Waterways Experiment Station (WES). Instrumentation support was provided and data processing was performed by Messrs. James L. Pickens and Cary B. Cox, Instrumentation Services Division, WES.

Commanders and Directors of WES during the work and preparation of this report were COL Nelson P. Conover, CE, COL Tilford C. Creel, CE, and COL Robert C. Lee, CE; Mr. F. R. Brown was Technical Director, Mr. Bryant Mather was Chief, SL. At the time of publication of this report, COL Allen F. Grum, USA, was Director of WES and Dr. Robert W. Whalin was Technical Director.

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## CONVERSION FACTORS, NON-SI TO SI (METRIC) UNITS OF MEASUREMENT

Non-SI units of measurement used in this report can be converted to SI (metric) units as follows:

Multiply	By	To Obtain
acceleration of gravity (g's)	9.80665	metres per second squared
cubic feet	0.02831685	cubic metres
cubic inches	16.387064	cubic centimetres
feet	0.3048	metres
g's per pound (force)	2.20462243	metres per second squared per newton
gallons (U. S. liquid)	0.00378541	cubic metres
gallons (U. S. liquid) per minute	0.00006309	cubic metres per second
horsepower (electric)	746.0	watts
inches	2.54	centimetres
inches	0.0254	metres
inches per second	0.0254	metres per second
kips (1000 lb force)	4448.222	newtons
ounces (mass)	28.34952	grams
pounds (force) per square inch	6894.757	pascals
square inches	6.4516	square centimetres
tons (2000 1b mass)	907.18474	kilograms

### FORCE-PULSE TESTS OF TACTICAL COMMUNICATION EQUIPMENT

PART I: INTRODUCTION

#### BACKGROUND

- 1. Mobile tactical communication systems used by the U. S. Army are housed in a shelter typically mounted on a 2-1/2-ton\* truck. Airblast generated by high-explosive or nuclear weapons loads the shelter, thereby inducing transient vibrations to the communication equipment. Shelters containing the equipment have been tested in Events DICE THROW and MISERS BLUFF, and the shelters were scheduled to be part of the MILL RACE Event during 1981. Acceleration of various individual pieces of equipment as well as the equipment racks are typically measured in the field events.
- 2. The transient loadings and resulting motion of the equipment, as measured in the field events, are quite severe. However, due to the nature of the tests, it has not been possible to determine operational characteristics of the equipment before, during, and after the tests. A laboratory simulation device capable of generating specified force-time histories could be used to subject individual pieces of equipment to similar motions as measured in the actual field events. Such laboratory testing would be highly desirable in terms of cost-effectiveness and data obtainable and could be used to determine the vulnerability/ survivability of the different classes of equipment. A program was initiated to develop such a simulation device and subject various components of the communication system to loadings as might be encountered in a battlefield environment. Program management is being provided by the U. S. Army Electronics Research and Development Command (ERADCOM) and the U. S. Army Ballistic Research Laboratory (BRL). The simulation system is being developed and implemented jointly by the U. S. Army Engineer Waterways Experiment Station (WES) and Agbabian Associates (AA).

<sup>\*</sup> A table of factors for converting non-SI units of measurement to SI (metric) units is presented on page 3.

#### SCOFE

3. The simulation device, or force-pulse generator, is hydrau-lically driven. This report covers the design of the hydraulic power supply and discusses tests conducted on the communication equipment through December 1980. Design and operation of the pulse generator were previously described in the test plan (Reference 1), and calibration tests were reported in Reference 2.

#### PART II: HYDRAULIC POWER SUPPLY

- 4. The existing WES pulse-generator was powered by a hydraulic system using an air reservoir and air/hydraulic multiplier. This system is shown in Figure 1 and a schematic in Figure 2. The new system, consisting of two separate pulse generators and having a larger force capacity, would require a new power supply. Based on requirements of two pulsers each having a nominal 10,000-1bf capacity and a 13-in. stroke, a power system utilizing the air/hydraulic concept would have required:

  (a) a 14-in. to 3-in. air/hydraulic multiplier with a 20-in. stroke,

  (b) a 94-gal air reservoir, and (c) an input supply of 100 psi. Such a system would have been considerably larger than the existing system thereby greatly reducing the degree of portability. Due to size and operational characteristics, an alternative to the air/hydraulic multiplier concept was developed.
- 5. A power supply concept, utilizing a high-pressure pump and accumulator system, is shown schematically in Figure 3. Preliminary calculations, used for initial sizing of components, were made as follows:

Hydraulic cylinders:

associations continued by the property of the

10,000-lbf nominal design force 2500-psi operating pressure

Net area required =  $\frac{10,000 \text{ lb}}{2500 \text{ lb/in.}^2}$ 

$$= 4.0 in.^{2}$$

A 2-1/2-in. piston with 1-in. rod has a net effective area of  $4.124 \, \text{in.}^2$ 

Cylinder volume:

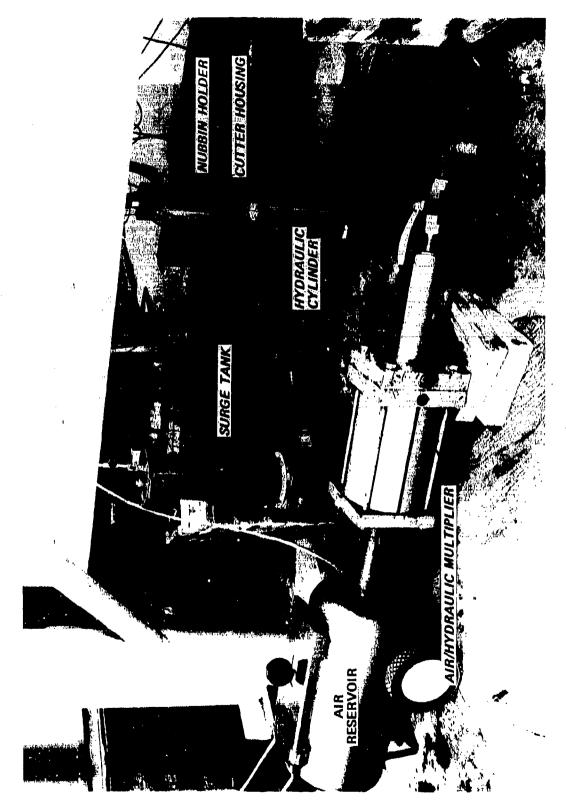
 $(4.124 \text{ in.}^2)(13-\text{in. stroke}) = 53.6 \text{ in.}^3$ 

Piston velocity: 120 in./sec (nominal)

Flow rate:

Single cylinder = 128.5 gal/min

Two cylinders = 257 gal/min



Initial WES pulser utilizing air/hydraulic multiplier in power supply. Figure 1.

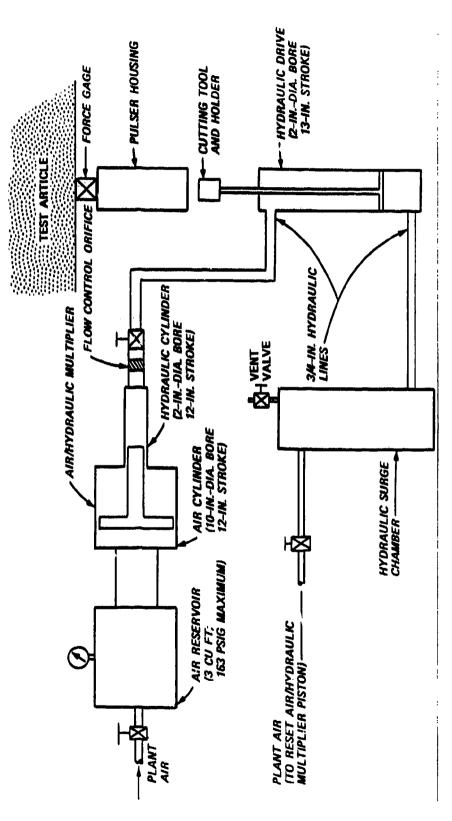


Figure 2. Schematic of initial WES pulser.

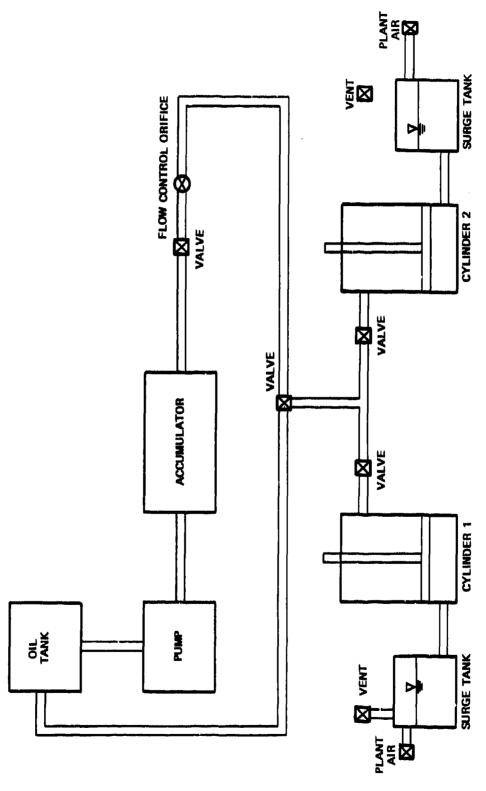


Figure 3. Hydraulic power supply schematic.

6. Gas-oil type accumulators operate on the principle of Boyle's Gas Law,  $P_1V_1^n=P_2V_2^n$ , where  $P_1$  and  $P_2$  are the initial and final gas pressures;  $V_1$  and  $V_2$  are the corresponding gas volumes; and n is the ratio between the specific heat of a gas at constant volume and its specific heat at constant pressure. For an isothermal condition (constant temperature) n = 1.0, and for an adiabatic condition (no heat transfer), n = 1.4. Generally, an adiabatic condition exists for rapid (less than 1 minute) expansion or compression of gas or when insulating materials are used in the accumulator. For sizing an accumulator as an auxiliary power source, the amount of fluid required from the accumulator to meet the system needs must be known. The following formula presents a simplified method of determining the capacity of the accumulator to be used (Reference 3, p. a-82):

$$v_{1} = \frac{v_{x} \left(\frac{P_{3}}{P_{1}}\right)^{1/n}}{1 - \left(\frac{P_{3}}{P_{2}}\right)^{1/n}}$$
 (1)

where:

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V<sub>1</sub> = size of accumulator necessary, in.<sup>3</sup> Maximum volume occupied by gas at precharge pressure

V<sub>x</sub> = volume of fluid discharged from accumulator, in.<sup>3</sup> Additional volume of fluid demanded by the system

 $P_1$  = gas precharge of accumulator, psi. Must be less than or equal to minimum system pressure,  $P_3$ 

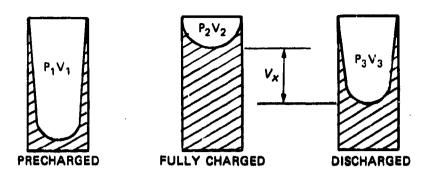
P<sub>2</sub> = maximum system design operating pressure, psi

 $v_2^-$  = compressed volume of gas at maximum system pressure, in.<sup>3</sup>

P<sub>3</sub> = minimum system pressure at which additional volume of fluid is needed, psi

 $V_3$  = expended volume of gas at minimum system pressure, in.<sup>3</sup>

n = 1.4



The maximum precharge pressure,  $P_1$ , was limited to that of a standard nitrogen bottle (2200 psi); considering the volume of two cylinders,  $V_{\chi}$  was taken to be 130 in.<sup>3</sup>; the required pressure for 10,000 lbf,  $P_2$ , was taken to be

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$$\frac{10,000 \text{ lbf}}{4.124 \text{ in.}^2} = 2424 \text{ psi}$$

and the minimum system pressure,  $P_3$ , was assumed to be 2100 psi. Using these parameters, Equation 1 resulted in a required accumulator volume of approximately 1400 in. A 10-gallon accumulator having a volume of 2080 in. Was selected as optimum.

7. The final system design is shown schematically in Figure 4, and a list of components is given in Table 1. Final sizing and selection of components and fabrication of the complete power supply was performed by Activation Inc. under contract to WES. A photograph of the completed power supply is shown in Figure 5.

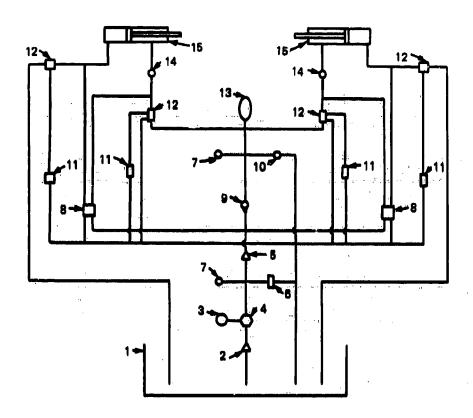


Figure 4. Hydraulic power supply circuit.

Table 1 Hydraulic Power Supply Component List

ITEM	QTY	DESCRIPTION	ITEH	QTY	DESCRIPTION
1	1	Thirty-gallon reservoir with cover, sight gage, flush drain, and baffles. Activation	9	1	Check valve, Gould No. DC 500
		No. T30L	10	1	Bull valve, Clayton Mark No. 1/2 CBB-790
2	1	Suction strainer, MPP No. 8R45	I		
3	١	Electric motor, 3HP, Lincoln No. 1827	11	4	Directional control valve, 4-way,   2-position, solenoid-operated,   Double A No. QJ-005-C-10B1
4	1	Pressure compensated pump, Hydura No. PVQ-06-LSAY-CNSN	12	4	Relief valve, Double A No. BT-12-12A2
5	1	High-pressure filter, 5 micron, MFF No. HF1-1-008	13	1	Ten-gallon accumulator, Greer No. 30A-10A
6	1	Relief valve, Sun No. RPGC-JAN-CEB	II		<u>                                     </u>
7	2	Pressure gage, 0-3000 psi, UCC No. UC-3907	14	2	Flow control valve, Double A No. YB12-10A1
			15	2	Hydraulic cylinder, 2-1/2-in. bore.
8	2	Directional control valve, 4-way, 3- position, solenoid-operated, Double A No. QF-01-C-10Fi			1-in. rod dismeter, 13-in. stroke, Sheffer Heavy Duty HH Series Hodel No. 2-1/2 HHRF13CRA

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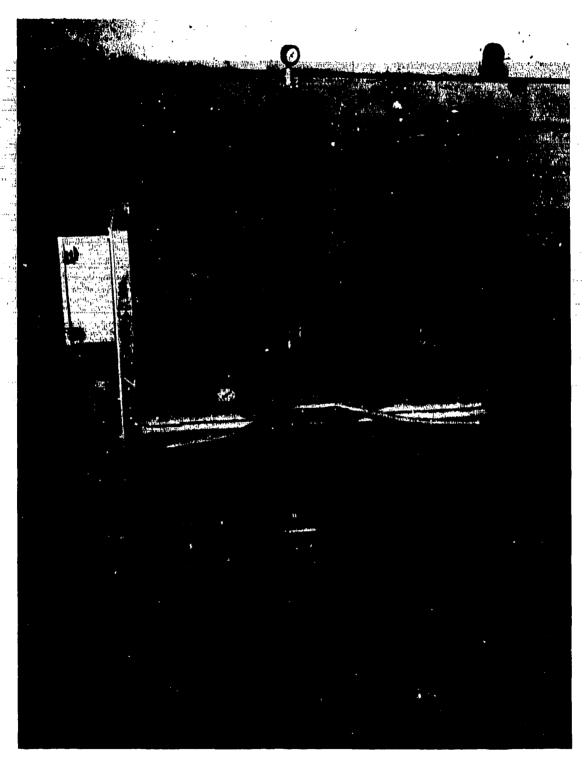


Figure 5. Hydraulic power supply.

#### PART III: TESTS CONDUCTED

#### CALIBRATION TESTS

An extensive series of calibration tests was conducted, and results are reported in Reference 2.

#### EQUIPMENT TEST SETUP

- The equipment rac': used in the mobile system and furnished to WES was modified for testing. The rack, constructed of 1-in. aluminum box tubing and 3/16-in. aluminum plate, was reduced in height to approximately 3 ft so that the total height of the system with vertical pulser attached would be less than 8 ft. To accommodate the largest pieces of radio equipment, maximum shelf spacing was not altered in the modified rack. Steel reaction structures to which the pulsers were attached were designed and fabricated. For attaching the pulsers to the equipment rack, a loading yoke ("quadrapod") arrangement was utilized. The loading yoke was fabricated from 3/4-in, steel box tubing and attached to the rack with three 3/8-in. bolts at four locations. The test setup is shown schematically in Figure 6 and as a photograph in Figure 7.
- INSTRUMENTATION
- 10. Accelerations were measured with piezoresistive shock accelerometers (Endevco Model 2264A). These are small gages (0.05 oz weight) having a high resonance frequency (30,000 Hz) and essentially zero damping thereby allowing accurate response to fast-rise-time, short-duration shock motion. Complete specifications are given in Table 2. The accelerometers were attached directly to the equipment with cynoacrylate adhesive.
- 11. Force measurements were made with piezoelectric force links (Kistler Model 936A) having a crystalline quartz sensing element. These transducers have a high natural frequency (25,000 Hz), high resolution (0.0004 percent), and very high sensitivity. Complete specifications are given in Table 2.
- 12. All data were recorded on an FM magnetic tape recorder and played back through a tuneable analog filter on oscillograph traces. Some data were also digitized for additional analysis. The tape

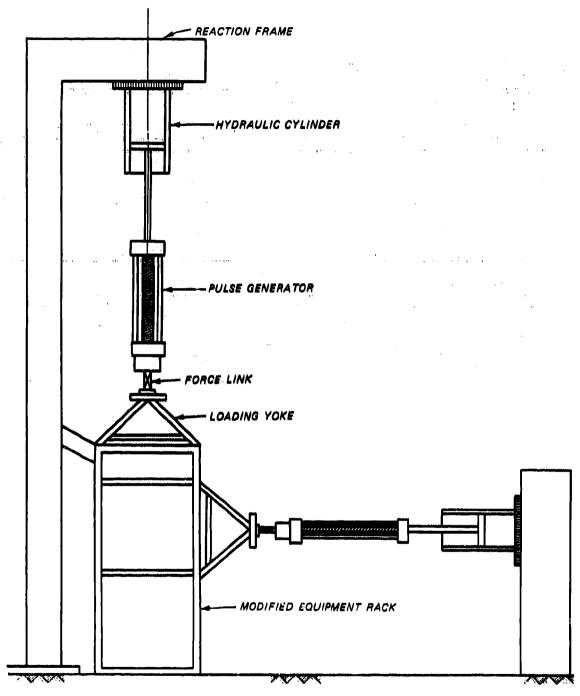


Figure 6. Schematic of biaxial test facility.



Figure 7. Setup for biaxial tests.

Table 2
Transducer Specifications

Endevco 2264A Accelerometer	
Range, g's pk	±2000
Sensitivity, mV/g at 10 Vdc, nom.	0.250
Mounted resonance frequency, Hz	30,000
Useful frequency response, Hz, dc to	5000
Environmental acceleration limit, g's pk	±5000
Damping factor, nom	0.002
Transverse sensitivity	5%, max
Weight, oz	0.05
Kistler 936A Quartz Force Link Range, 1b, tension	15,000
compression	25,000 0.1
Resolution, 1b	
Resonant frequency, Hz, unmounted	25,000
Weight, oz	50
Sensitivity	10 pC/lb
Overload capacity	10%
Linearity	1%
Useful frequency response, Hz, near dc to	5000
Environmental acceleration limit, g's pk	±5000
Rise time	20 µsec

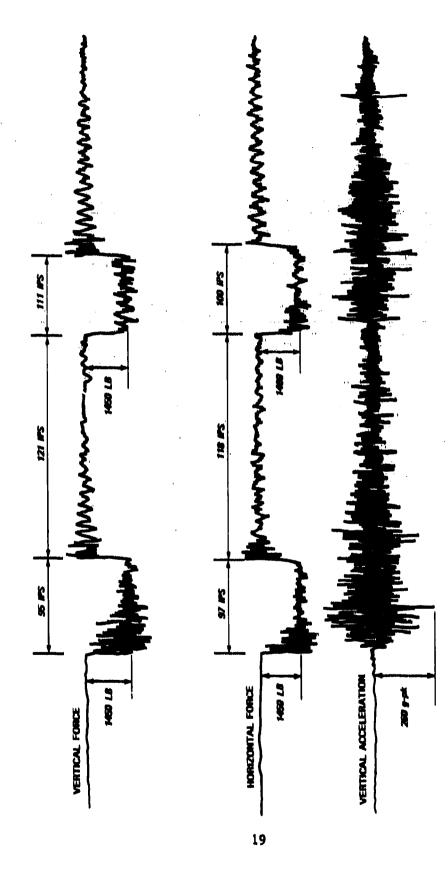
machines were operated at 30 ips, wide band, thereby having a frequency response from DC to 20 kHz (half-power point response).

BIAXIAL TESTS

- 13. After completion of the calibration test phase, a series of tests was conducted utilizing the biaxial capability of the pulser. As shown in Figure 7, both horizontal and vertical axes were excited. However, in lieu of an actual piece of communications equipment, lead weights were placed in the rack for the initial biaxial tests. No specific pulse train was used for these tests; arbitrary excitation was sufficient to check out the system. For these biaxial tests, both vertical and horizontal accelerations of the equipment rack were measured.
- 14. The two circuits in the hydraulic power supply have different response times. Delays of up to 30 msec are within specifications of the solenoid-controlled valves in the power supply. Thus, if both pulsers are simultaneously fired with a common circuit, one will start moving 10 to 30 msec before the other. For simultaneous initiation of both pulses (i.e., each cutter striking its first nubbin at the same point in time), it is necessary to position one cutter closer to the first nubbin than the other. After several test firings of both pulsers, simultaneous initiation of the first pulse was achieved with the following parameters:

Precharge accumulator pressure = 1250 psi
System operating pressure = 2000 psi
Flow control valve (vertical unit) = 4-1/2 turns
Flow control valve (horizontal unit) = 6-1/2 turns
Horizontal cutter initial run-up = 7.5 mm
Vertical cutter initial run-up = 5.1 mm

However, after repeated firing with the same parameters, initiation of pulses between the two units could be controlled to within only 7 msec. Typical data records from biaxial tests, produced directly from the tape recorder with no additional filtering, are shown in Figures 8 and 9. As can be seen in Figure 9, significant accelerations were recorded on the rack (800 g's both vertical and horizontal directions) with relatively moderate input forces (approximately 2000 lb). No significant problems



Biaxial pulse test data, simultaneous pulse initiation. Figure 8.

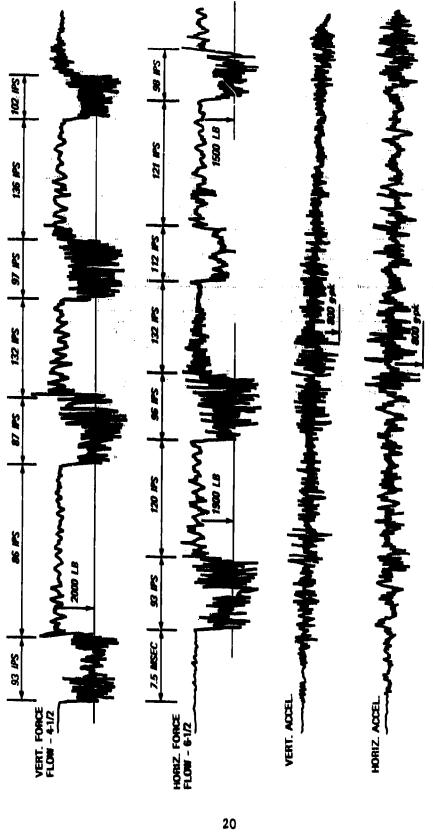


Figure 9. Biaxial pulse test data, delayed pulse initiation.

were encountered while operating the system in the biaxial mode, other than maintaining alignment, which is a critical inherent characteristic of the system.

#### IMPEDANCE TESTS

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- 15. Two sets of impedance tests were run. The first set, as specified in the original test plan (Reference 1), was conducted using a hammer to produce a force impulse. Data from these tests were to be used in deriving a pulse train to simulate field test response records. The other set of impedance tests was conducted using vibration sine sweeps. Frequency response data of the radio equipment and rack were obtained from these vibration tests. All impedance tests were conducted with the rack hard-mounted to the floor and reaction structure.
- and 11 were utilized for the hammer impedance tests. Force impulses, induced by a calibrated hammer having a force link attached to its head, excited the system, and the resulting accelerations were measured in both horizontal and vertical directions. Figure 10 shows the arrangement for the vertical impedance test. Horizontal testing was also performed with the vertical pulser attached and the input being applied horizontally. Source impedance measurements were made, as shown in Figure 11, by exciting the pulse generators directly (pulse generators disconnected from equipment rack) and measuring the drive point acceleration. A nonoperational AN/GRC-103 radio system was mounted in the rack for all the impedance tests. The data were recorded on FM magnetic tape and later digitized. The digital records were forwarded to Agbabian Associates for analysis and use in the pulse optimization algorithms.
- 17. Vibration tests. Once the degree of acceleration attenuation from equipment rack to radio had been observed, vibration tests were conducted. Frequency response data of both the equipment rack and AN/GRC-103 radio were obtained from these tests. The pulsers were disconnected from the rack, and an electromagnetic vibrator was attached to the horizontal loading yoke (Figure 12). Frequency sweep tests from 100 to 10,000 Hz were conducted using a 2-1b sinusoidal input. Other

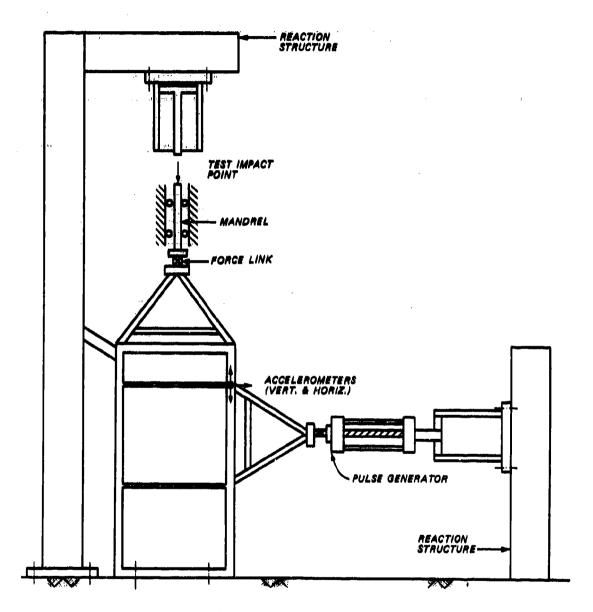
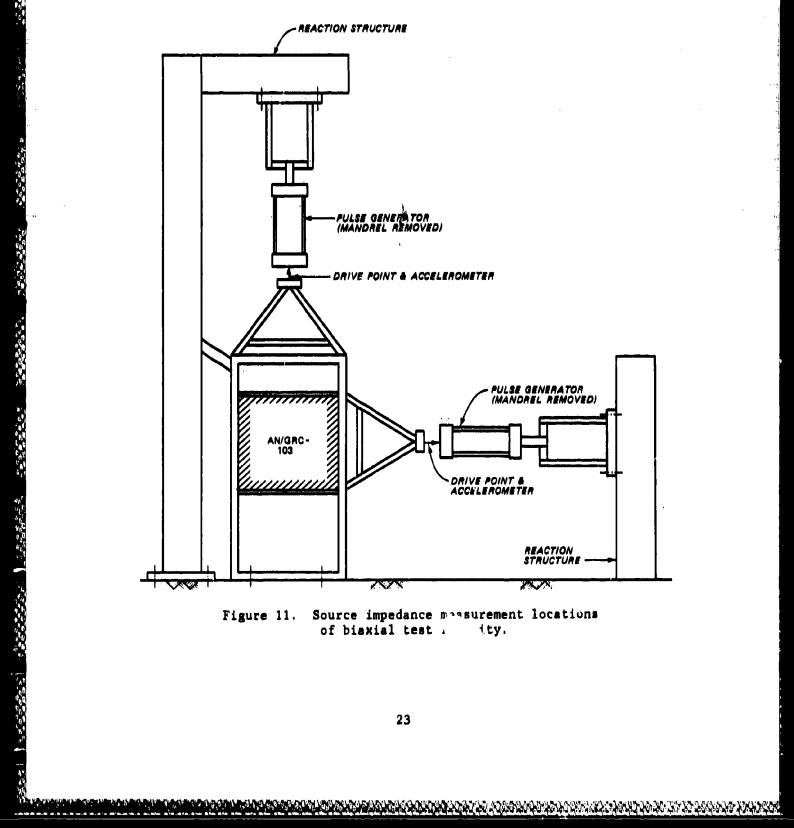


Figure 10. Vertical impedance test of biaxial test facility.



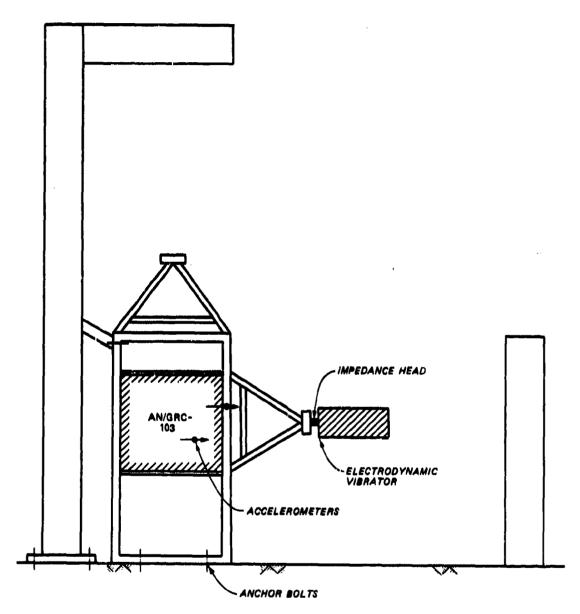


Figure 12. Vibration test setup.

tests were conducted using 1-, 5-, and 20-lb input forces. Accelerations were measured at the point of loading (drive point acceleration) on a vertical leg of the rack and on the face of the AN/GRC-103 transmitter (directly in front of a stiffening plate).

18. Data from the vibration tests are shown in Figures 13-16.

Figures 13, 14, and 15 are inertance plots, (acceleration response)

versus frequency. Figure 13 is the drive point inertance, i.e.,

(drive point acceleration); Figure 14 is the inertance plot of the rack; input force

and Figure 15 is the inertance plot of the radio as mounted in the rack. A comparison of Figures 14 and 15 reveals that general response characteristics of the rack and radio are quite similar. Several resonance peaks are observed in the 100- to 400-Hz range, with distinctive antiresonances particularly prevalent in the 400- to 700-Hz range. Numerous resonances occur at frequencies above 1000 Hz, the strongest being at approximately 2600 Hz. Above 4000 Hz, the radio response has more antiresonances than the rack. Figure 16 is a transmissibility plot, the ratio of radio and rack accelerations versus frequency. As can be seen, there are numerous, high Q (lightly damped) resonances and antiresonances throughout the frequency range. There are distinctive peaks in the curve at frequencies below 400 Hz, particularly in the 150- to 200-Hz range, representing amplifications of up to 10 of radio versus rack accelerations. However, the curve generally falls in the 0.1 to 1.0 amplitude range signifying reduced accelerations being transmitted from the rack to the radio. This energy is apparently lost in the radio/ rack bolted connection.

#### TESTS OF OPERATIONAL EQUIPMENT

19. Two units of operational equipment were delivered to WES for testing: (a) AN/GRC-103 Transmitter and Receiver, and (b) TD660 Multiplexer. These units were brought to WES, installed in the rack, and operated on-line during testing by personnel from the U.S. Army Depot, Tobyhenna, Pa. Horizontal, uniaxial pulse tests were conducted on these units utilizing the same instrumentation as in the impedance tests; i.e.

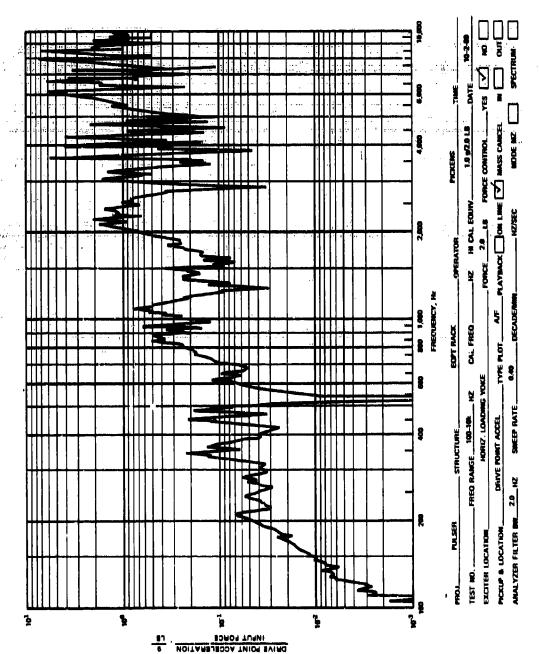


Figure 13. Vibration test data, drive point inertance.

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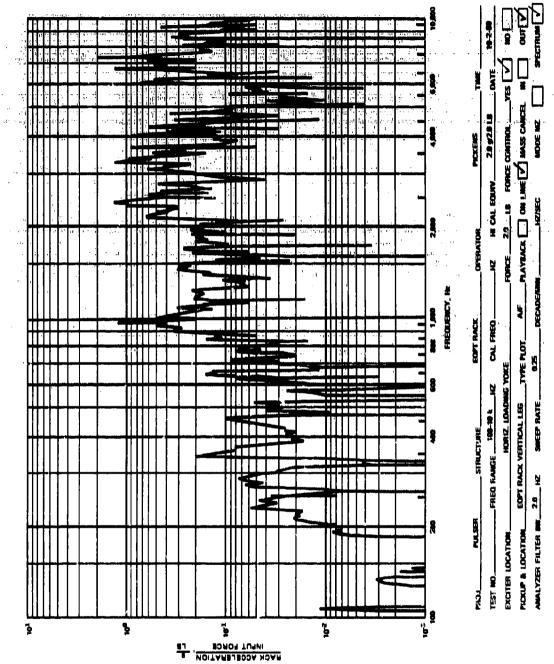


Figure 14. Vibration test data, equipment rack inertance.

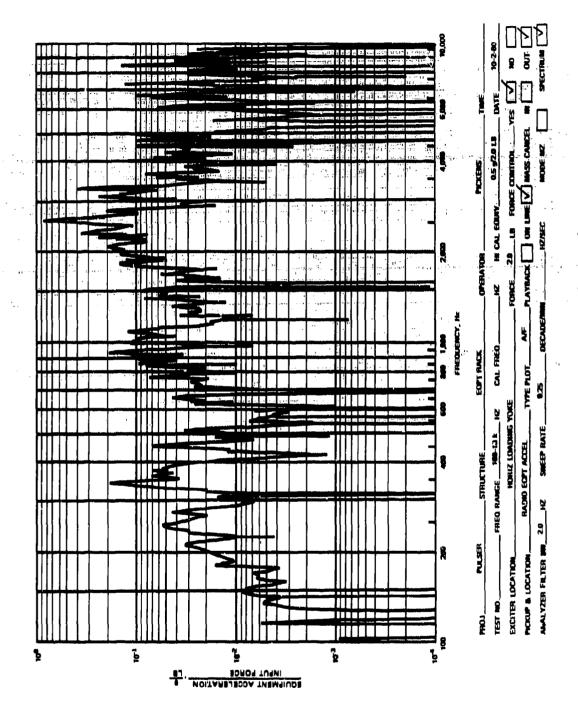


Figure 15. Vibration test data, equipment inertance.

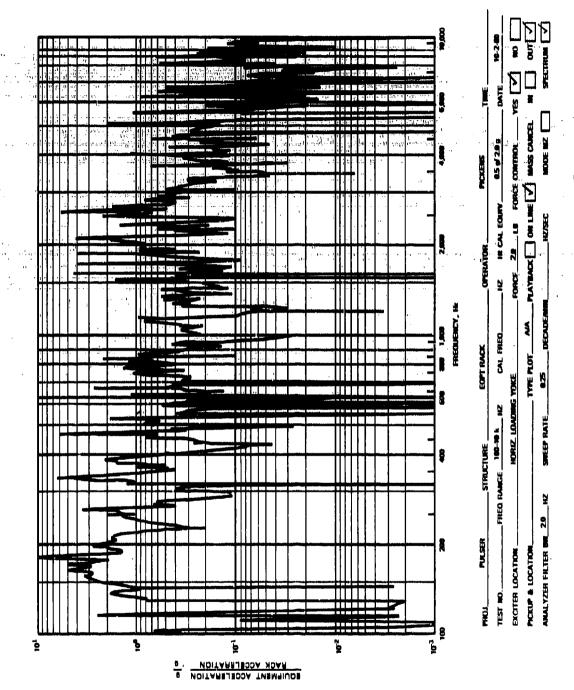


Figure 16. Vibration test data, transmissibility plot.

accelerometers on the rack vertical leg and on the face of the equipment along with the force link. Initial tests were run with the AN/GRC-103 installed, but not on-line, followed by on-line tests of both the AN/GRC-103 and TD660.

- 20. Both transmitter and receiver were mounted in the rack using standard mounting angles and screws. Six tests were conducted utilizing different pulse trains (Figure 17) in the horizontal axis. The tests are summarized in Table 3, and data for each test are presented in Appendices A-F. The data were digitized at rates of 20,000 and 10,000 samples per second, thereby limiting the useful frequency ranges to 10 kHz and 5 kHz, respectively (based on conventional digitizing procedures). Data for each test are in the form of time histories, fast Fourier transforms (FFT's), auto correlations, cross correlations, and cross spectral density records. For the data in the frequency domain (at the 10K digitizing rate), both fine and coarse plots were made; i.e., the curves were defined with either 1024 points or 512 points. The coarse data plots simply give gross approximations of the curves, often better for visualization of the data.
- 21. There was no apparent damage to any of the pieces of equipment during the tests. The off-line tests of equipment were conducted using nonoperational units followed by the tests of operational units. For the operational, on-line tests, the units were turned on several minutes before testing and allowed to operate several minutes after testing. Even though relatively high acceleration levels were measured on the rack (up to 200 g's peak), the equipment suffered no apparent damage. This result led to a closer look at what the measured accelerations actually indicate.
- 22. The motions measured on the rack and those transmitted to the equipment are highly dependent on the particular pulse train. Maximum utilization of the pulse generator, leading to maximum loading of the equipment, can only be accomplished using specifically designed pulse trains taking into account the dynamic characteristics of the total system being tested. The energy accepted by the equipment is frequency-sensitive, and the frequency content of a particular pulse train cannot

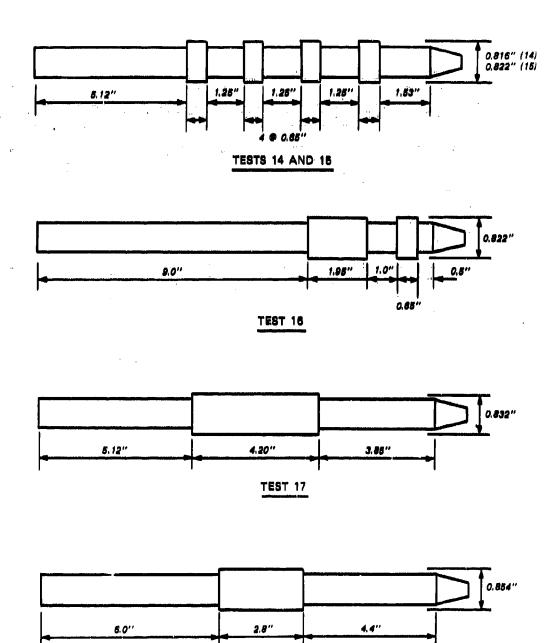


Figure 17. Pulse trains for tests 14-20, rack hard-mounted.

TESTS 18 AND 20

Summary of Tests 14-20, Rack Hard-Hounted

		Rack A	Acceleration	Г	Equipmen	it Acce	Equipment Acceleration			
	Average		3		•	3			Low-Pass	
Test No.	Force (1b)	МАХ	ROES	RHS HAX	WW	SHE	IRES	EQPT RPS	Filter (kBz)	Hotes
14	1100	516	111	0.23	<b>93</b>	20	0.23	0.18	01	AN/GRC-103, off-line
	1000	911	<b>%</b>	0.24	69	18	97.0	25.0	S	
15	1400	428	3	0.16	165	92	91.0	0.38	10	
	1400	232	**	0.16	<u>8</u>	20	0.18	0.52	'n	
91	1500	477	<b></b>	0.17	130	77	0.18	0.30	2	
	1400	175	<b>58</b>	0.16	98	15	0.17	0.54	10	
17	2400	896	176	0.18	208	**	0.23	0.27	2	
	2200	202	47	0.23	152	7.7	0.78	0.89	2	
81	0004	2029	257	0.13	167	33	0.19	0.12	2	AN/GRC-103, on-line
	0007	772	114	0.15	ž.	Z	0.26	0.19	· S	
70	0005	1750	302	0.17	148	8	0.20	0.10	01	TD660, on-line
	0004	908	*	0.24	<b>22</b>	22	0.25	0.12	VC.	

be predetermined without a detailed analysis. However, the general pulse trains used in these initial tests were quite useful in determining how the equipment responds to motions input to the rack.

The accelerations measured on both the rack and the equipment contain sharp, high-frequency peaks. The 10-kHz filtered data contain rack acceleration peaks of up to 4.8 times those of the 5-kHz limited digitized records (Test 17, Table 3). For Test 15, the 10-kHz rack peak accelerations were only 1.8 times those of the 5-kHz records. However, the pulse trains of Tests 15 and 17 are significantly different. Perhaps more important than peak response is the RMS value of measured accelera-The 5-kHz RMS acceleration response of the equipment in Test 17 is nearly equivalent to that of the rack (42 g's versus 47 g's). For this pulse train, then, the transmissibility of energy from rack to equipment was quite high (89 percent). However, in Test 20, the RMS equipment response was only 22 g's compared to rack RMS response of 194 g's, giving a transmissibility of only 12 percent. The ratios of equipment RMS response to rack RMS response are shown in Table 3 for each test. It is of interest to note that although the pulse trains of Tests 20 and 17 are quite similar (except for depth of cut; i.e., nubbin diameter), these two tests represent the extremes of equipment-to-rack response ratios (0.89 for Test 17 and 0.12 for Test 20). Upon comparison of the FFT's for the force input for each test (plot No. 90, Appendix D, for Test 17 and plot No. 98, Appendix F, for Test 20), it is seen that the primary frequency content is 2880 Hz for Test 17 and 2620 Hz for Test 20. This relatively small difference in frequency content of input force can significantly change the transmissibility ratio of equipment-to-rack response. This is quite apparent when considering the transmissibility plot of Figure 16 from the vibration tests. Extremely large differences in the ratio, on the order of 1000 to 1, exist for only minute changes in frequency. This is simply the nature of the rack-equipment system. It is a high Q, lightly damped, ringing type of structure having numerous resonances. It must also be kept in mind that the vibration tests were conducted using a 2-lbf input and the resonances and antiresonances for both rack and radio could change significantly for a

force input of several thousand pounds due to nonlinearity in the system.

24. Consider the force input FFT for Tests 14-20. With the exception of Tests 14 and 15, there is a predominant peak occurring in the 2500- to 3000-Hz region, indicating that most of the input force is at this frequency. This harmonic motion can also be seen on the auto correlation functions of the force inputs. This motion is due to chatter in the pulse generator as the cutter plows through a nubbin. The chatter is also apparent in the cut nubbin as the cut surface contains ripples rather than being smooth. The deeper and longer the cut, the greater the amount of chatter. During the calibration tests, a different cutter geometry was tried in an effort to reduce this chatter. However, the new cutter, which utilized a rake angle, did not reduce chatter; it actually increased it somewhat. The pulse train used in Tests 14 and 15 contained several short nubbins, as opposed to a single longer nubbin, and the chatter is not as severe. This fact is reflected in the FFT's of the input force for these tests, which do not exhibit the large spike in the 2500- to 3000-Hz ragion.

# TESTS OF SOFT-MOUNTED EQUIPMENT

25. In an effort to introduce higher accelerations into the equipment at lower frequencies, the equipment rack mounting configuration was changed. For all previous tests, the rack was secured directly to the concrete floor with four anchor bolts and tied to the reaction structure with a steel angle and bolts. This type mounting arrangement was considered to be a "hard mount." The "soft-mount" arrangement, shown schematically in Figure 18 and in the photographs, Figures 19-21, consisted of using Firestone Airmount Isolators (air bags) for all horizontal support. The floor anchor bolts were removed as was the horizontal steel angle positioned at the top of the rack. The rack was attached to four air bags (two at the top and two at the bottom) as shown in Figures 18-21. The sir bags used, Firestone Airmount No. 125, have a natural frequency of 160 Hz and will deflect approximately 3 in. before bottoming out. Tests were conducted with the bags pressurized at values ranging from 30 to 70 psi. Having an effective area of 11.5 in. 2 each, or 46 in. 2 total, the four bags could resist a load of up to 1380 lb with each bag

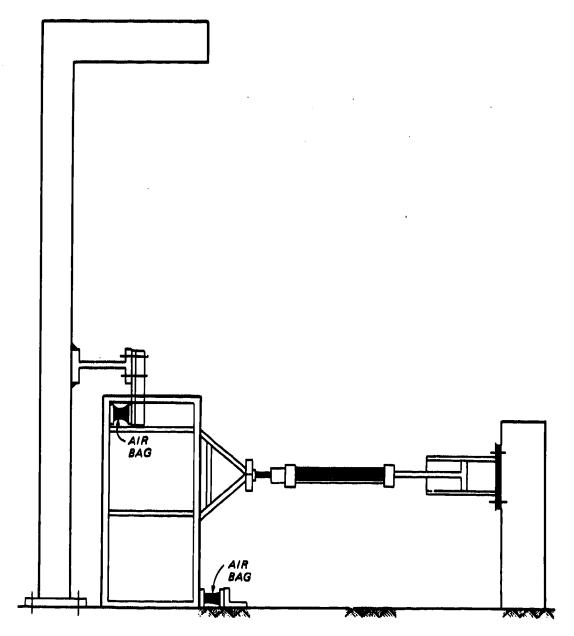


Figure 18. Schematic of soft-mount test setup.



Figure 19. Test settip with rack solt-mounted (view A).



Figure 20. Test setup with rack soft-mounted (view B).

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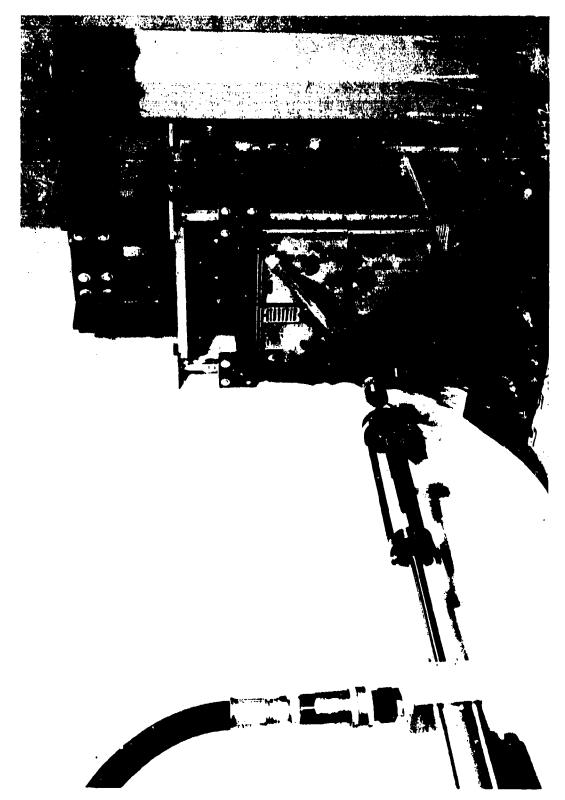
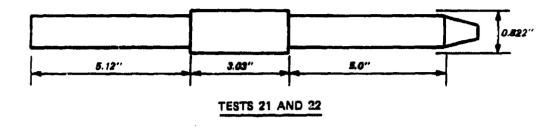


Figure 21. Test setup with rack soft-mounted (view C).

pressurized at 30 psi and up to 3220 lb at 70 psi (net area × pressure).

- 26. A nonoperational AN/GRC-103 transmitter and receiver was placed in the rack and the system was instrumented as in the previous tests; i.e. a force link, an accelerometer on the rack vertical leg, and an accelerometer on the equipment face plate. Five tests were run using the pulse trains shown in Figure 22. The data are summarized in Table 4 and presented in graphical form in Appendices G-K.
- 27. In general, the RMS value of the rack accelerations was 9 to 20 percent that of the peak acceleration, whereas, for the equipment, the RMS value ranged from 14 to 40 percent that of the peak. Playbacks of the time histories, FFT's, and cross spectral density records using low-pass filters of 10, 5, 3.5, 2.5, 1.5, and 0.5 kHz are given in the appendices. Such filtered plots offer a better picture of the frequencydependent equipment motion. For the five tests conducted, the RMS value of the 500-Hz filtered equipment acceleration ranged from 25 percent to 50 percent that of the 10-kHz filtered RMS values. The ratio of equipment-to-rack RMS accelerations varied from 0.08 to 0.16 (considering only 5- and 10-kHz filtered data). For the hard-mounted tests, this same ratio ranged from 0.10 to 0.89. Thus, in general, more energy was transmitted from the rack to the equipment in the hard-mounted tests than in the soft-mounted tests. Displacements for the soft-mounted tests were substantially larger, however, as the rack was observed to move approximately 2 in. For the hard-mounted tests, displacements were on the order of 0.1 in.



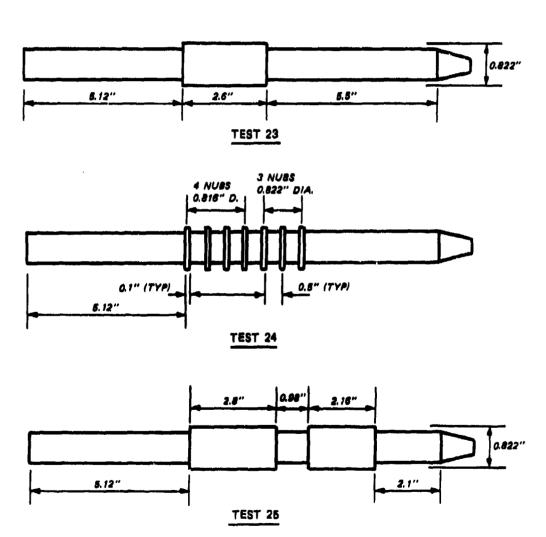


Figure 22. Pulse trains for Tests 21-25, rack soft-mounted.

Table 4
Summary of Tests 21-25, Rack Soft-Mounted on Air Bags

	Average Force (1b)	Rack Acceleration			Equipment Acceleration, (g)				Low-Pass
Test No.		MAX	(g) RMS	RMB MAX	MAX	RMS	RMS MAX	EQPT RACK RMS	Filter (kHz)
21	900	1244 1145 1400 1150 900 680	233 252 227 203 177 138	0.19 0.22 0.16 0.18 0.20 0.20	84 70 70 55 30 25	19 19 18 15 9	0.23 0.27 0.26 0.27 0.30 0.32	0.08 0.08 0.08 0.07 0.05 0.06	10 5 3.5 2.5 1.5 0.5
22	1000	1302 1160 1300 1100 930 600	203 210 192 169 142 102	0.16 0.18 0.15 0.15 0.15 0.18	63 50 70 30 22 20	16 16 15 11 8	0.25 0.32 0.21 0.37 0.36 0.40	0.08 0.08 0.08 0.06 0.06 0.08	10 5 3.5 2.5 1.5 0.5
23	1000	1252 1100 700 650 350 130	134 142 120 93 65 31	0.11 0.13 0.17 0.14 0.19 0.24	98 90 70 38 22 21	19 19 17 12 8 8	0.19 0.21 0.24 0.32 0.36 0.38	0.14 0.13 0.14 0.13 0.12 0.26	10 5 3.5 2.5 1.5 0.5
24	1500	1221 1200 1500 1226 850 320	155 163 132 107 84 54	0.13 0.14 0.09 0.09 0.10 0.17	99 80 70 40 25	14 14 12 8 7 5	0.14 0.18 0.17 0.20 0.28 0.33	0.12 0.09 0.09 0.07 0.08 0.09	10 5 3.5 2.5 1.5 0.5
25	1100	1238 1174 1150 900 700 350	184 191 160 120 93 57	0.15 0.16 0.14 0.13 0.13 0.16	152 130 130 80 22 20	29 28 27 16 8 7	0.19 0.22 0.21 0.20 0.36 0.35	0.16 0.15 0.17 0.13 0.08 0.12	10 5 3.5 2.5 1.5 0.5

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#### PART IV: CONCLUSIONS AND RECOMMENDATIONS

## CONCLUSIONS

- 28. The effort to date has been successful. A unique test device has been developed and has been shown to be an effective means for subjecting communication equipment to acceleration levels that might be encountered in a battlefield condition. Based on results of tests thus far conducted, the following conclusions are drawn:
- <u>a</u>. A biaxial force-pulse generating device (pulser) has been developed which has a force output capacity of approximately 10,000 lb.
- $\underline{b}$ . The pulser can be controlled to initiate simultaneous force pulses in both horizontal and vertical axes.
- $\underline{c}$ . Up to 2000-g-peak and 300-g-RMS acceleration levels have been induced in the equipment rack with up to 200 g's peak and 48 g's RMS being transmitted to the equipment in the rack.
- d. The AN/GRC-103 and TD660 communication equipment sustained no damage while being operated on-line during force-pulse tests which produced 300 g's peak (32 g's RMS) in the equipment.
- e. For the pulse trains thus far used in the test program, acceleration levels measured on the equipment were 10 to 89 percent (average of 34 percent) of those measured on the rack with the rack hard-mounted and 8 to 16 percent with the rack soft-mounted on air bags (considering 5-kHz low-pass filtered data).
- <u>f.</u> The present cutter being used in the pulser produces a certain amount of tool chatter as the aluminum nubbins are cut. This chatter results in a high concentration of force in the 2500- to 3000-Hz region. For cuts without the chatter, the force input is more broadbanded without large concentrations of energy at particular frequencies. RECOMMENDATIONS
- 29. For a successful program of testing communication equipment with the force-pulse generator, the following recommendations are offered:
- a. Specific pulse trains should be designed utilizing the dynamic characteristics of the equipment and equipment-rack system.

Furthermore, the dynamic characteristics should be obtained from highlevel excitation such as an actual force-pulse test.

- <u>b</u>. Additional development work should be done in an effort to reduce tool chatter. Areas to consider include cutter shape, depth of cut, and nubbin material. Perhaps a softer material, such as nylon, Teflon, Micarta, etc., using greater depths of cuts with the existing cutter would produce acceptable force levels with reduced chatter.
- <u>c</u>. A realistic acceleration standard, to which the communication equipment could be subjected in a laboratory test environment, should be developed. Once this standard is known, a pulse train should be designed which will result in equipment response matching the standard.

### REFERENCES

- Agbabian Associates. 1979 (Dec). "Biaxial Transient Vibration Simulation for C<sup>3</sup> Equipment to Match Field Test Records-Tests Plan," Report R-8014-4994, El Segundo, Calif.
- US Army Engineer Waterways Experiment Station and Agbabian Associates. 1980 (Oct). "Biaxial Transient Vibration Simulation for C<sup>3</sup> Equipment-Pulse Train Calibration," Report R-8014-5136, El Segundo, Calif.
- 3. Industrial Publishing Co. 1972/73. Fluid Power Handbook and Directory, Cleveland, Ohio.

#### APPENDIX A: TEST 14 DATA

TEST 14 Equipment Rack Hard-Mounted AN/GRC-103 in Rack, Off-Line

Plot Heading Nomenclature (Tests 14-20)

CHAN 1 - Input force (data at 20,000 digitizing rate)

CHAN 2 - Rack acceleration (data at 20,000 rate)
- Input force (data at 10,000 rate)

CHAN 3 - Equipment acceleration (data at 20,000 rate)
- Rack acceleration (data at 10,000 rate)

CHAN 4 - Equipment acceleration (data at 10,000 rate)

RMS, AVG, MAX, XMAX, XMIN, MEAN - RMS, average, maximum, minimum and mean values of data plots

FREQ - Frequency at which maximum value occurs

FFT 0.0 - Fast Fourier transform of time history

TIME 0.0 - Time history record

CSDF 0.0 - Cross spectral density function

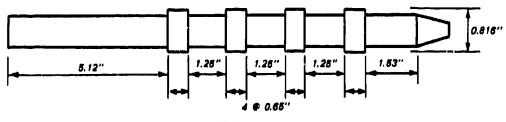
CH/2 1.0 Channel number of the second channel for plots involving

CH/2 2.0 two-channel functions (CSDF)

CH/2 3.0

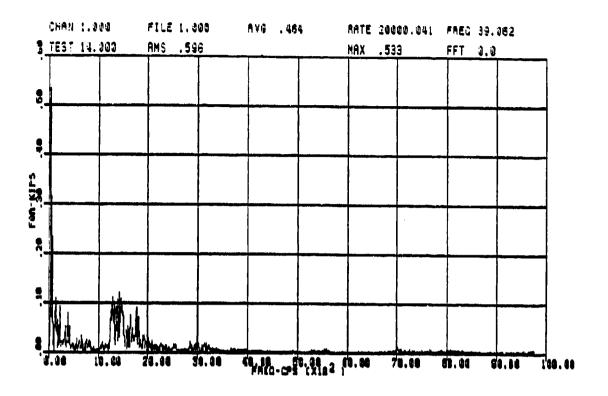
POINTS - Number of points used for plotting the record

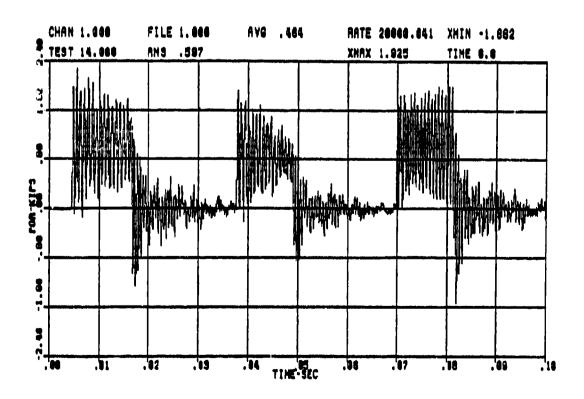
RATE - Digitizing rate, samples per second



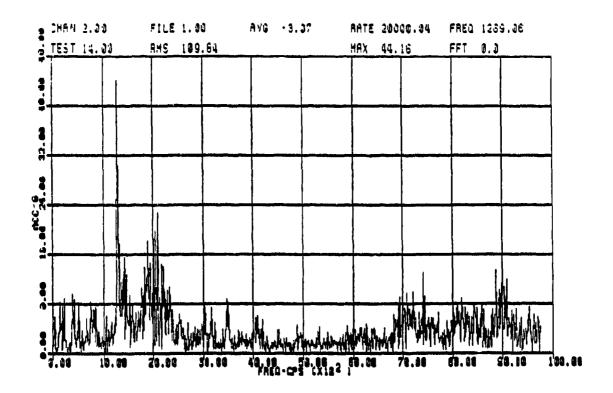
PULSE TRAIN - TEST 14

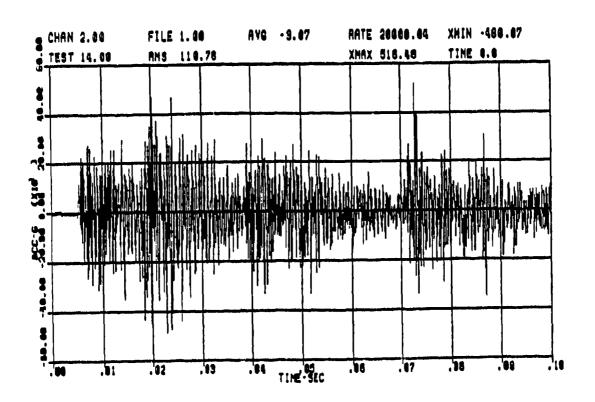
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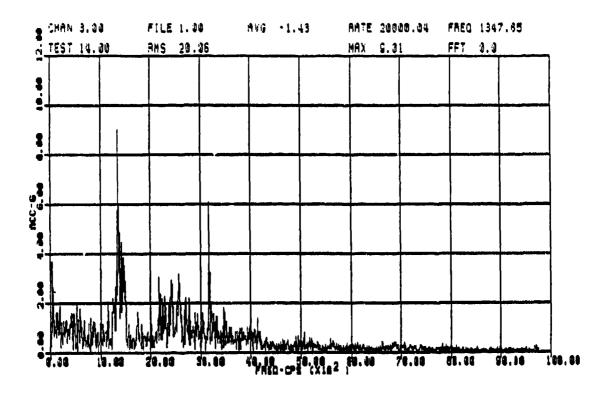


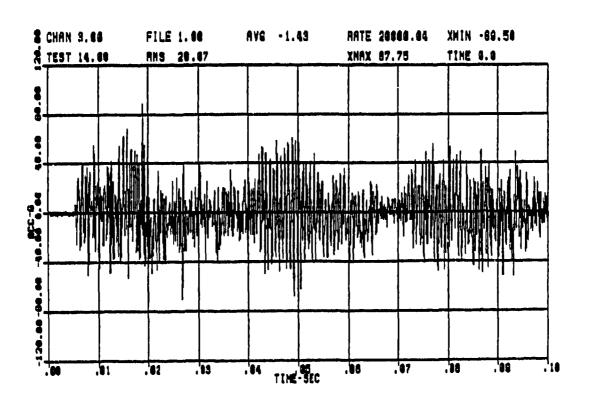


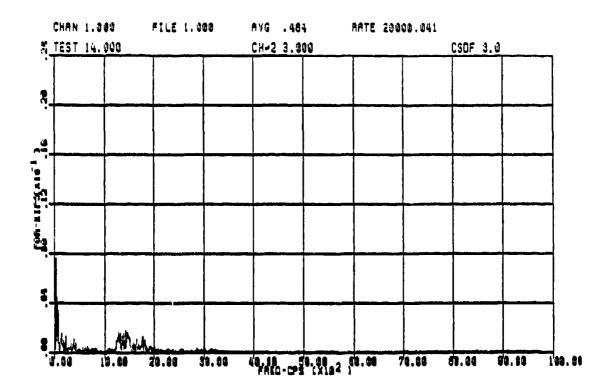
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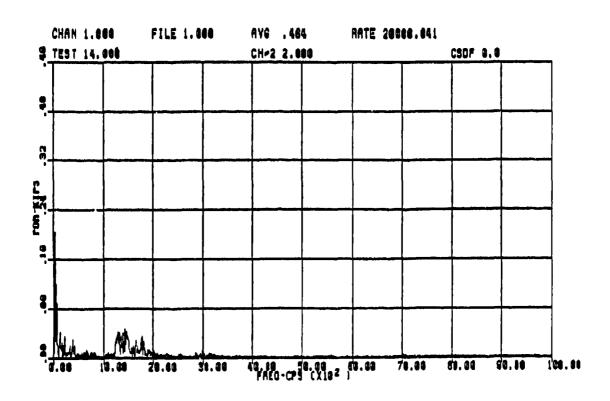


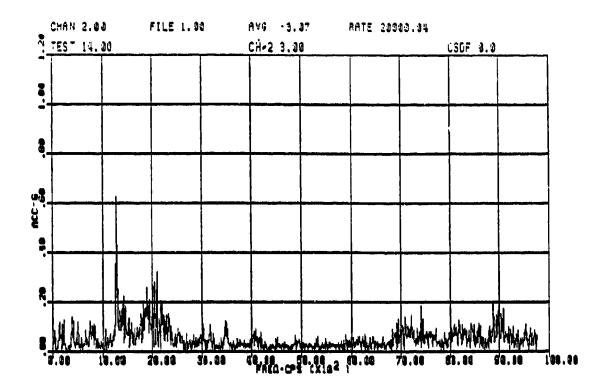


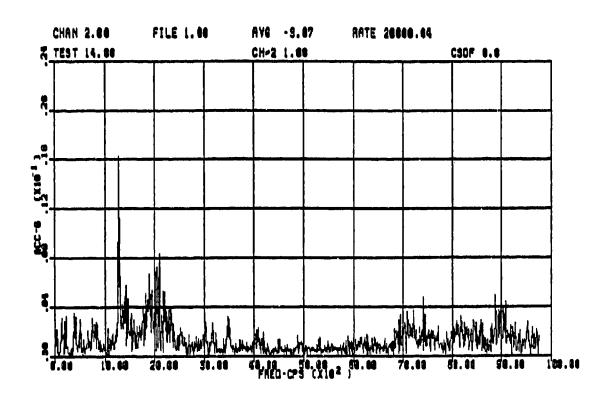


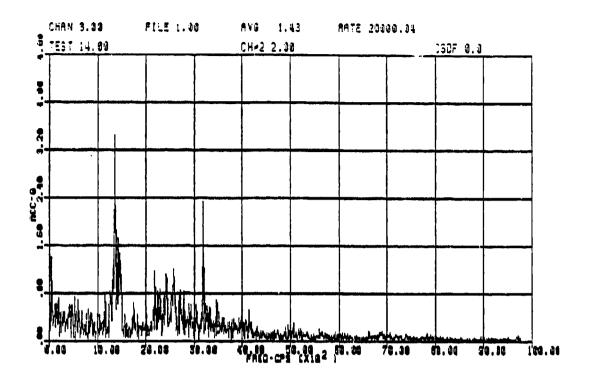


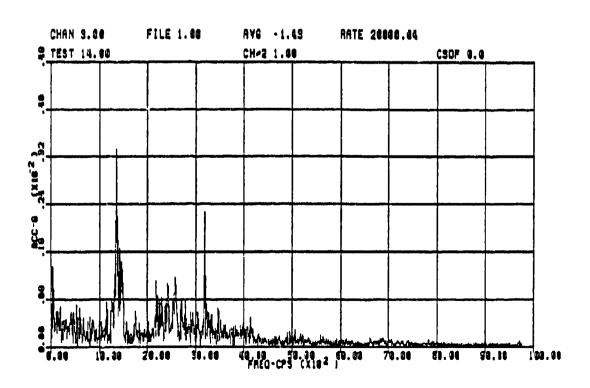


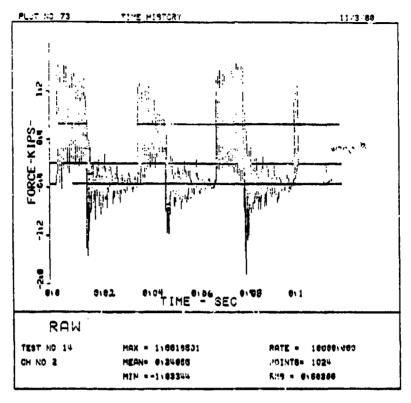






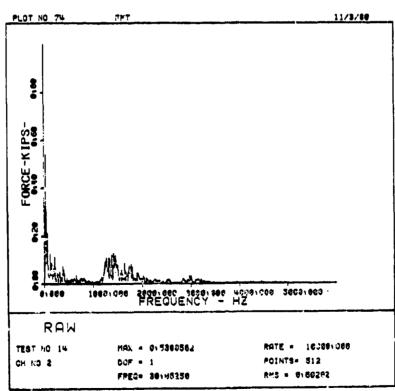




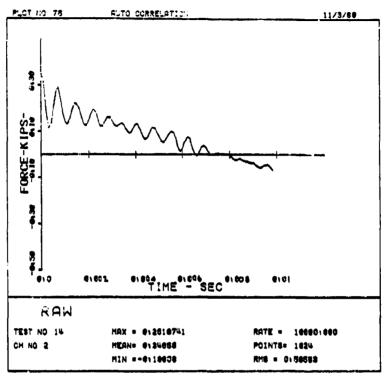


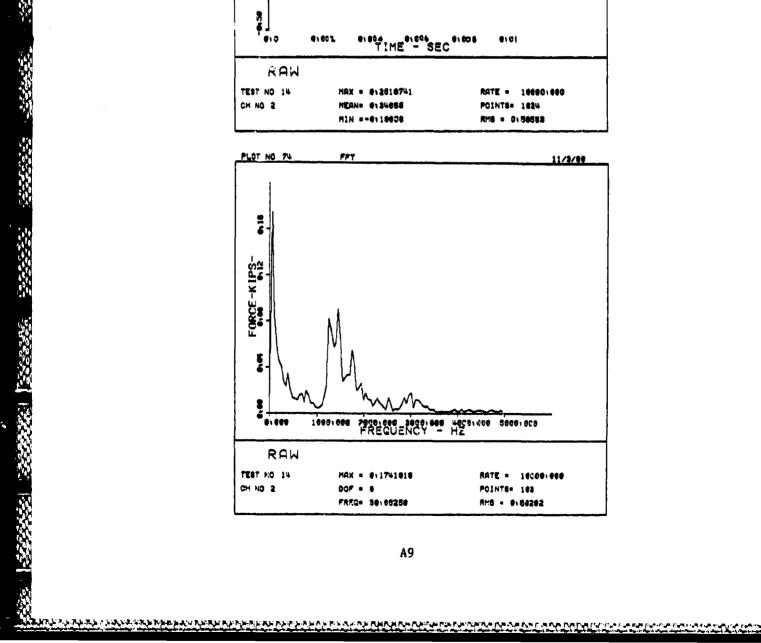
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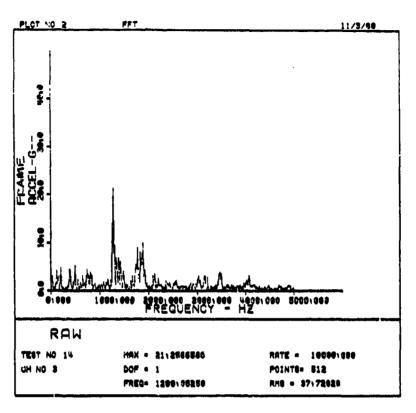
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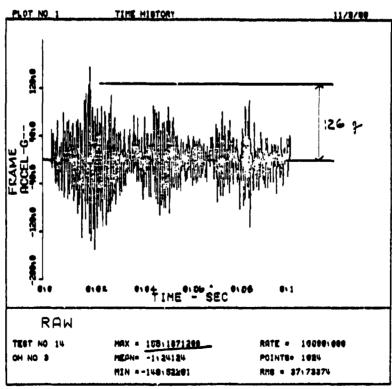


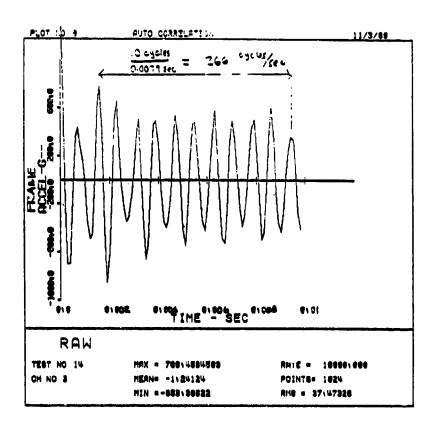


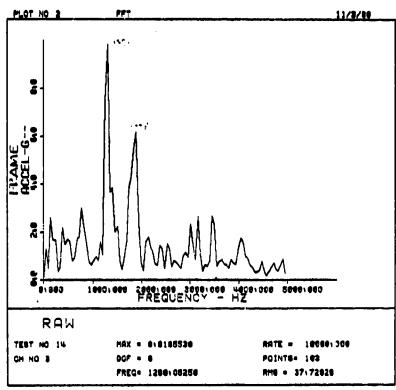


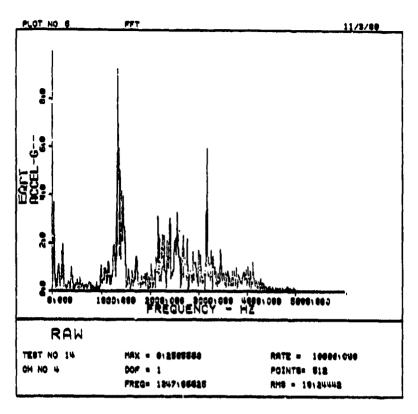
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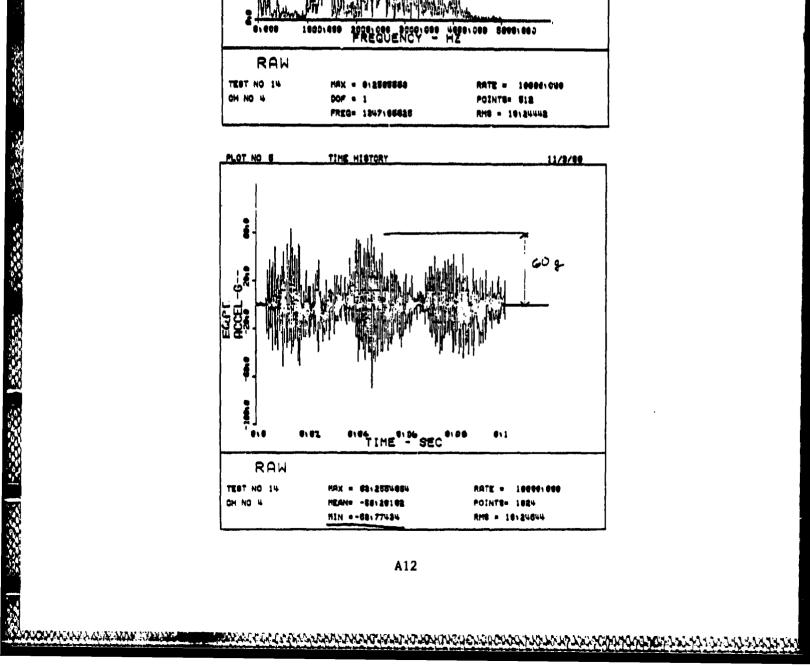
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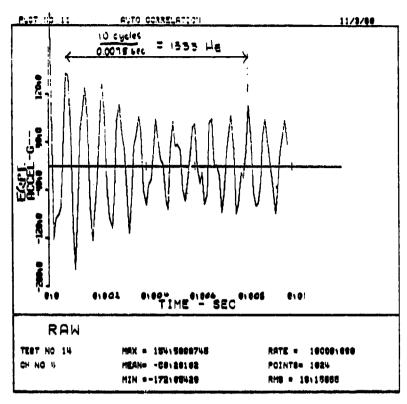


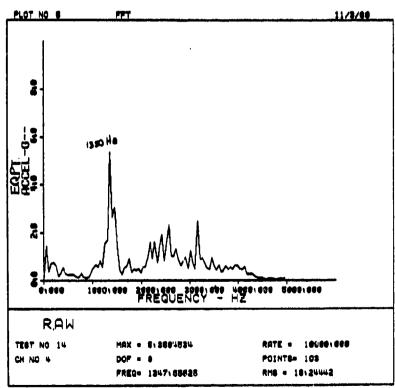


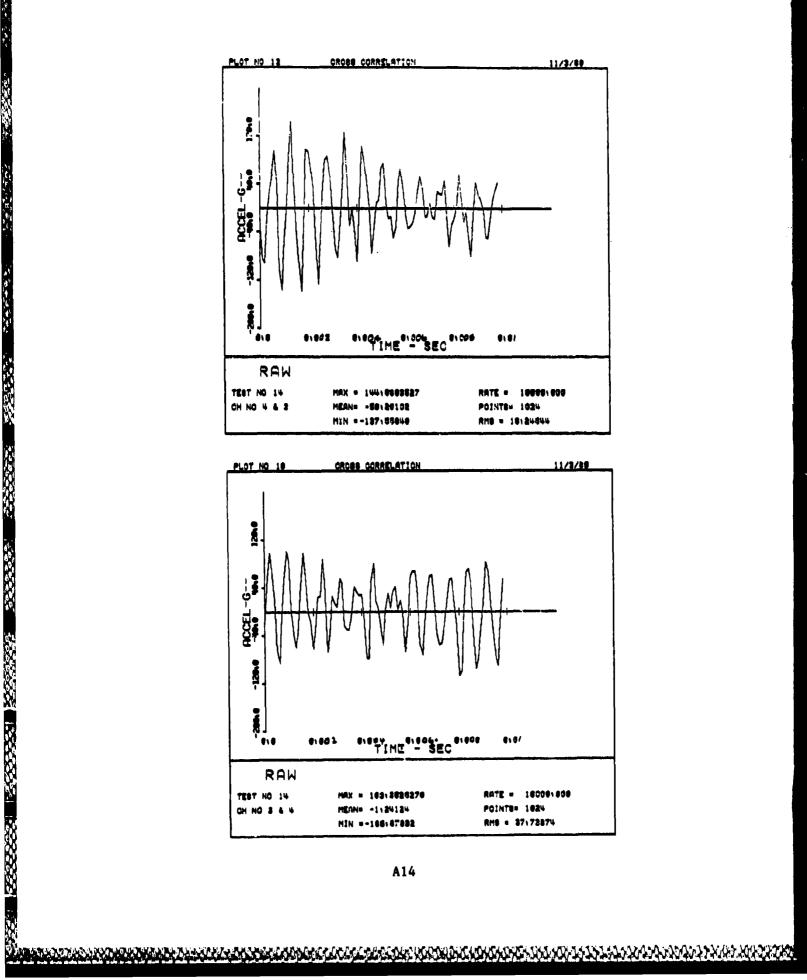


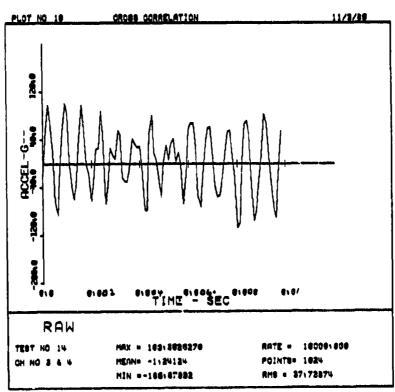


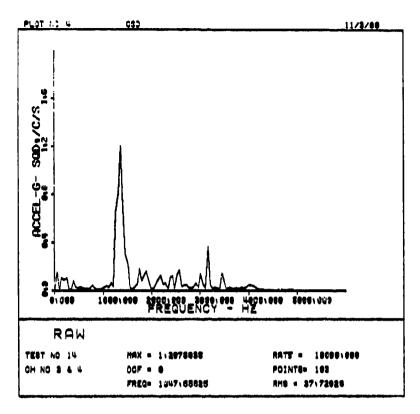


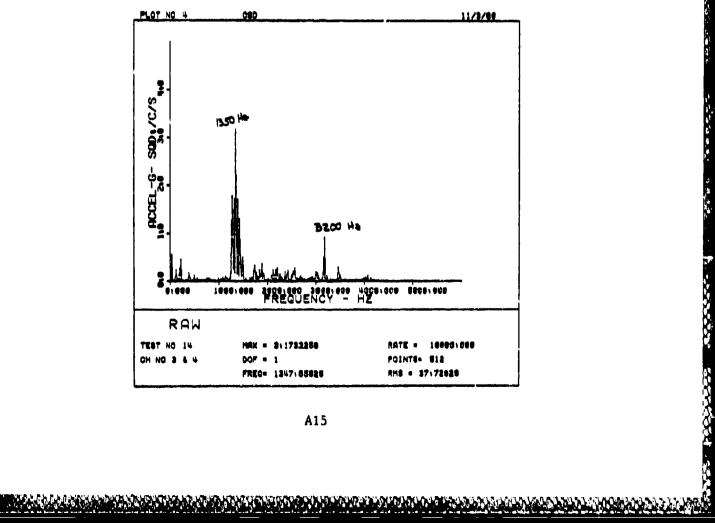












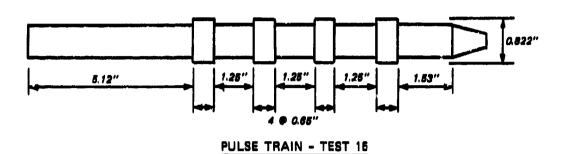
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# APPENDIX B: TEST 15 DATA

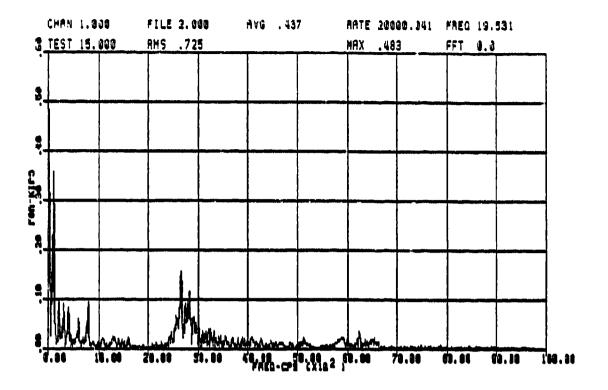
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Equipment Rack Hard-Mounted
AN/GRC-103 in Rack, Off-Line

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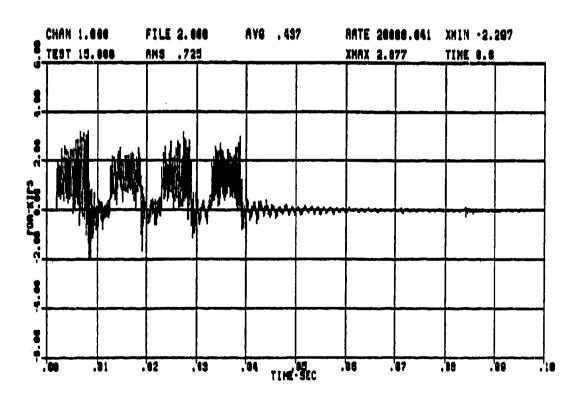
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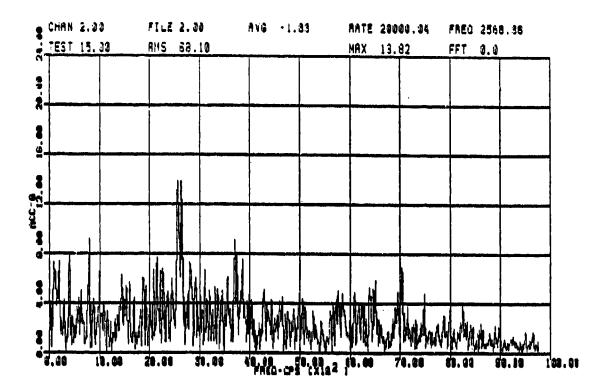


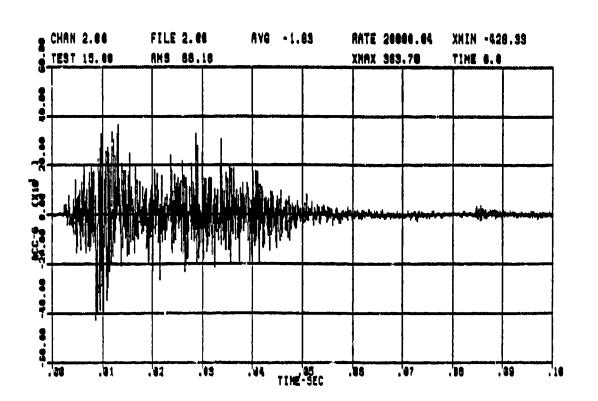
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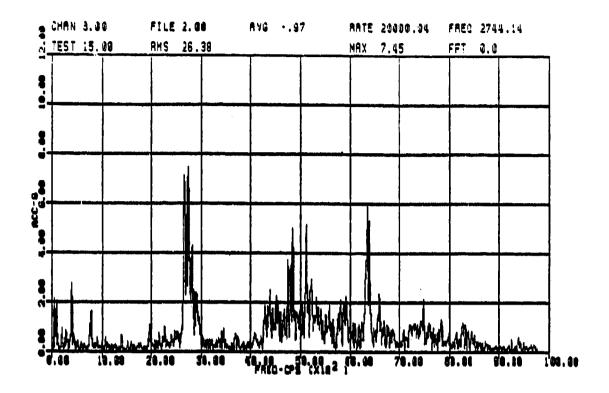
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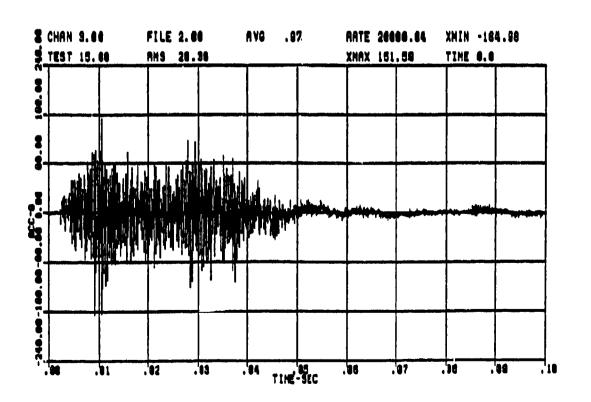


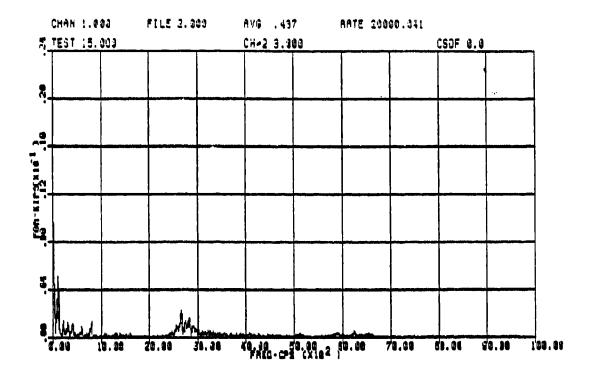


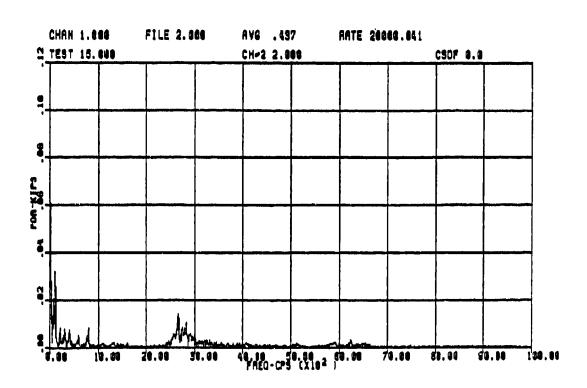
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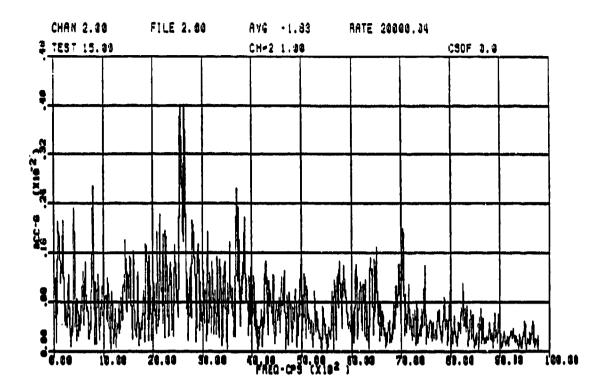


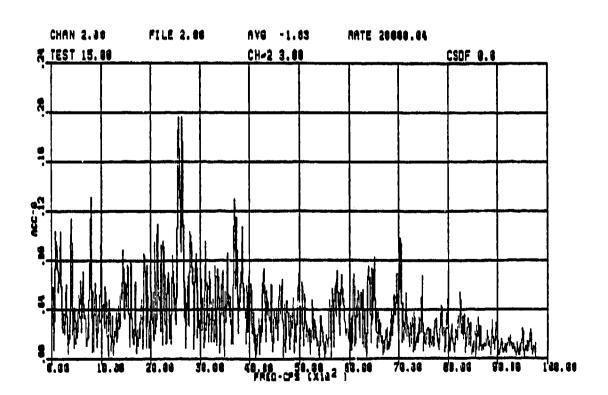




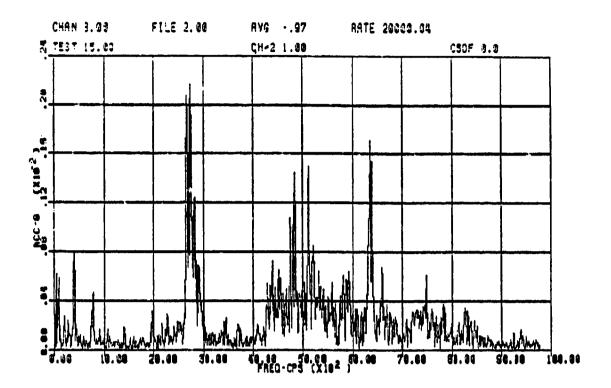
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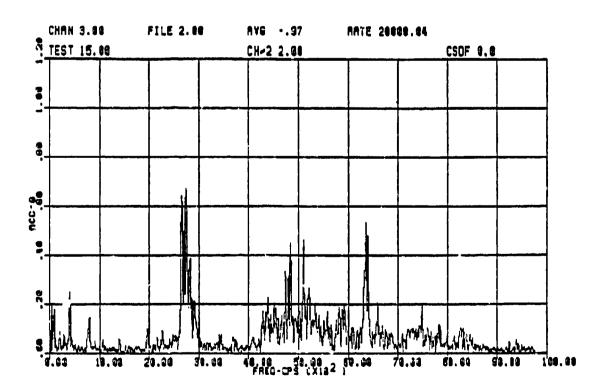
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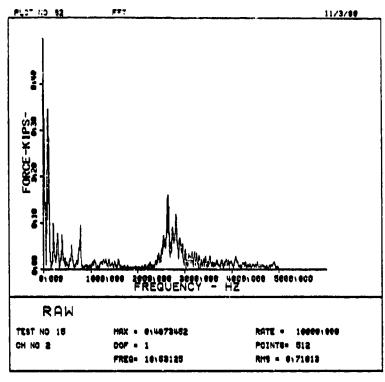


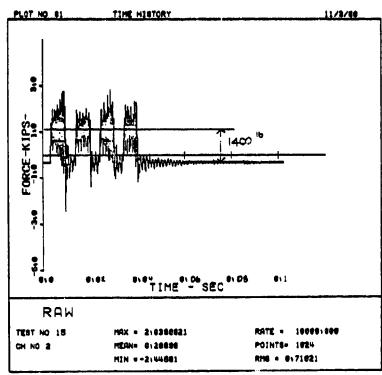


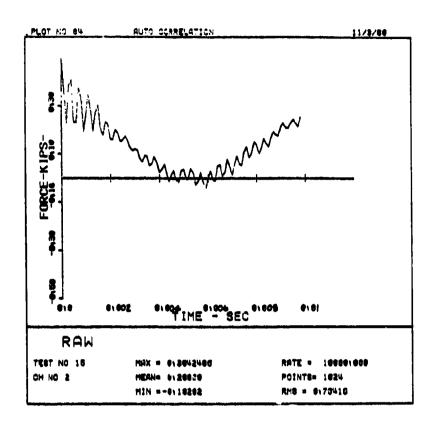
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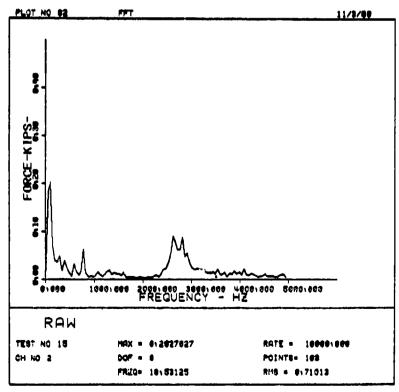


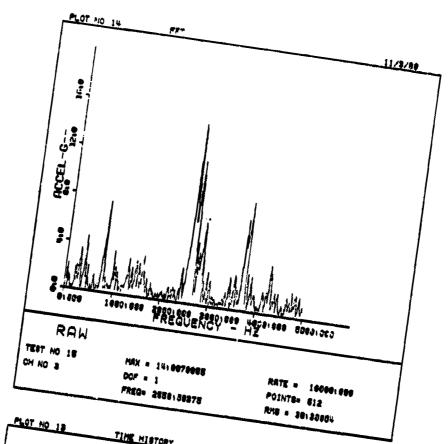


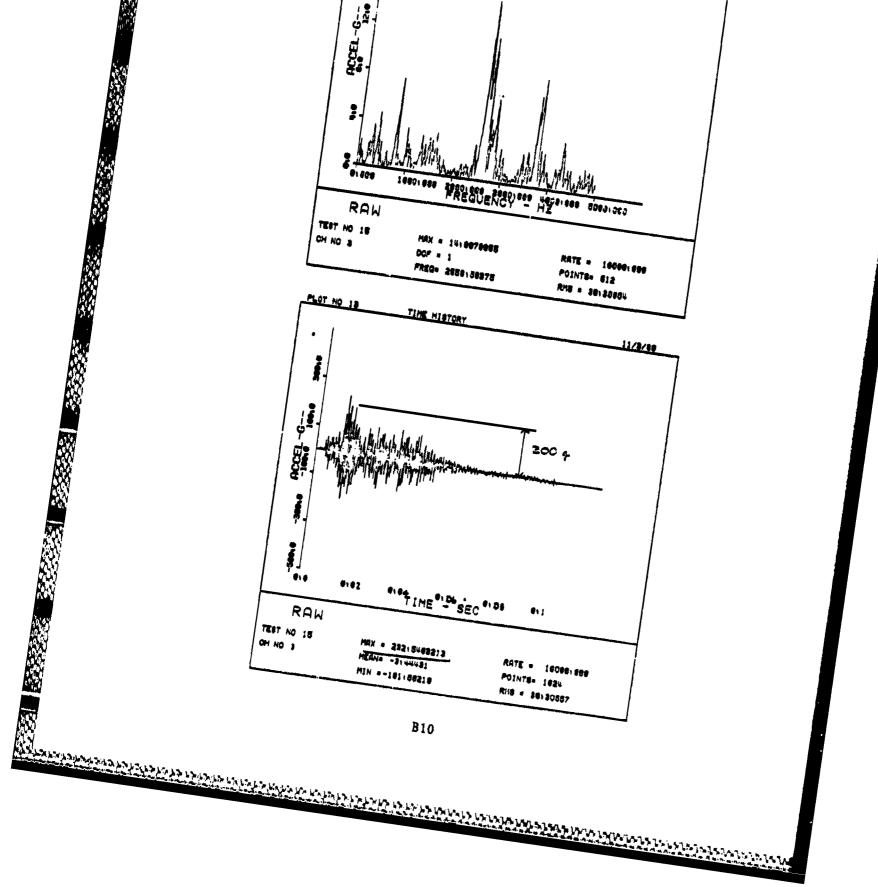


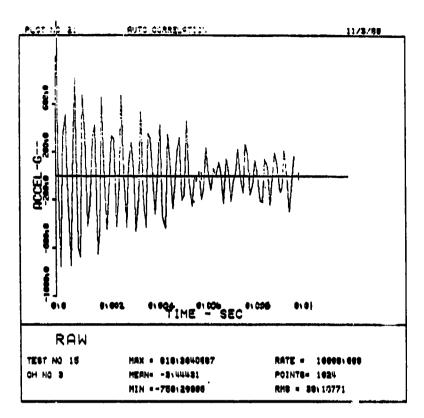




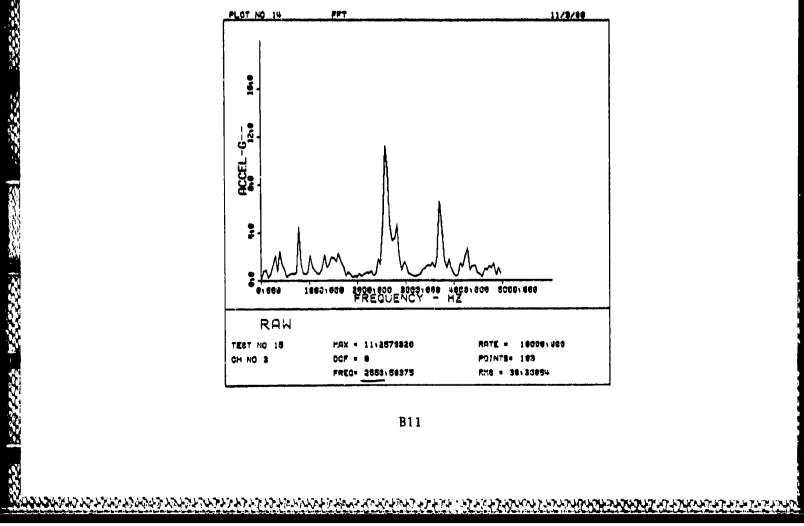


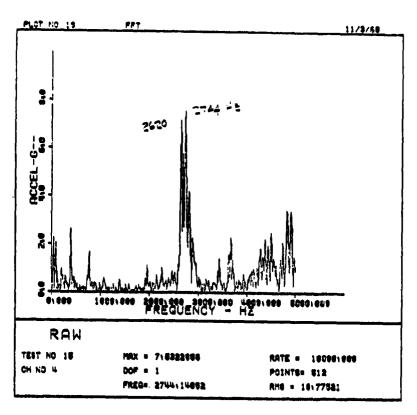


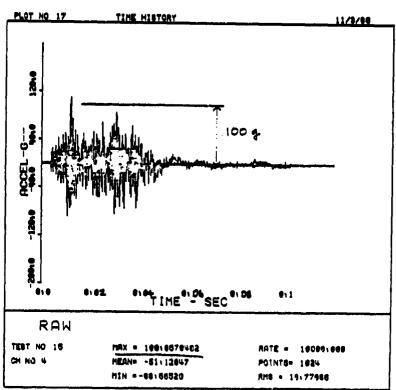


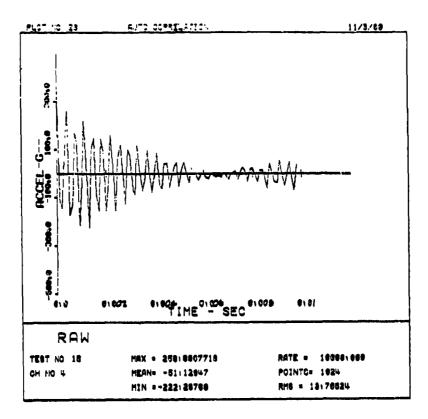


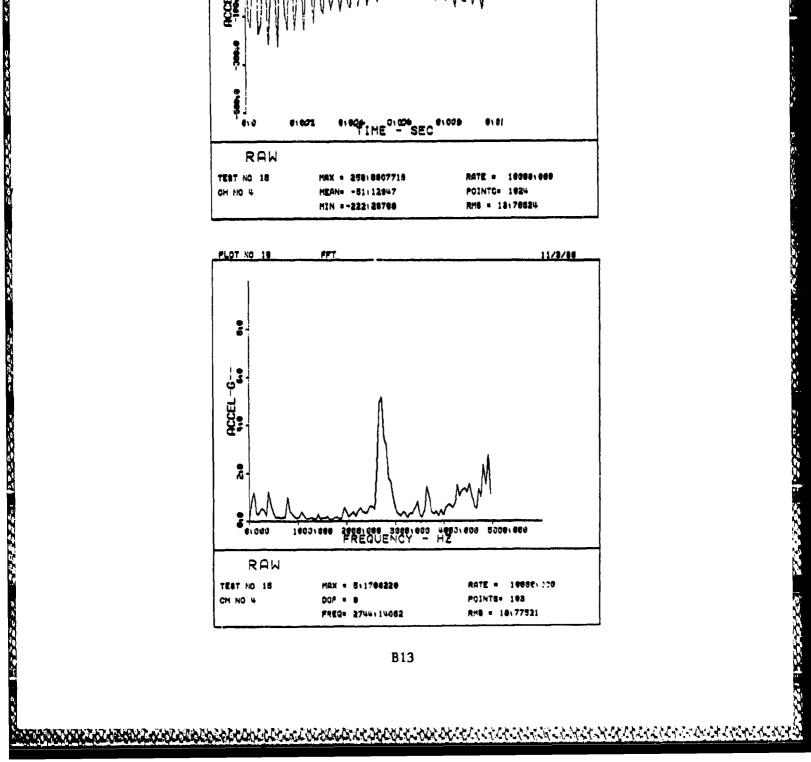
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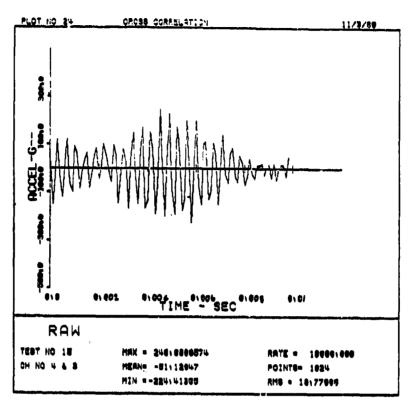


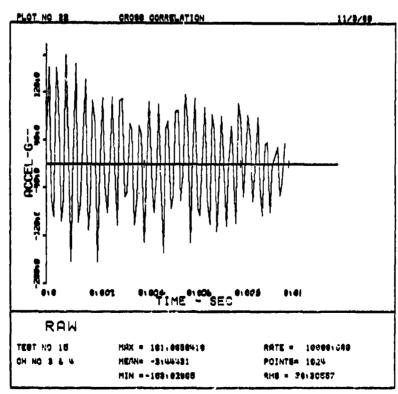


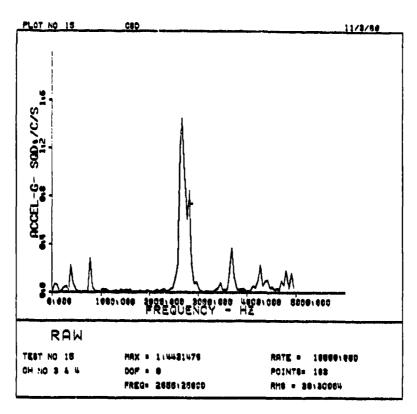






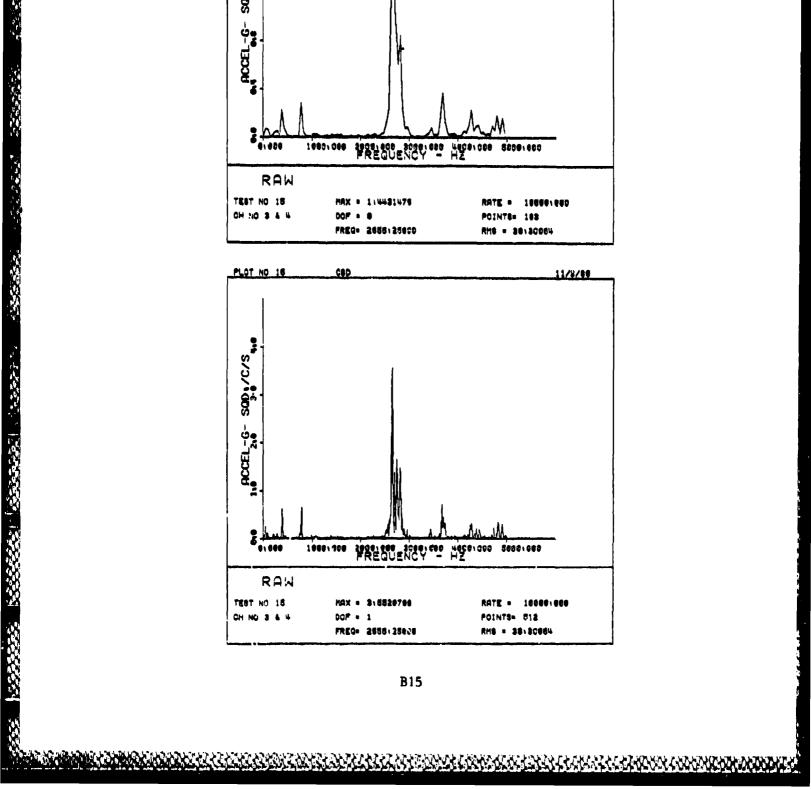






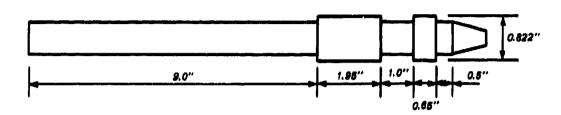
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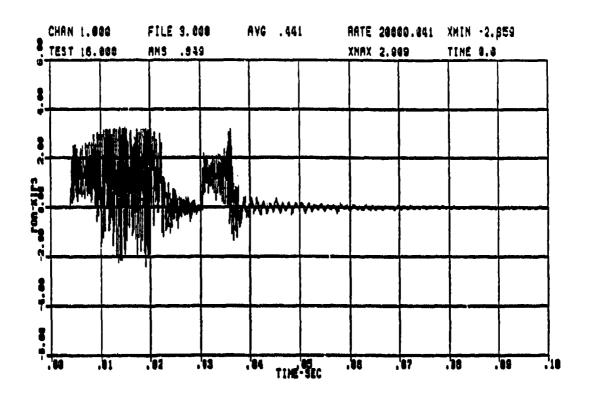
## APPENDIX C: TEST 16 DATA

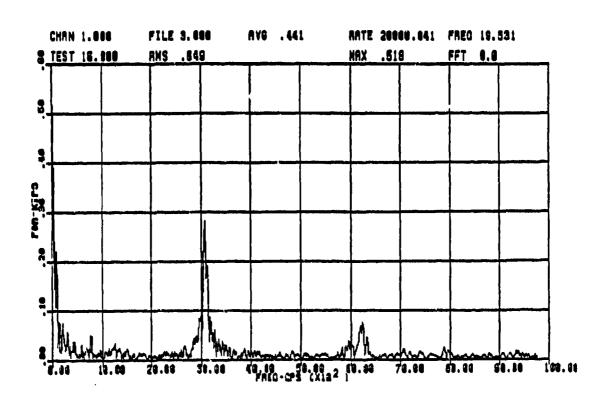
Test 16
Equipment Rack Hard-Mounted
AN/GRC-103 in Rack, Off-Line

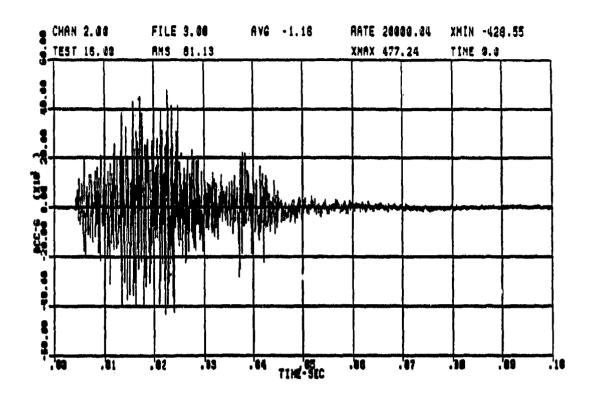


## PULSE TRAIN - TEST 16

PRECHARGE PRESSURE = 1250 PSI SYSTEM PRESSURE = 2000 PSI FLOW CONTROL #61/2



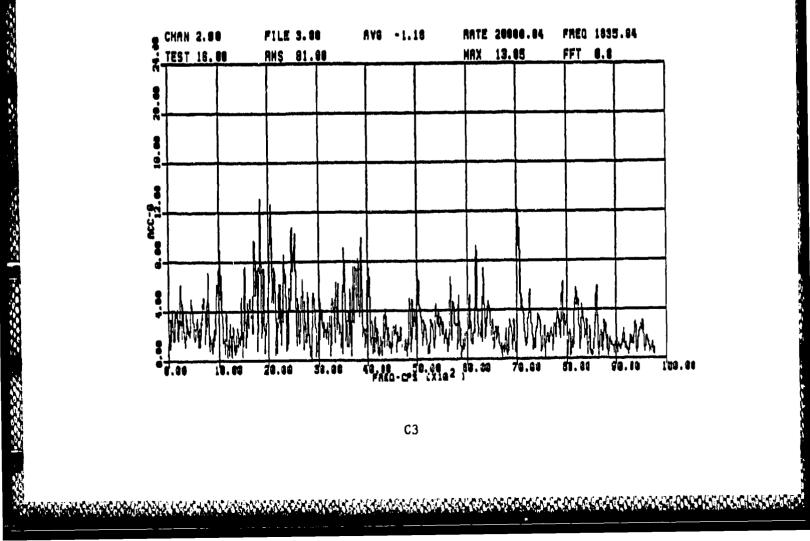


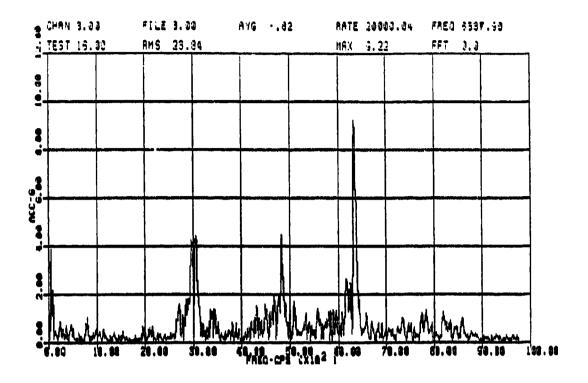


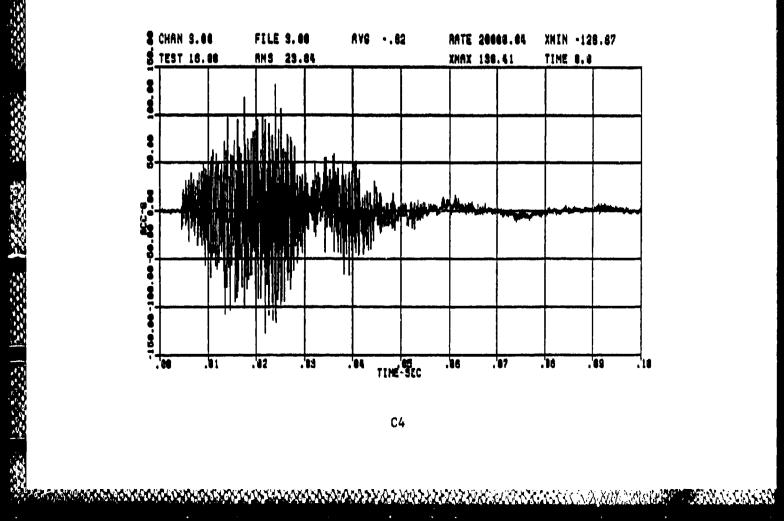
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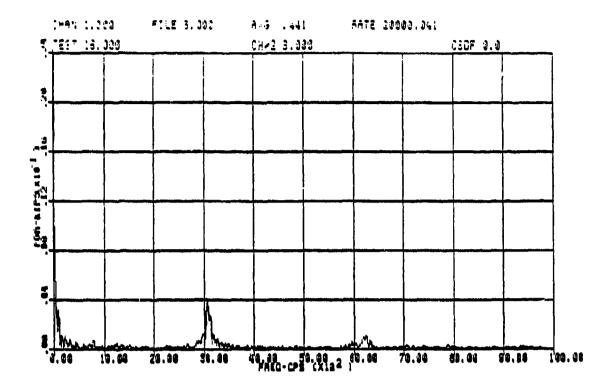
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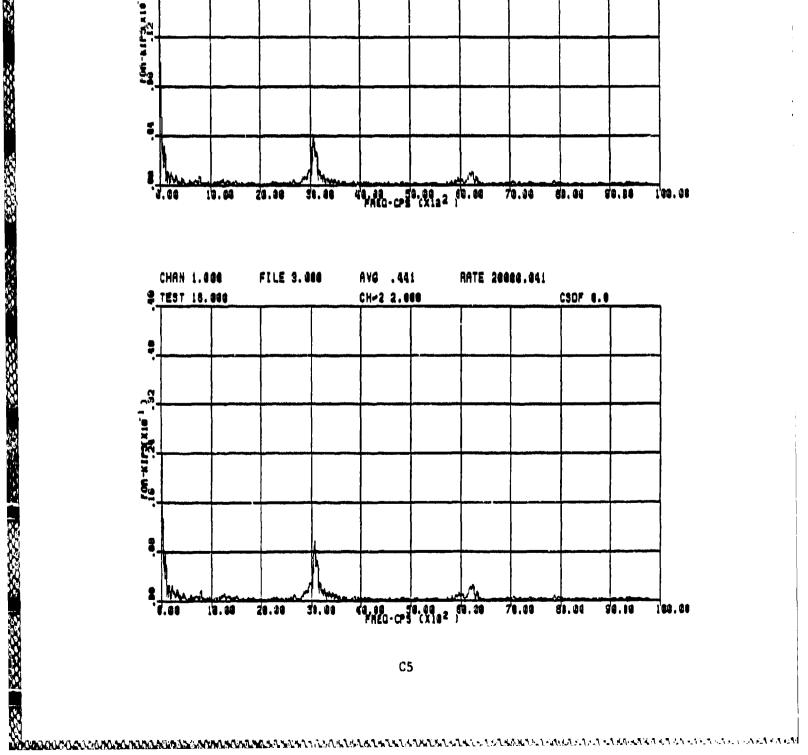
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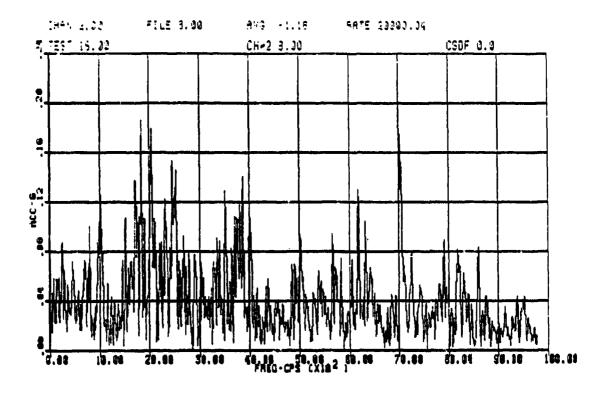


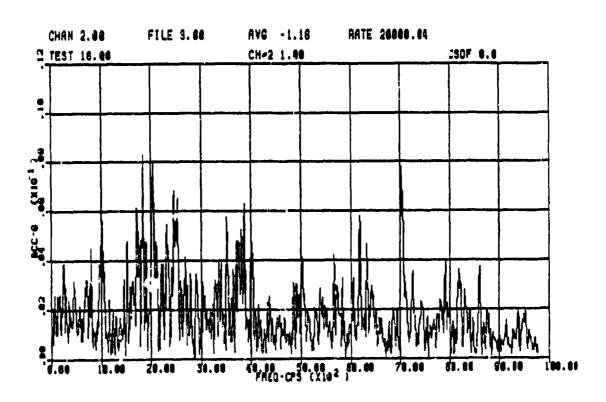




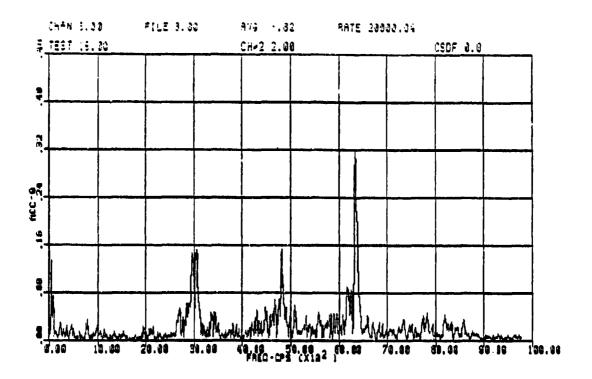


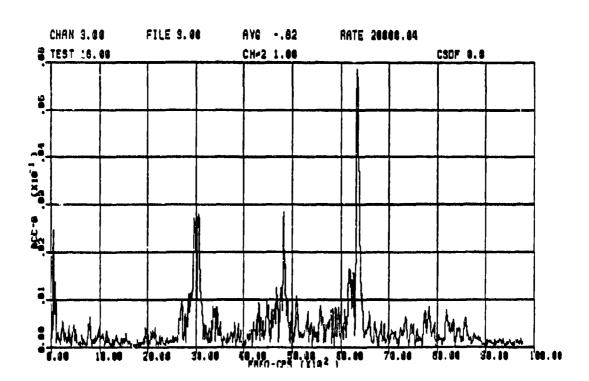






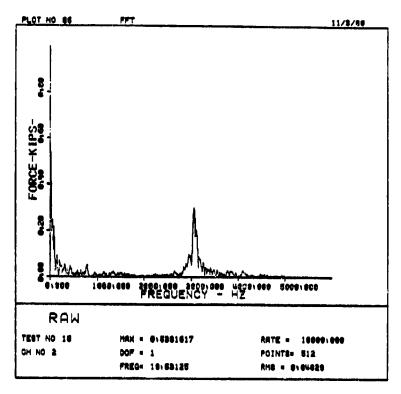
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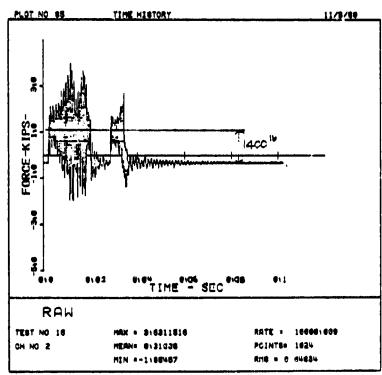


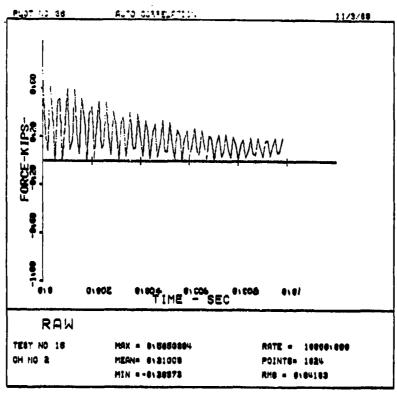


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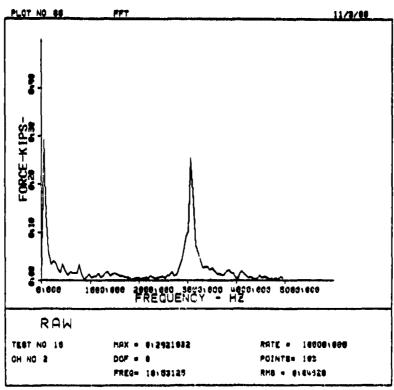
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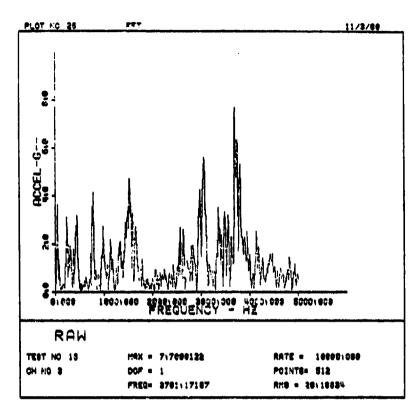






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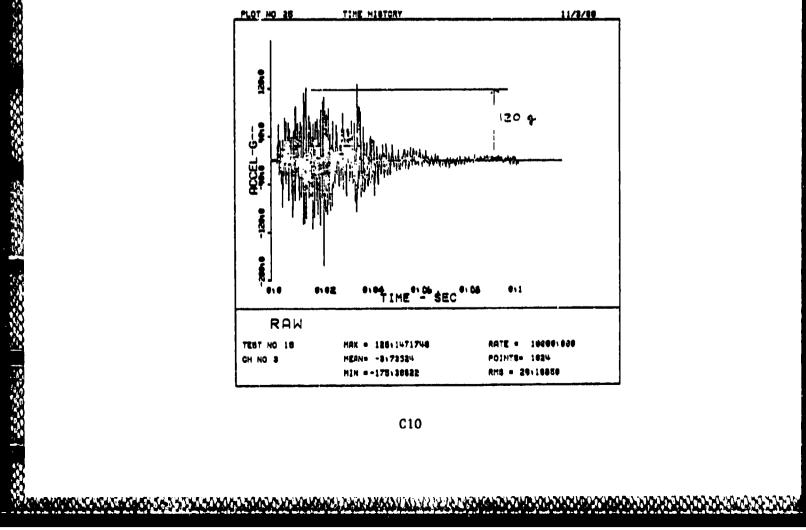


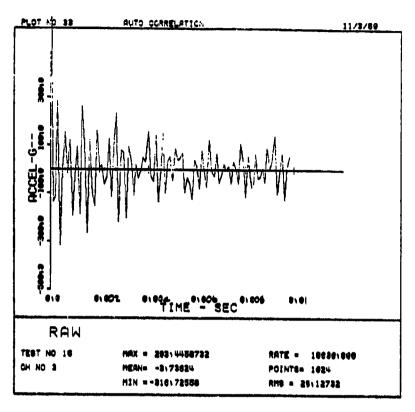


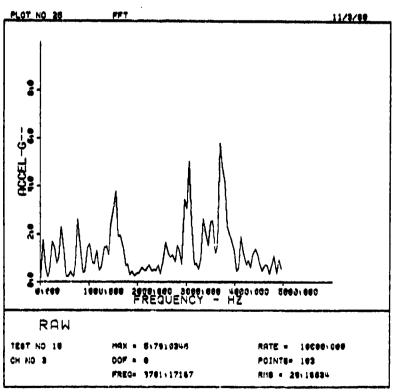
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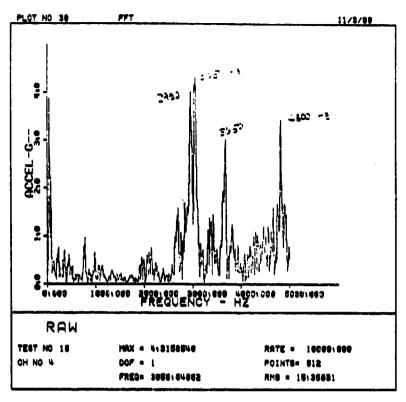
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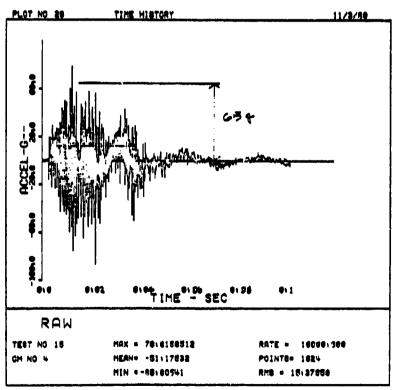
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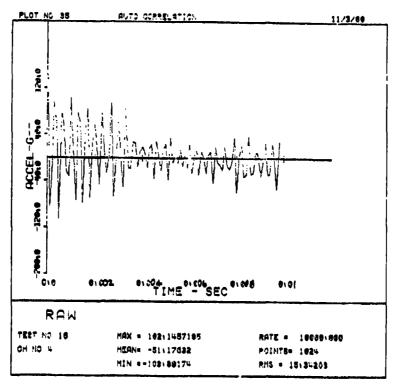


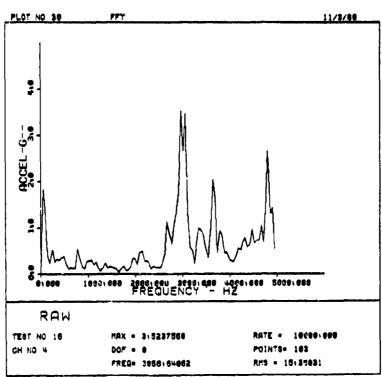


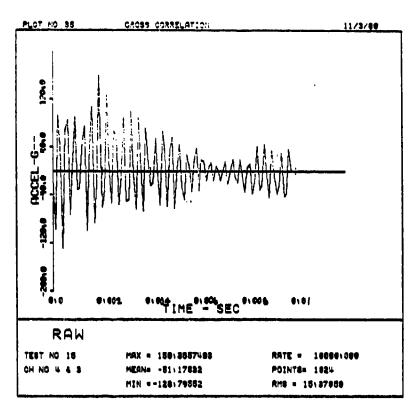


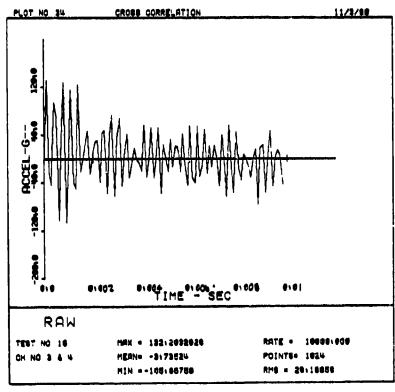


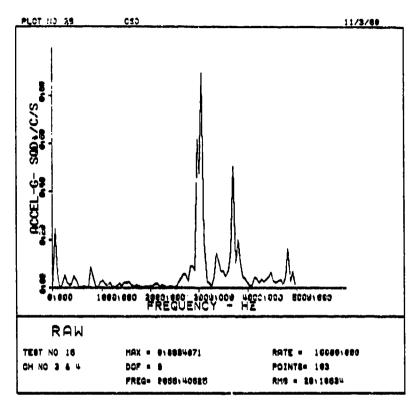


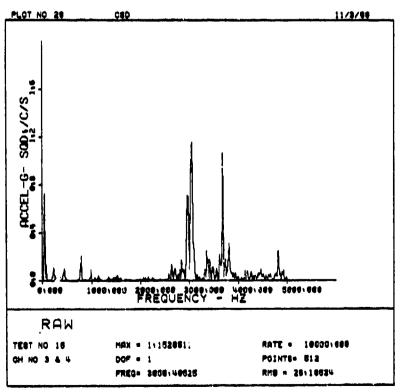






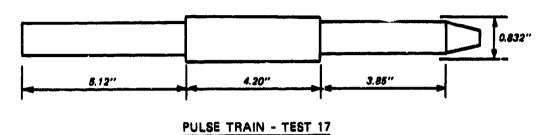






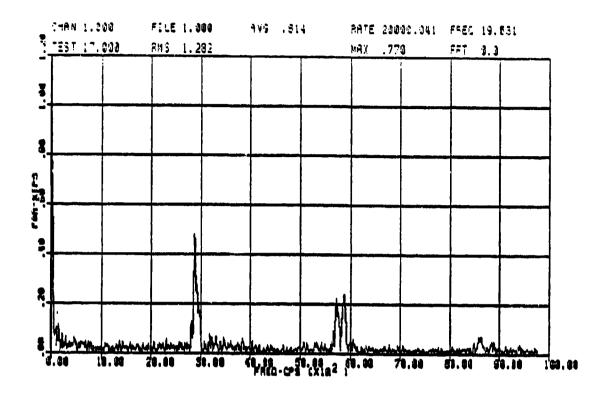
## APPENDIX D: TEST 17 DATA

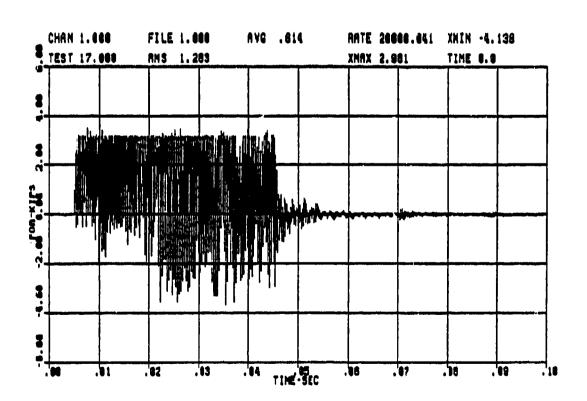
Test 17 Equipment Rack Hard-Mounted AN/GRC-103 in Rack, Off-Line

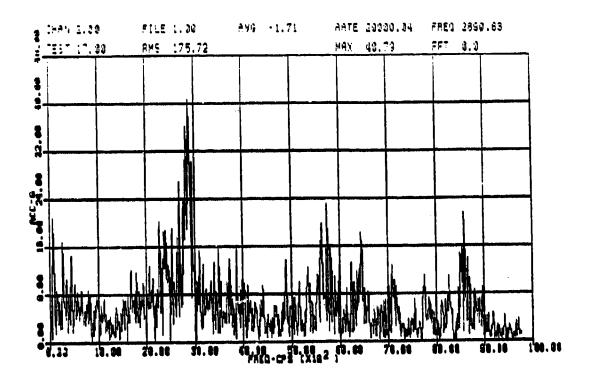


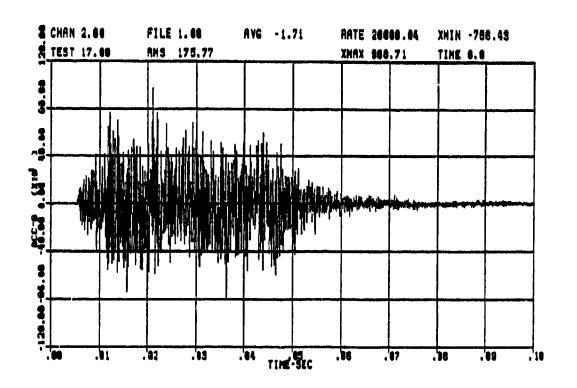
PRECHARGE PRESSURE = 1250 PSI SYSTEM PRESSURE = 2000 PSI FLOW CONTROL #6%

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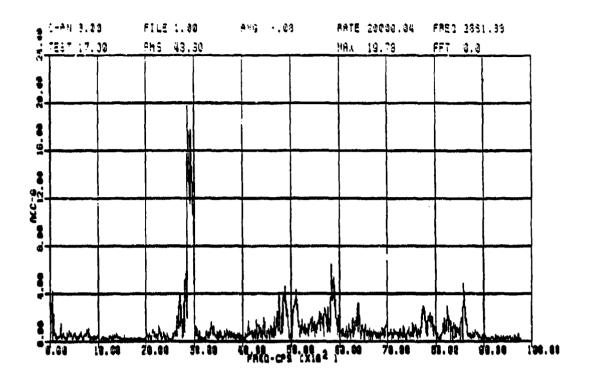




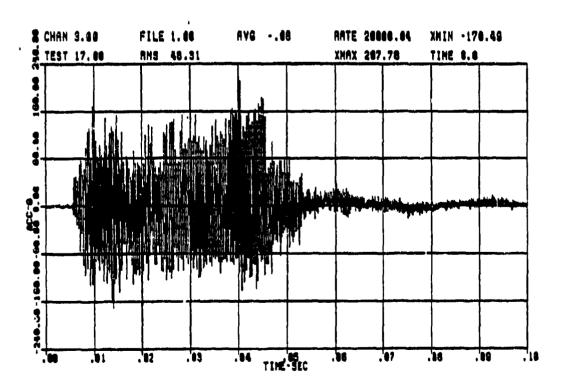


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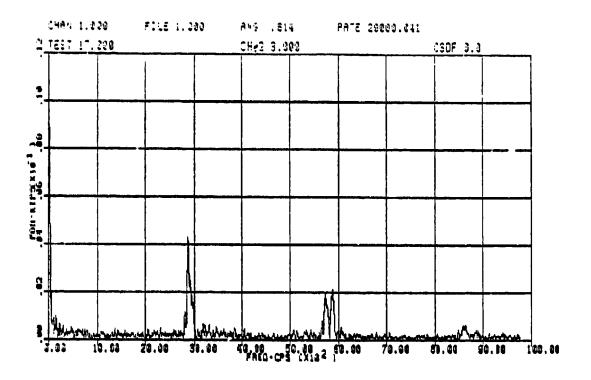
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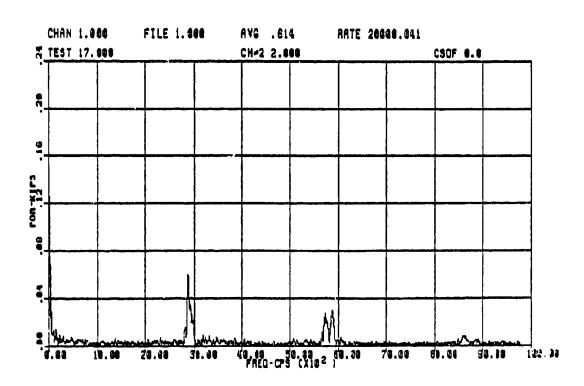


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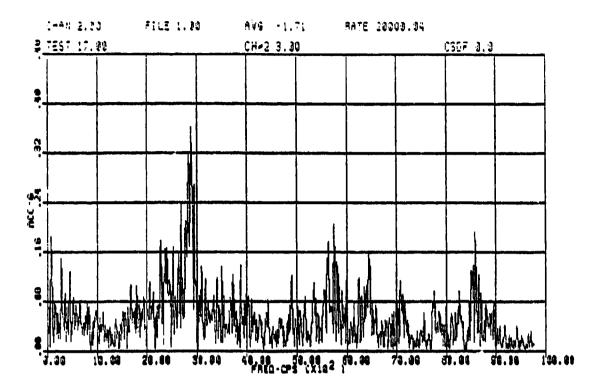


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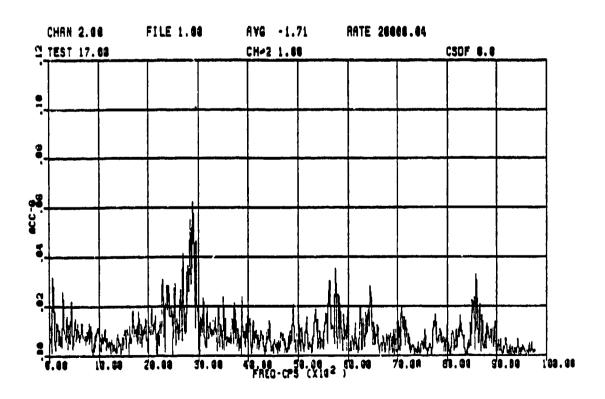


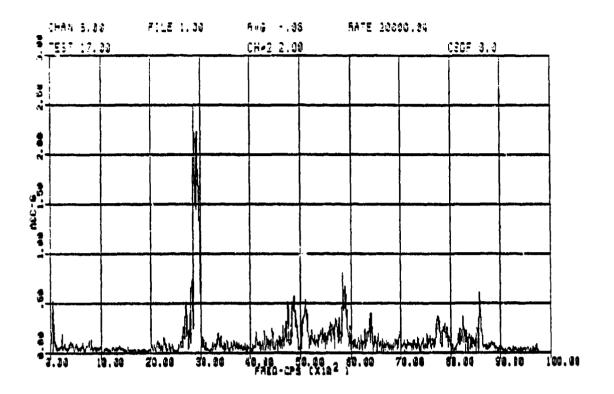


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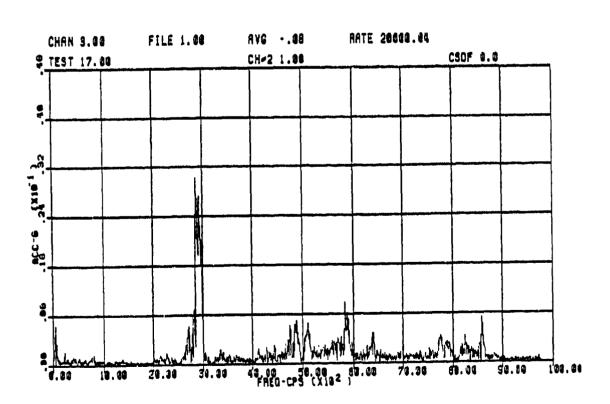


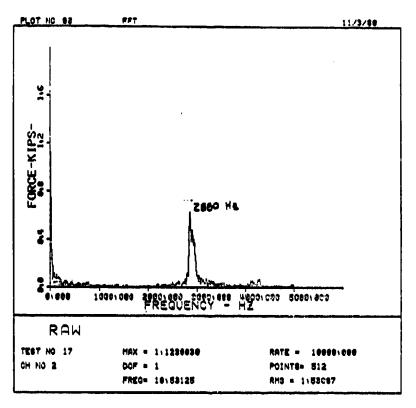
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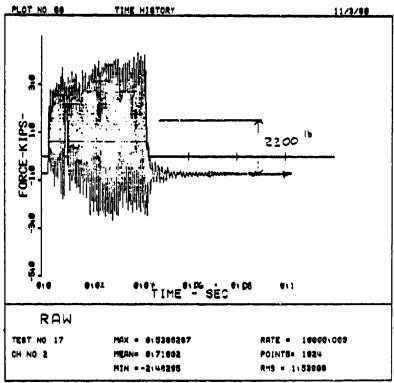


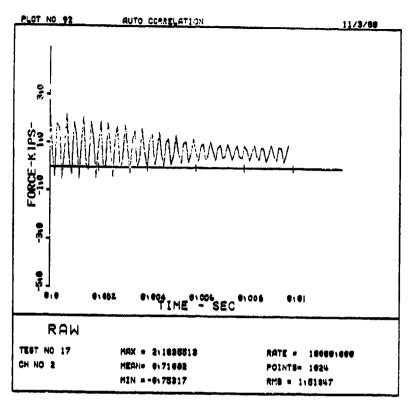


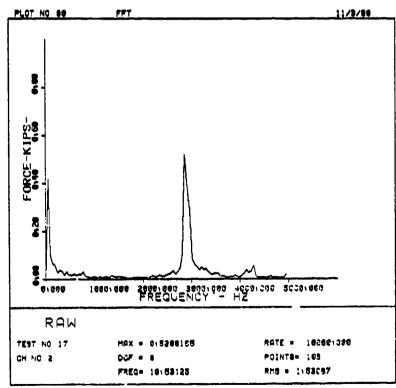
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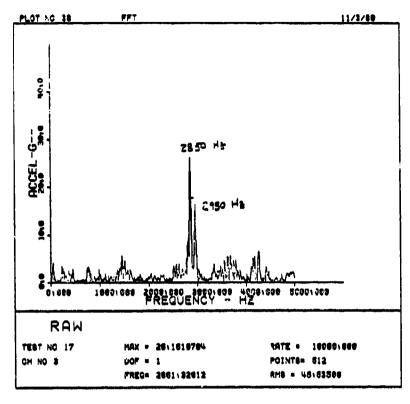


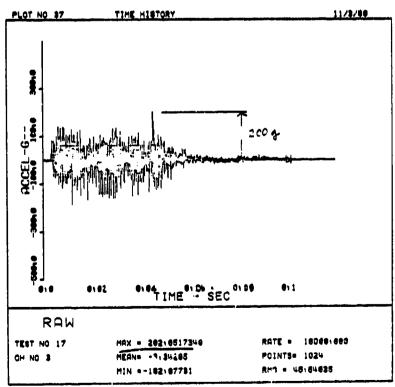


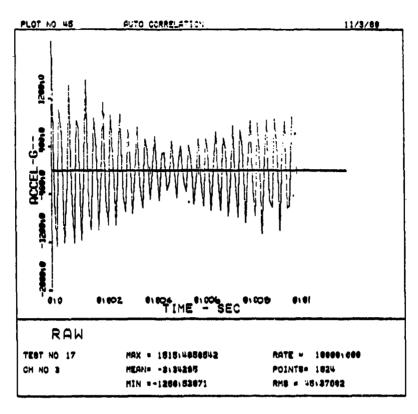


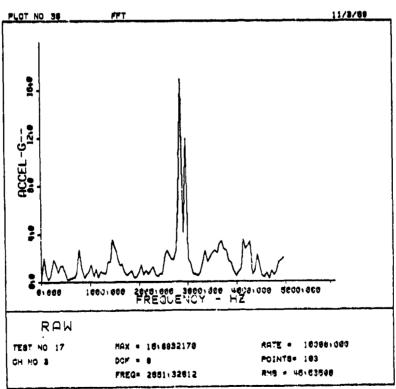




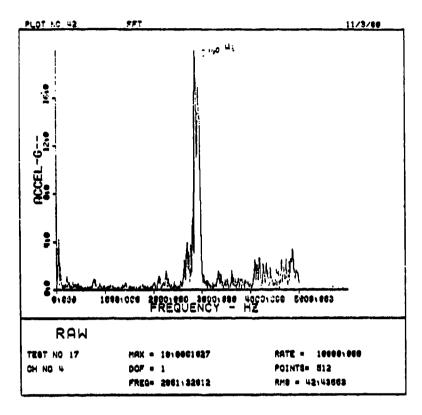






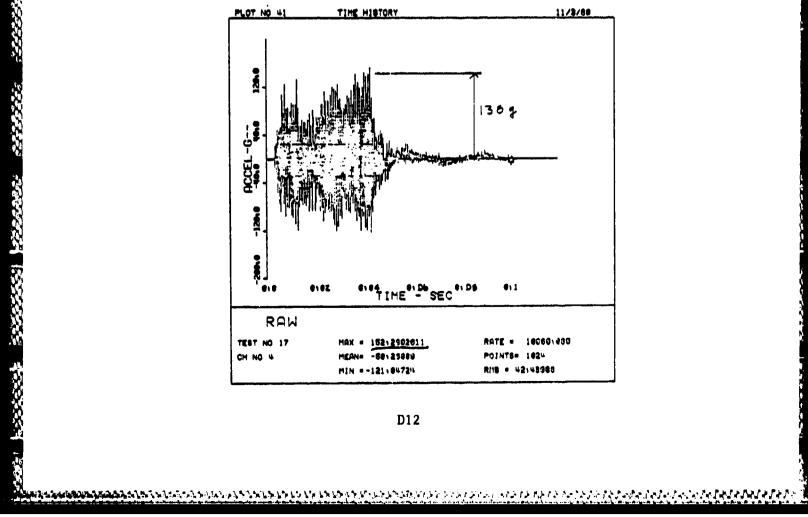


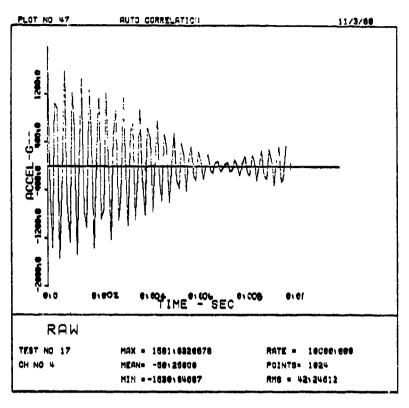
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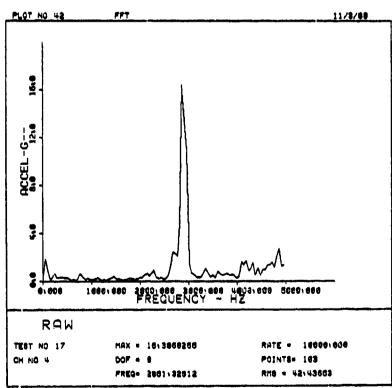


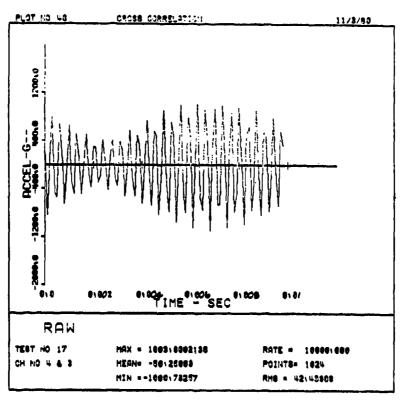
Section 1

Service States

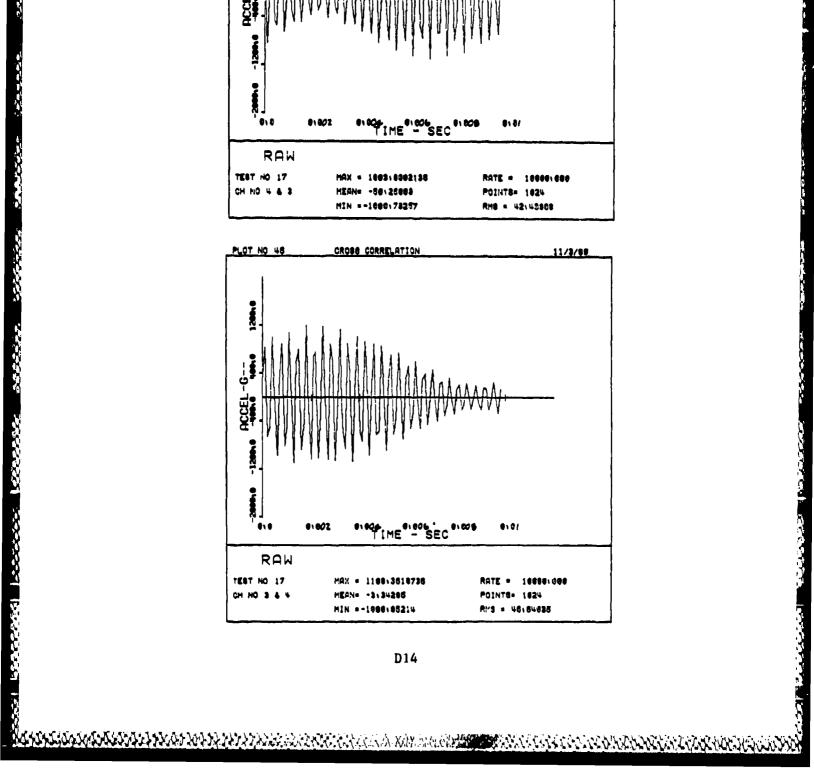


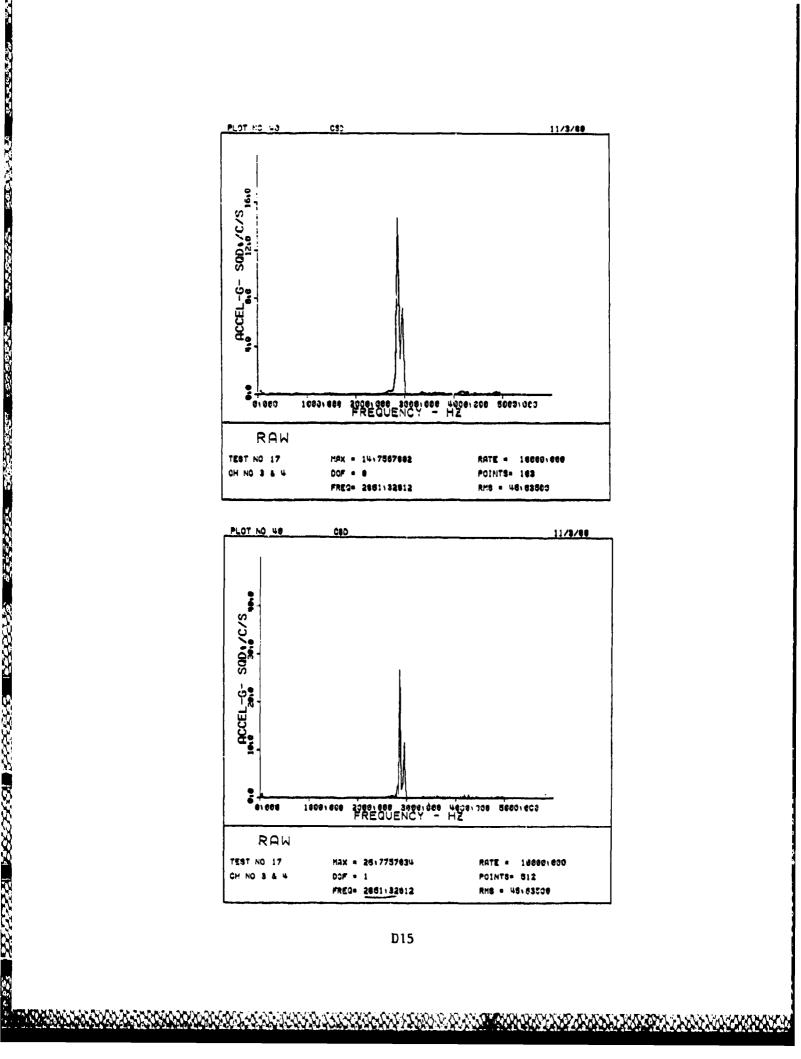


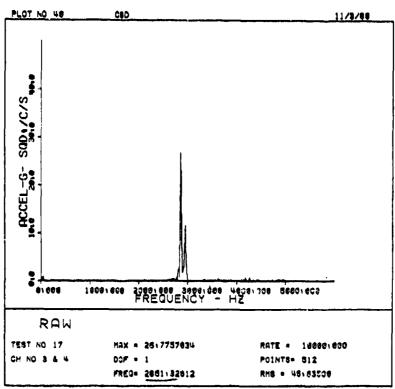




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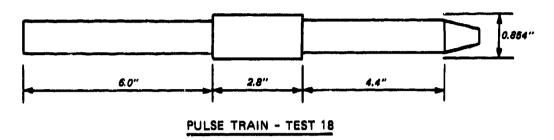




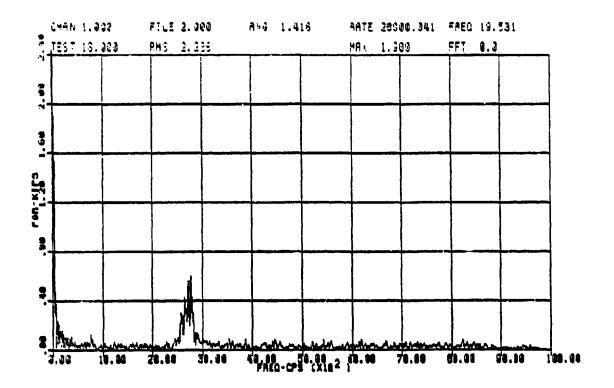


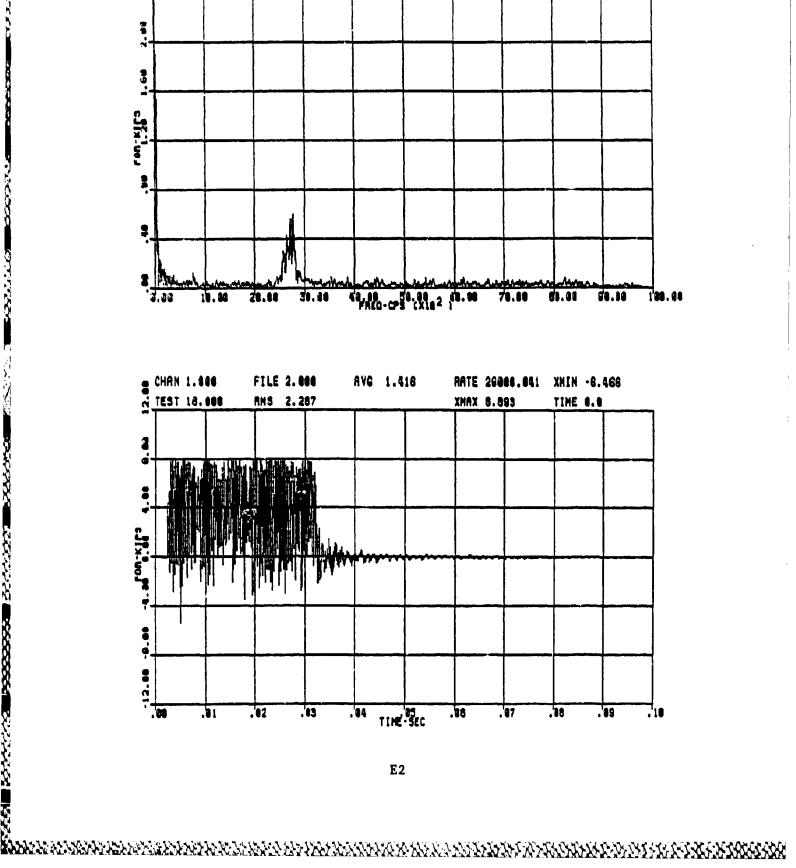
## APPENDIX E: TEST 18 RESULTS

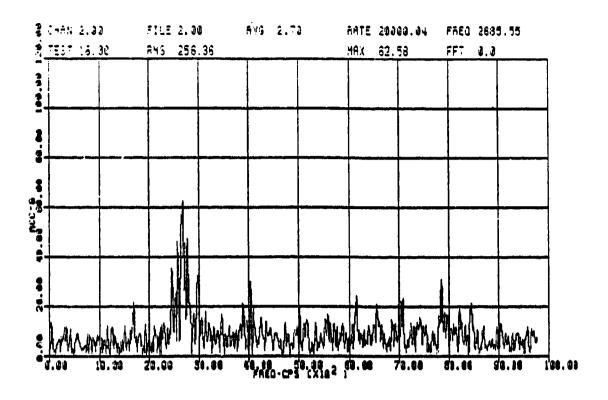
Test 18
Equipment Rack Hard-Mounted
AN/GRC-103 in Rack, On-Line

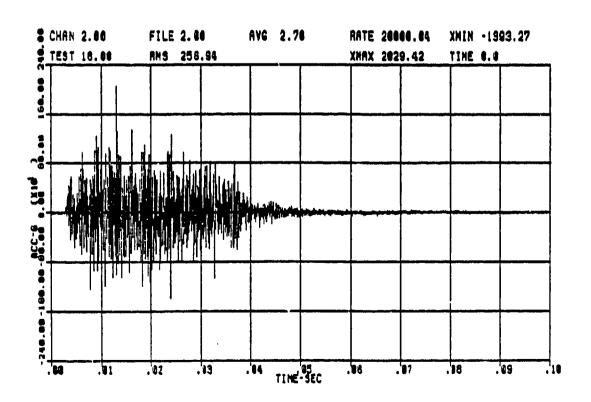


PRECHARGE PRESSURE = 1250 PSI SYSTEM PRESSURE = 2000 PSI FLOW CONTROL #8½

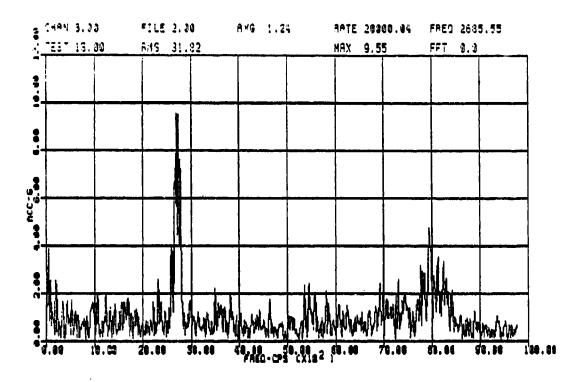


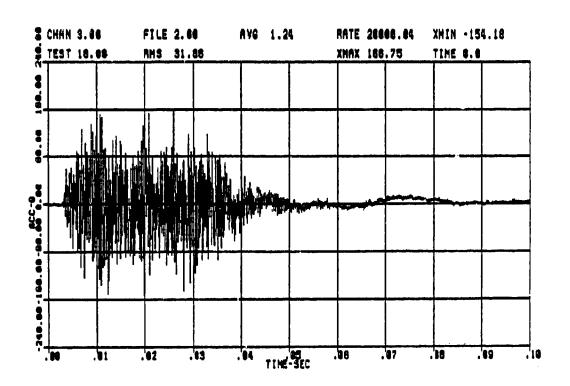


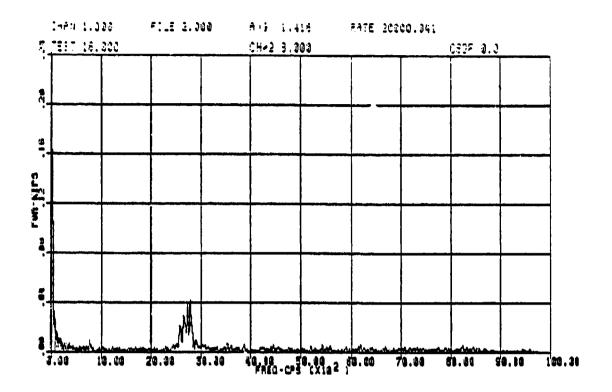


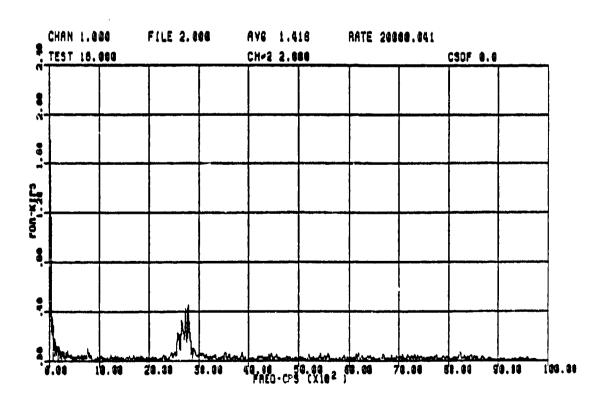


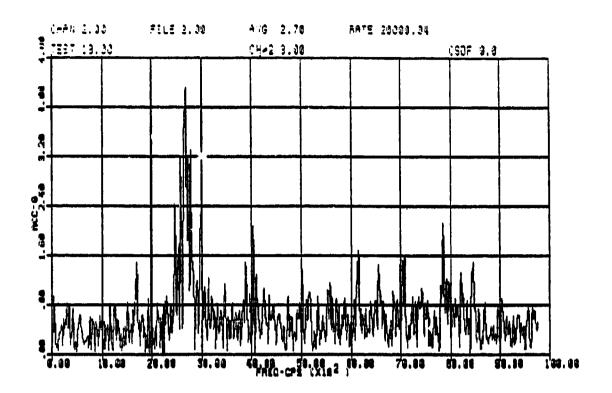
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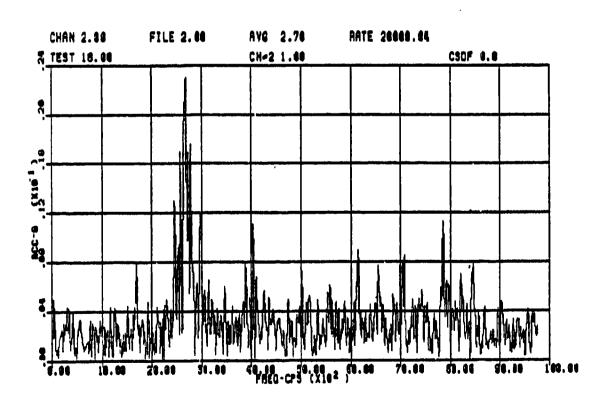


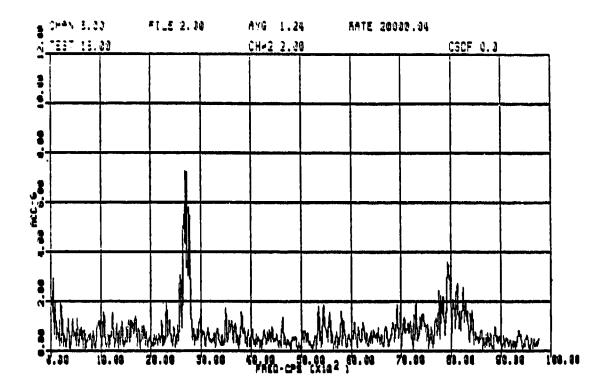


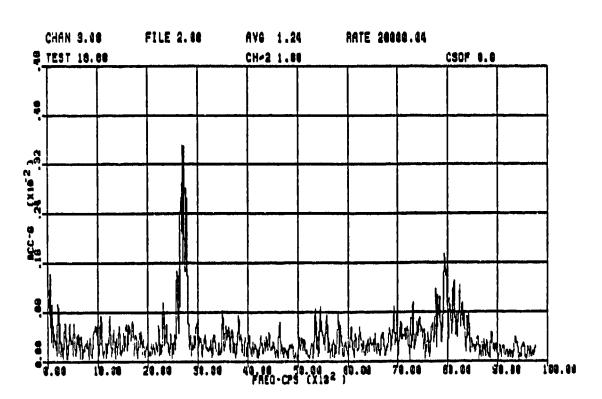
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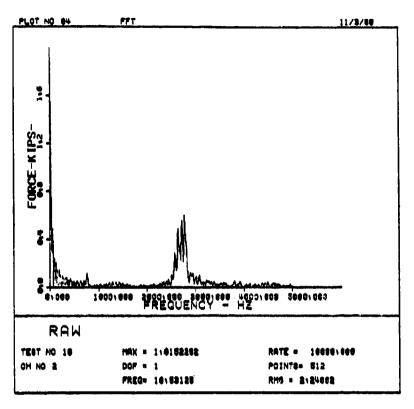
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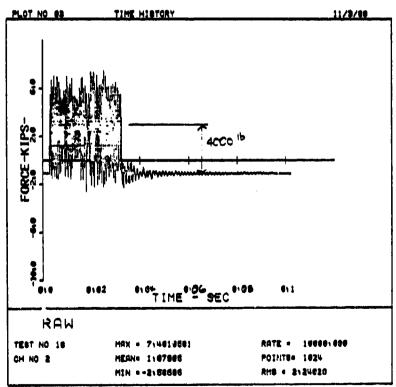
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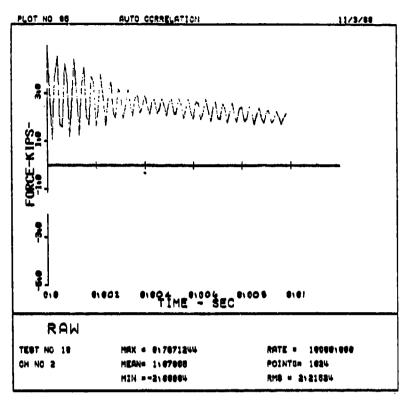




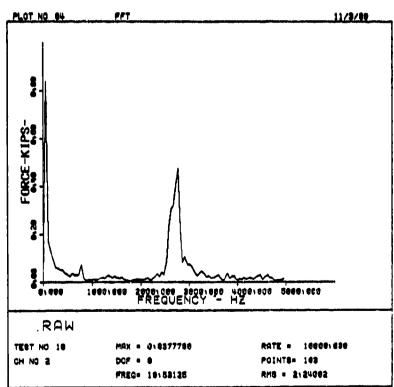




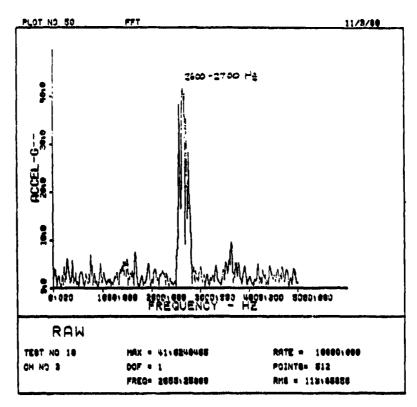


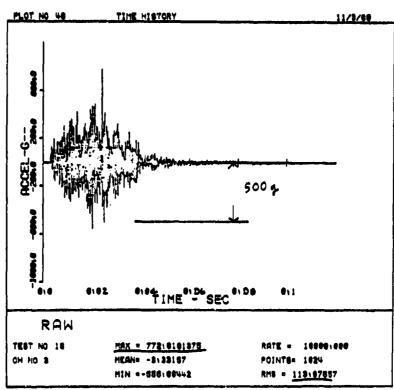


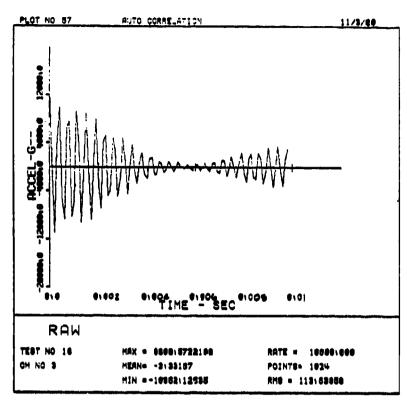
**经验证据** 

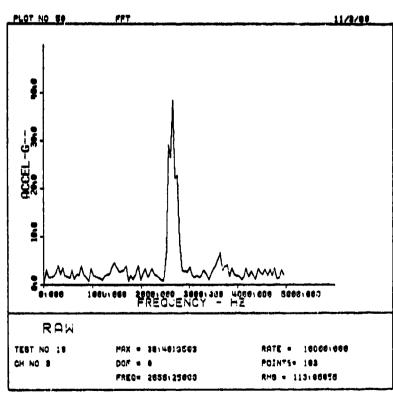


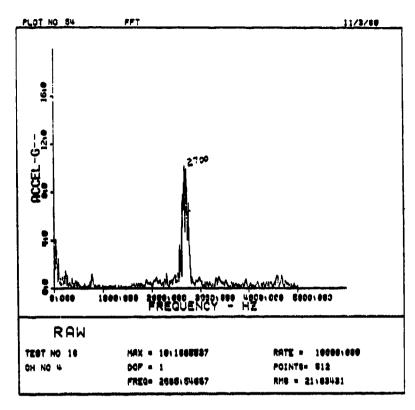
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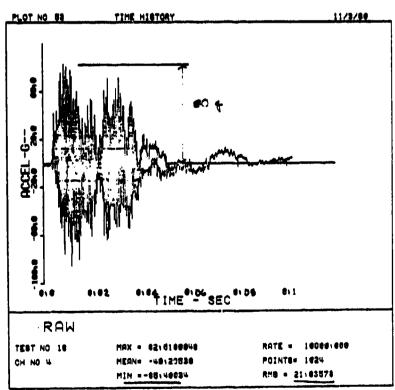


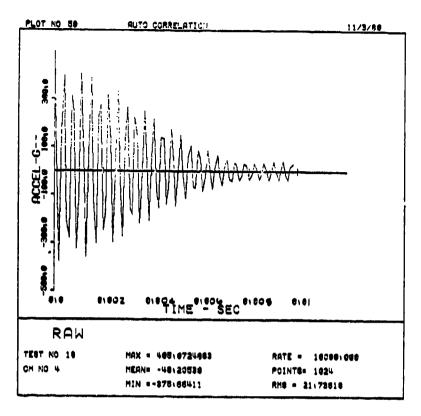










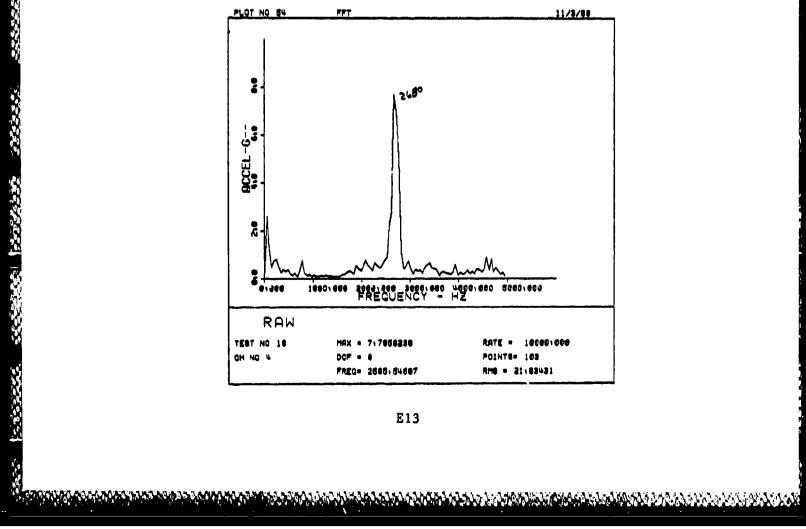


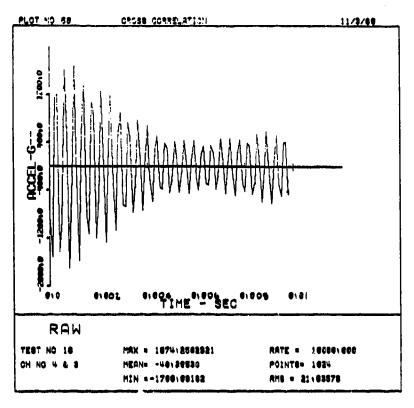
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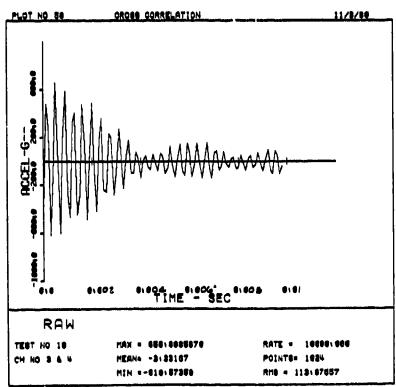
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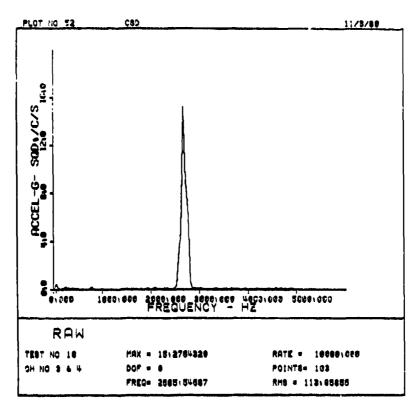
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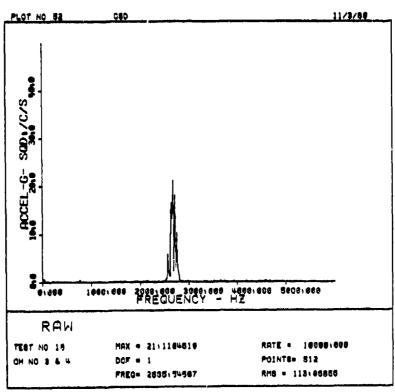






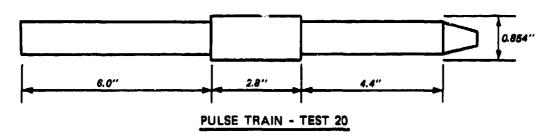


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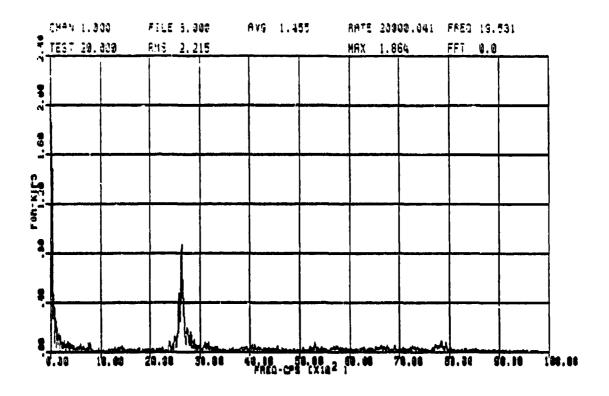
## APPENDIX F: TEST 20 RESULTS

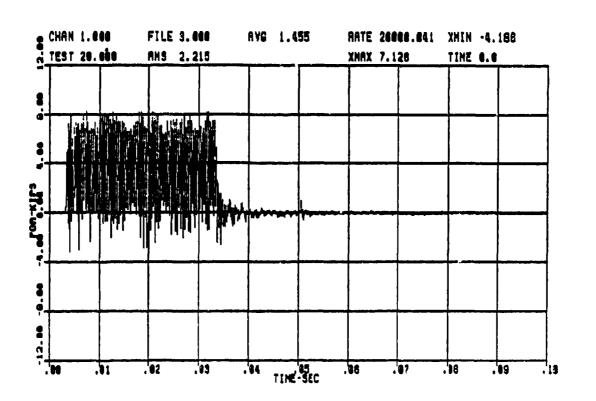
Test 20 Equipment Rack Hard-Mounted TD660 in Rack, On-Line

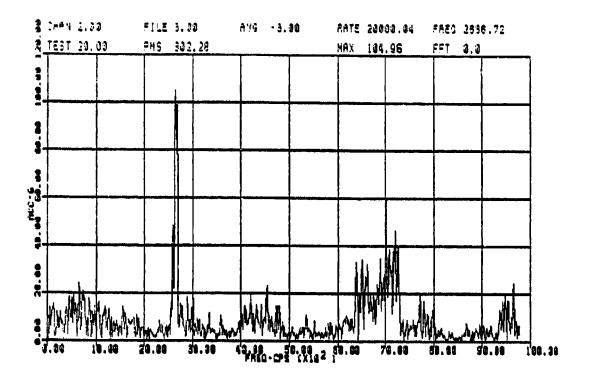


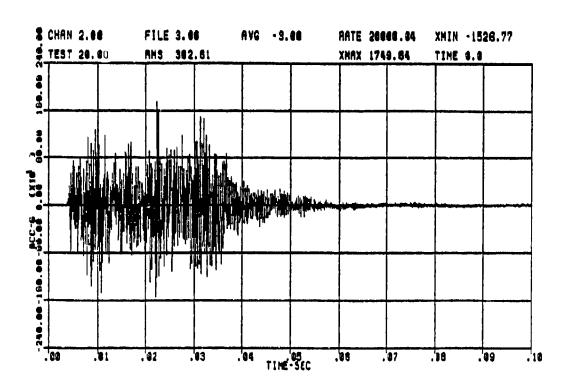
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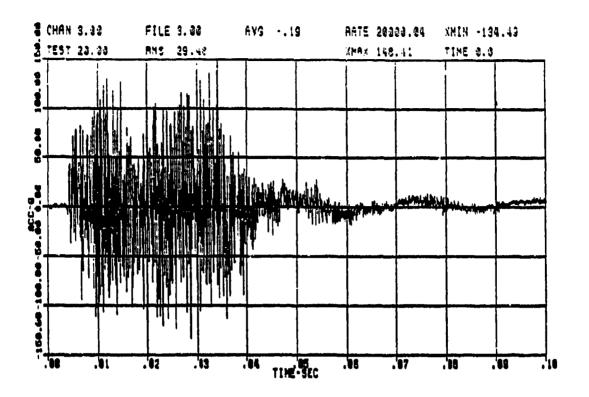
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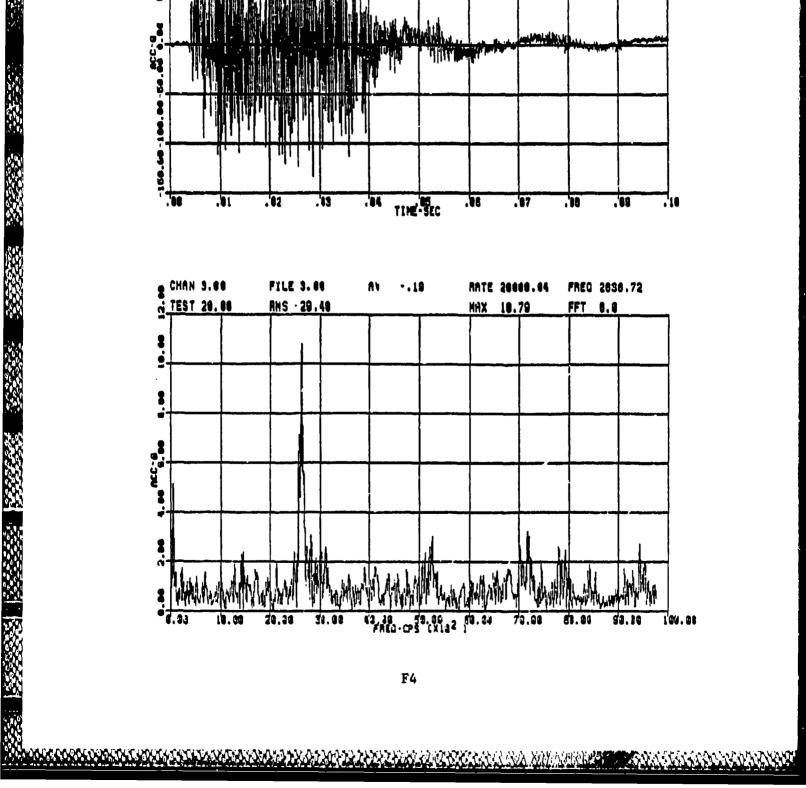


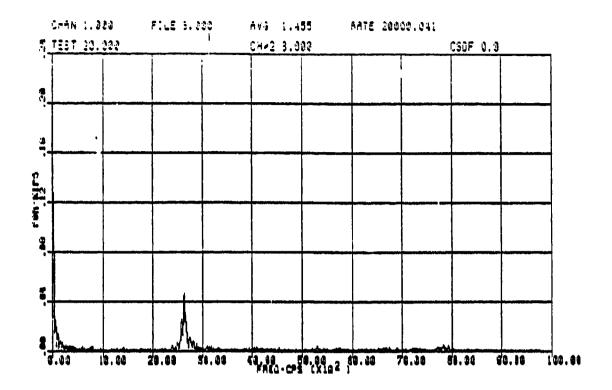


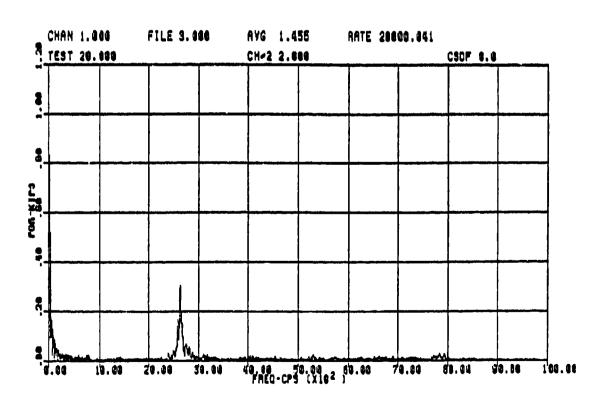


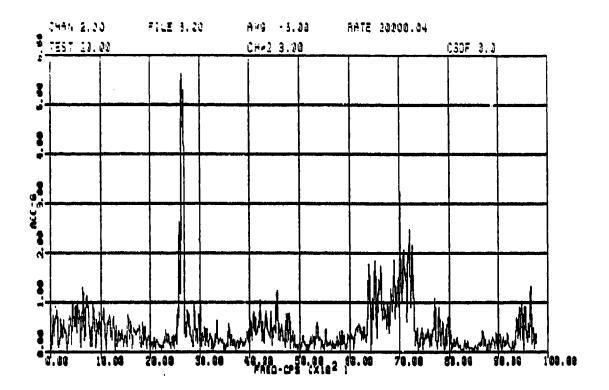


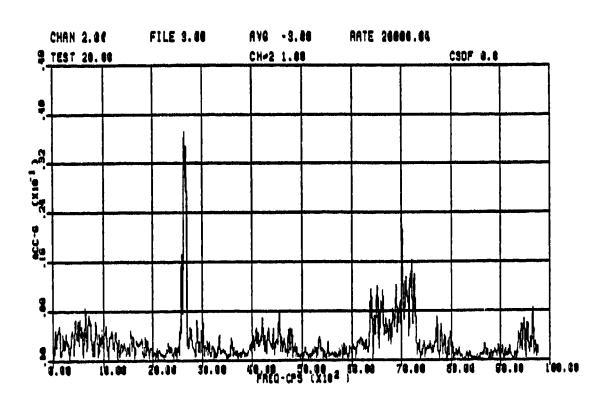


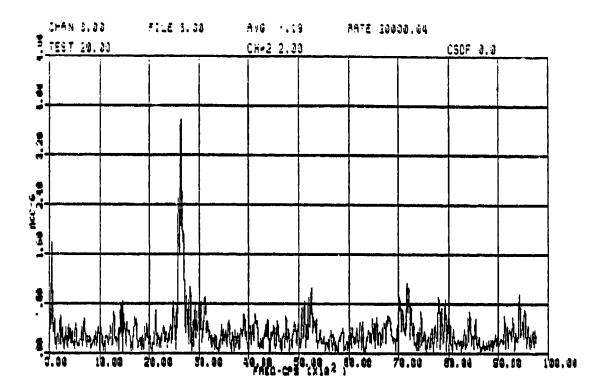


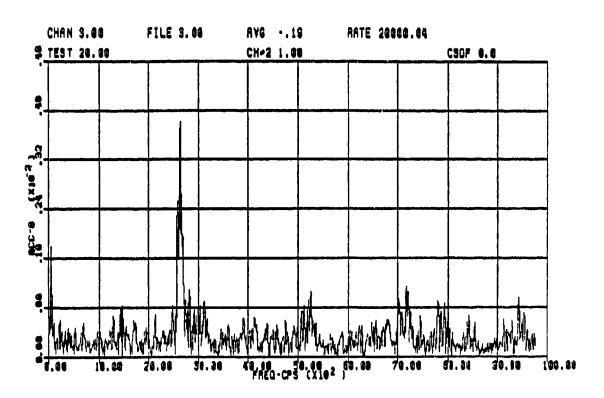


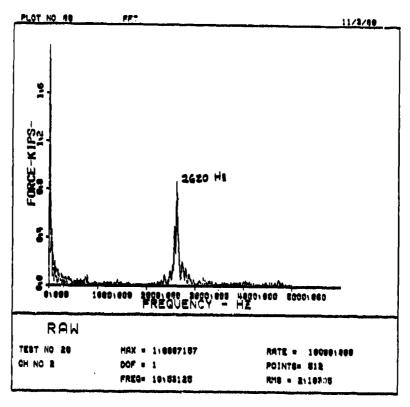


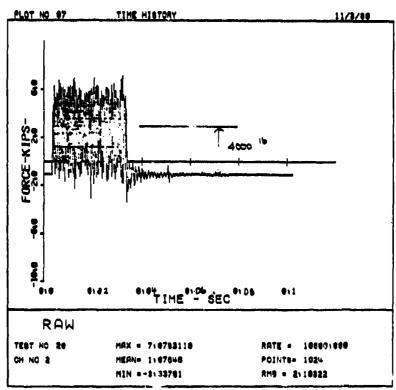




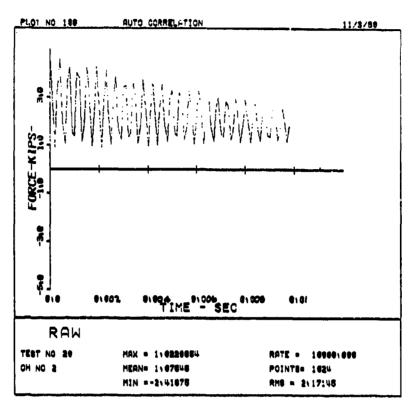


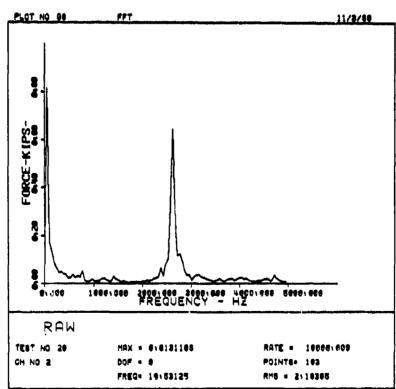




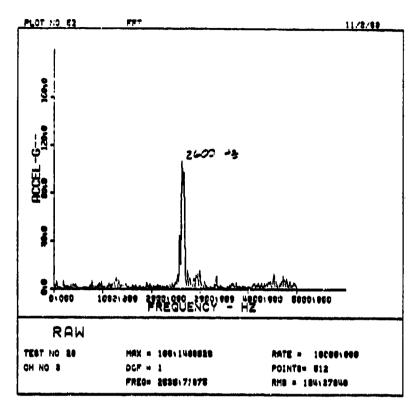


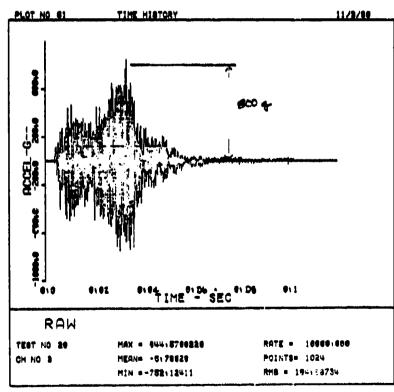
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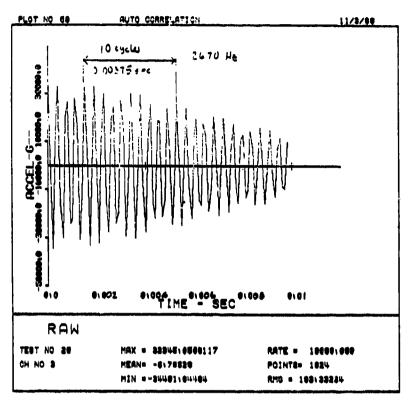


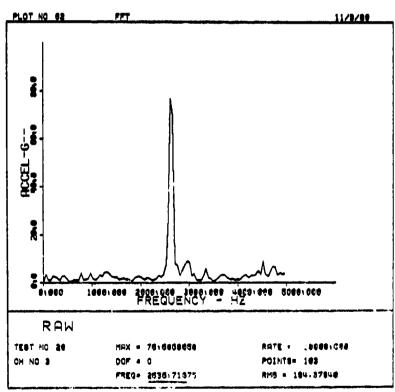
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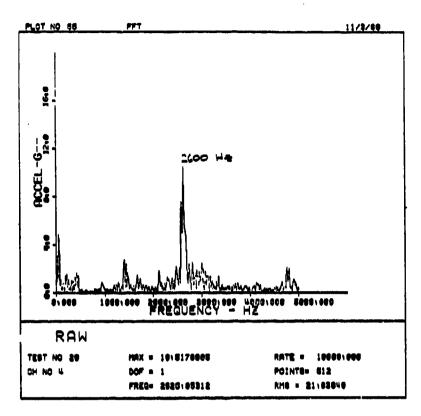




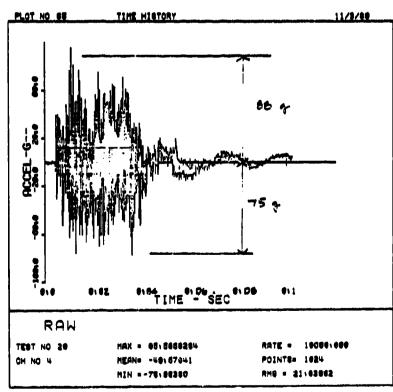
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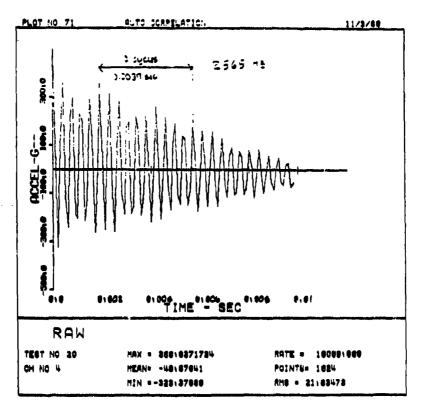




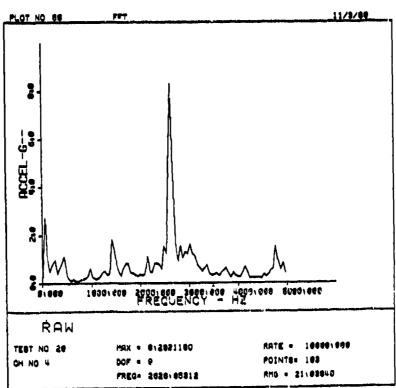


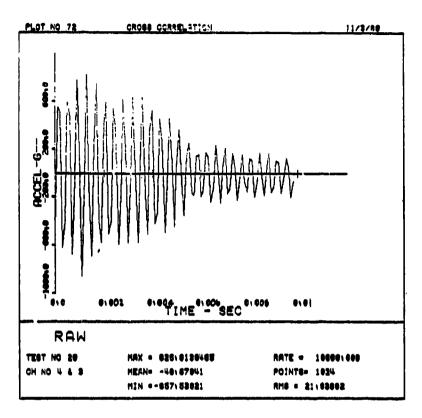
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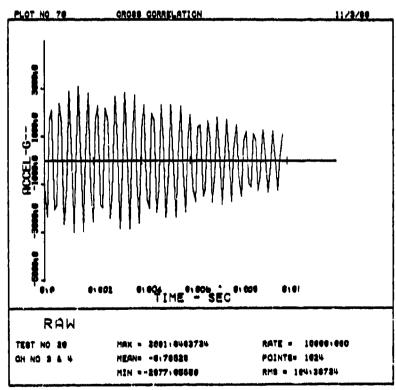


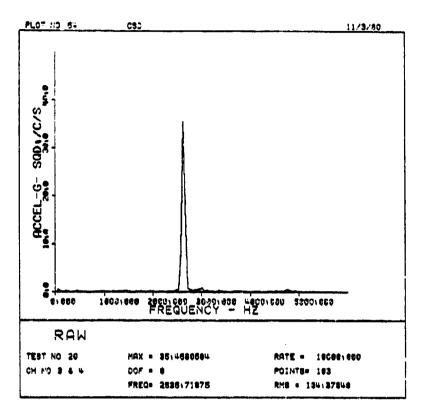


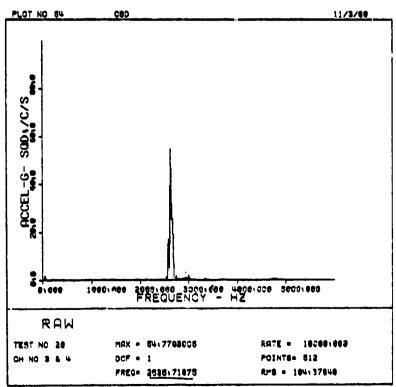
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## APPENDIX G: TEST 21 RESULTS

Test 21
Equipment Rack Soft-Mounted
AN/GRC-103 in Rack, Off-Line

Plot Heading Nomenclature (Tests 21-25)

CHAN 1 - Input force

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CHAN 2 - Rack acceleration

CHAN 3 - Equipment acceleration

RMS, AVG, MAX, XMAX, XMIN - RMS, average, maximum, and minimum values of data plots

FREQ - Frequency at which maximum value occurs

FFT 0.0 - Fast Fourier transform of time history

TIME 0.0 - Time history record

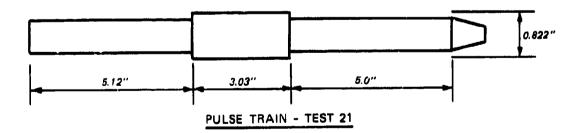
CSDF 0.0 - Cross spectral density function

CH/2 1.0 Channel number of the second channel for plots involving

CH/2 2.0 two-channel functions (CSDF)

CH/2 3.0

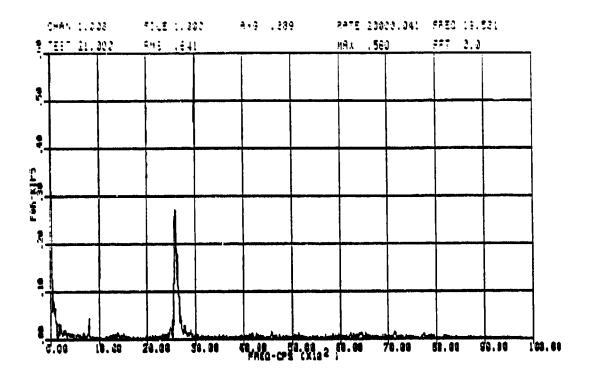
RATE - Digitizing rate, samples per second

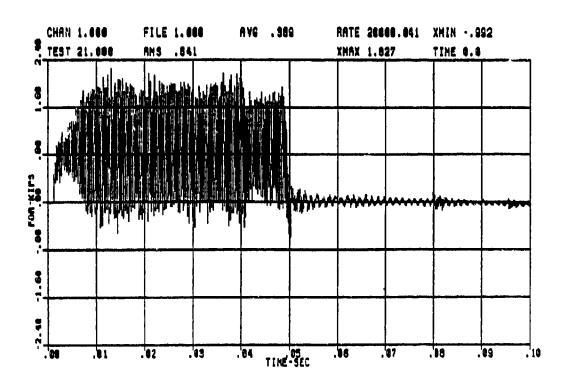


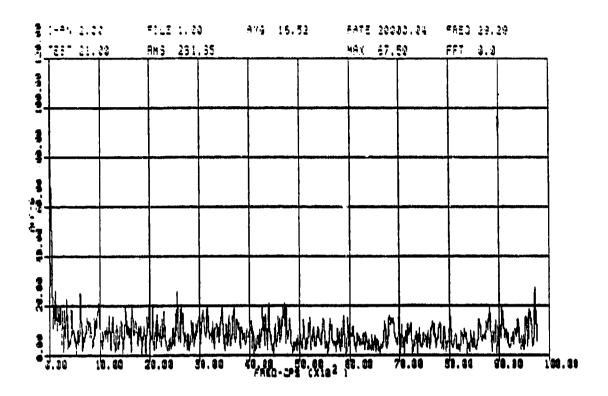
PRECHARGE PRESSURE = 1250 PSI SYSTEM PRESSURE = 2000 PSI

FLOW CONTROL #6%

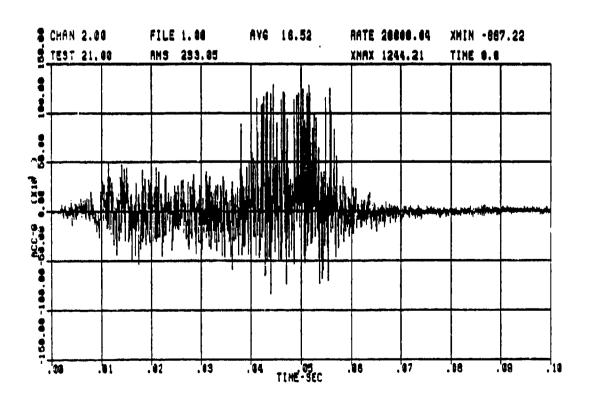
AIR BAG PRESSURE = 50 PSI



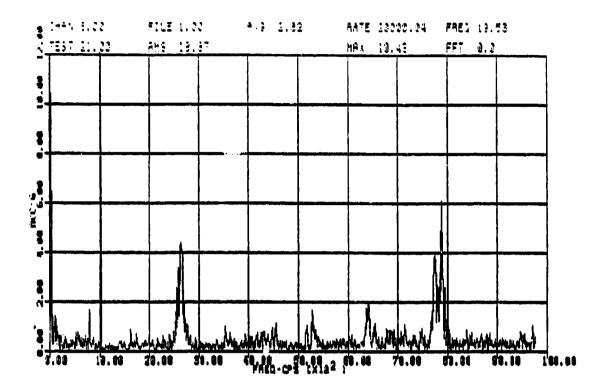


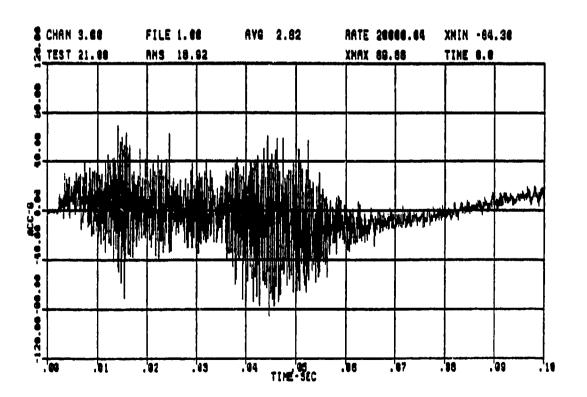


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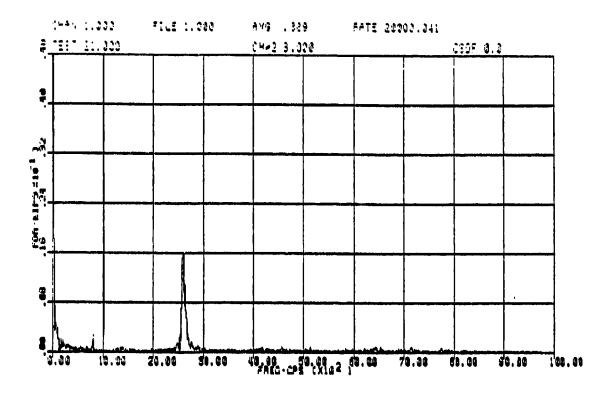


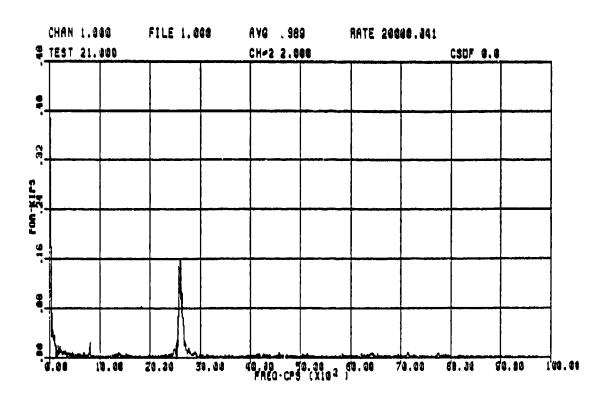
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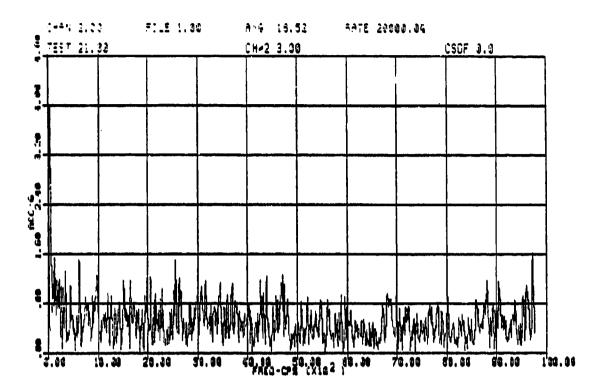


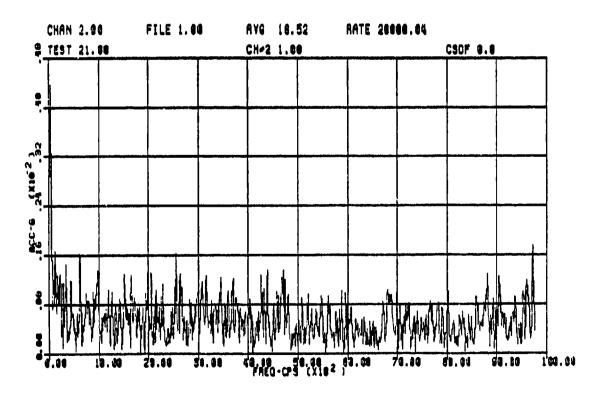
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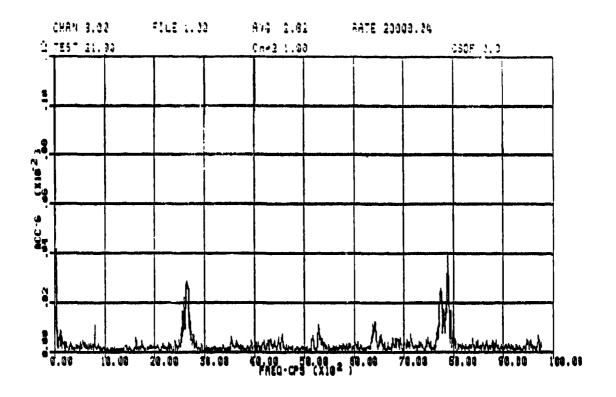


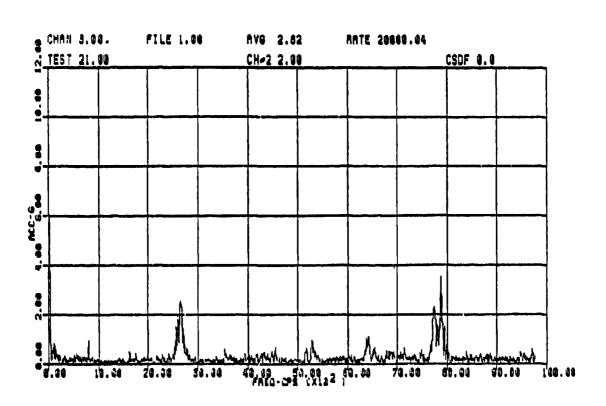


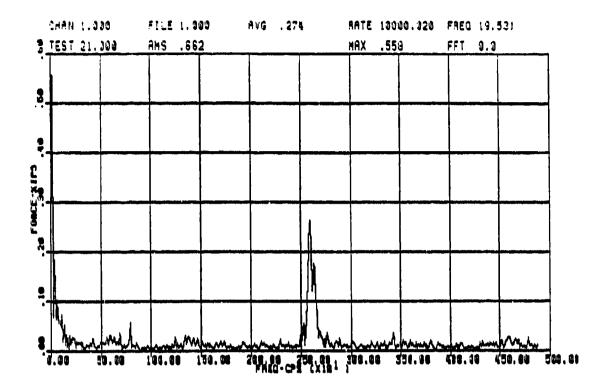
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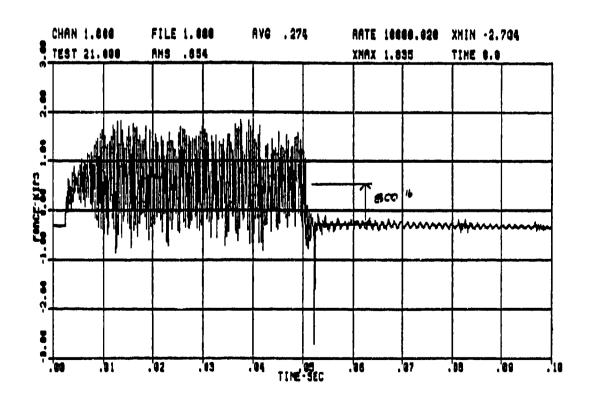


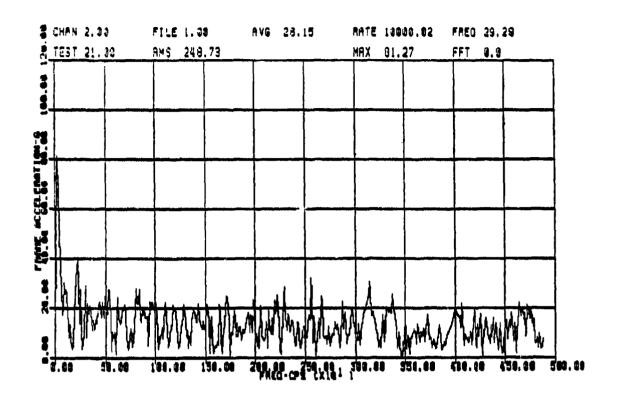


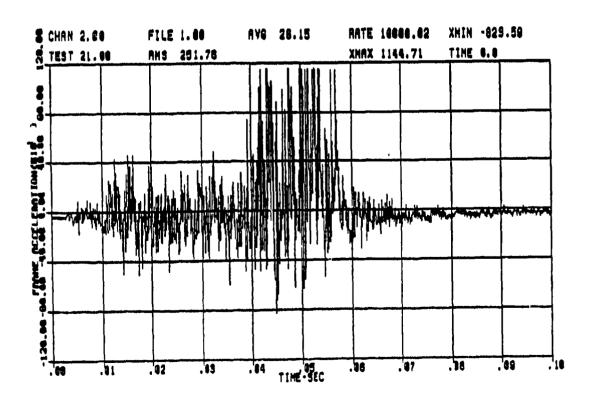


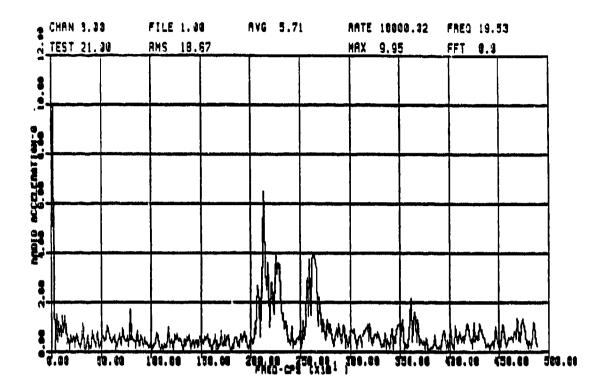


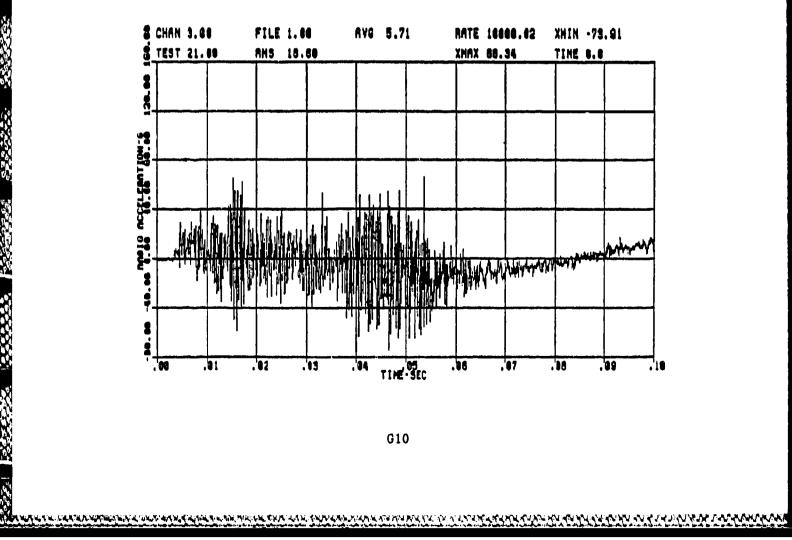


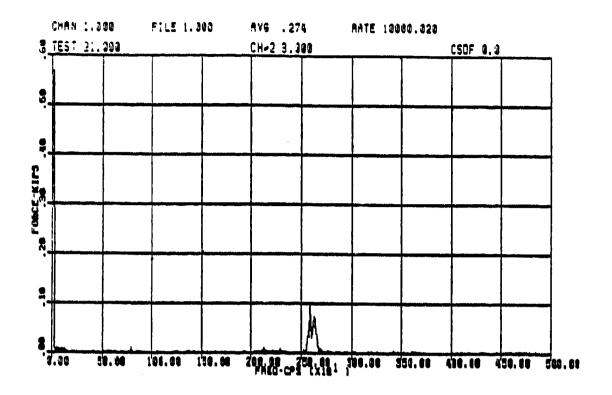


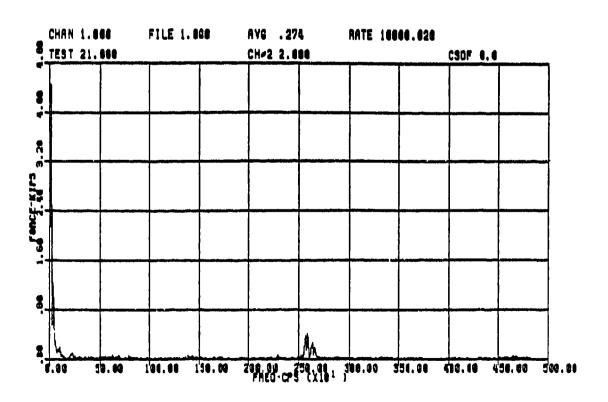


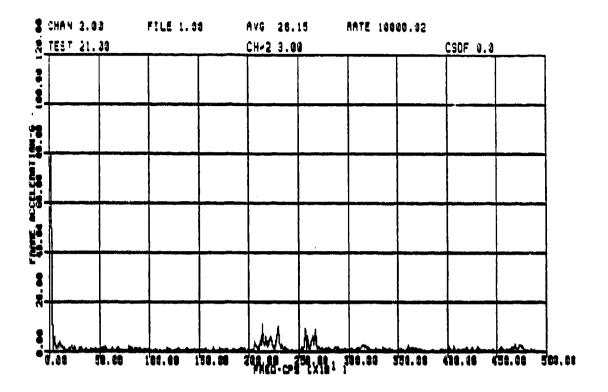




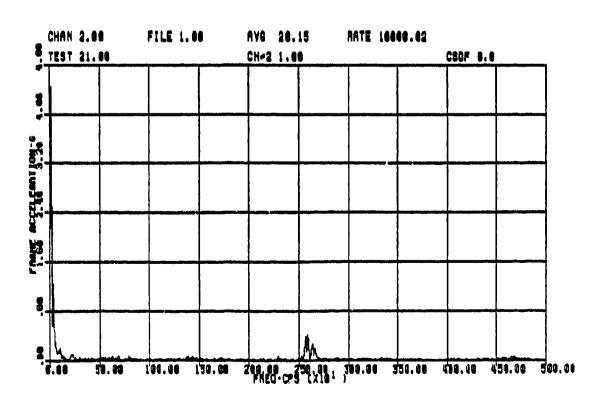




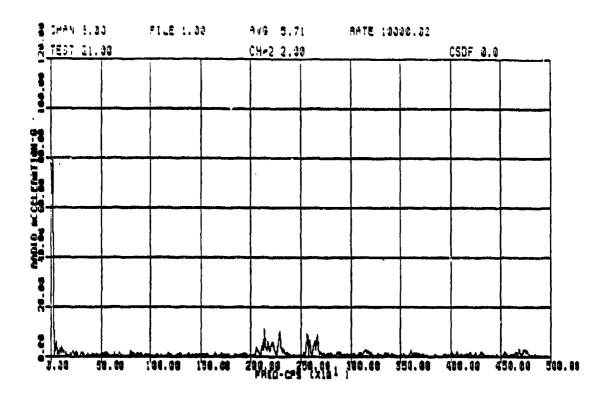


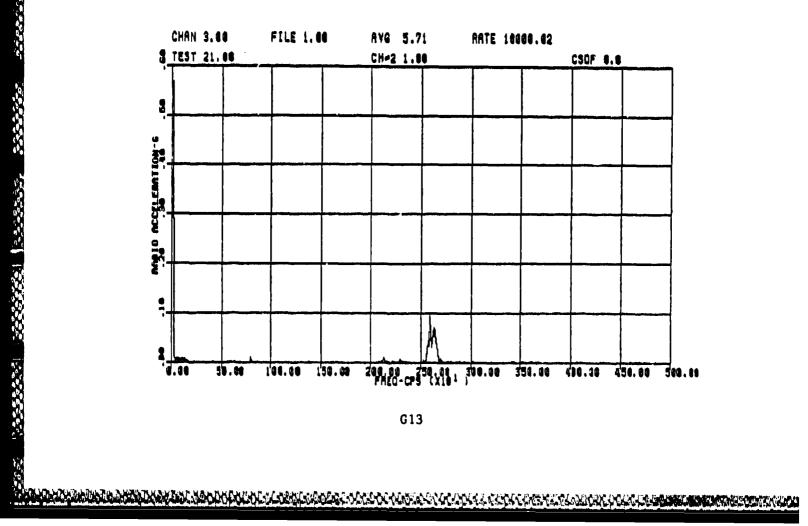


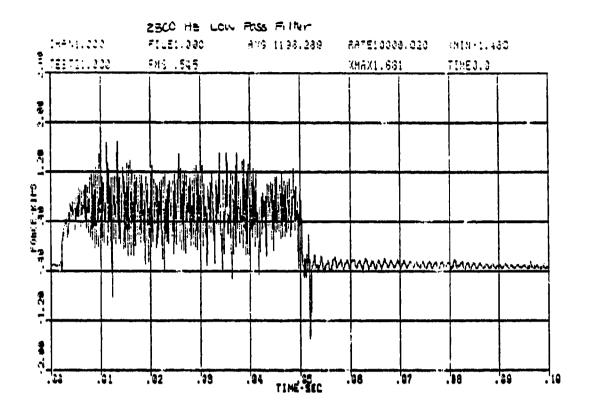
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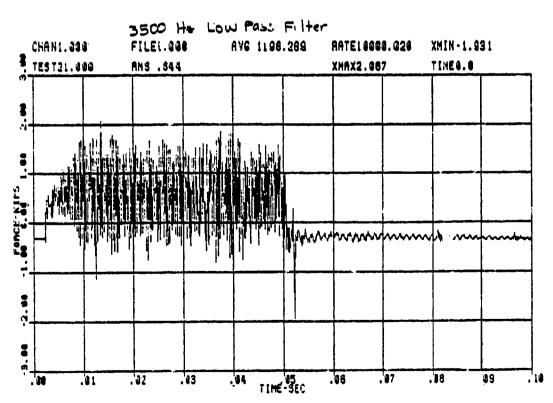


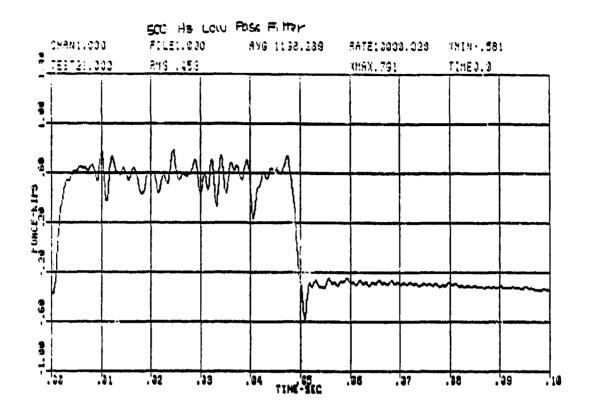


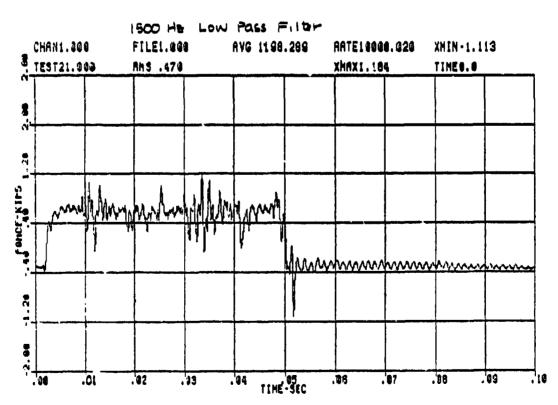


Section 1

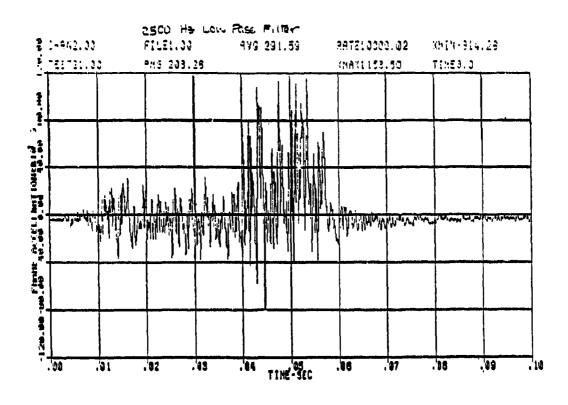
33.53.34 | 33.53.55 | 32.53.55 | 32.53.55

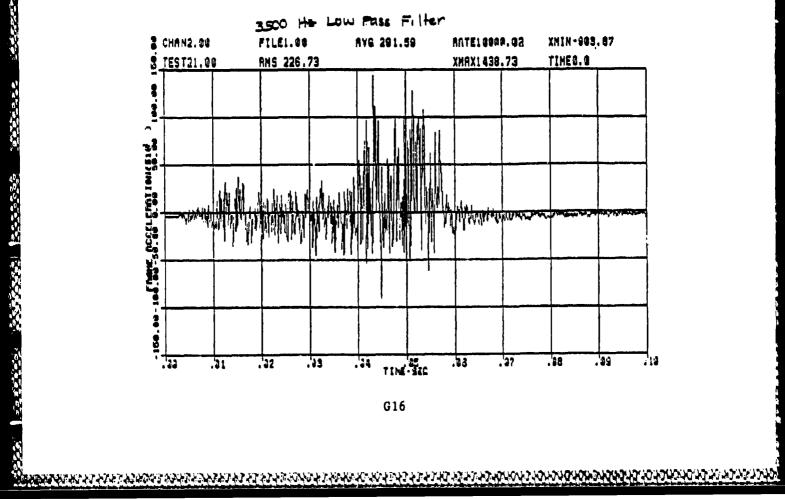


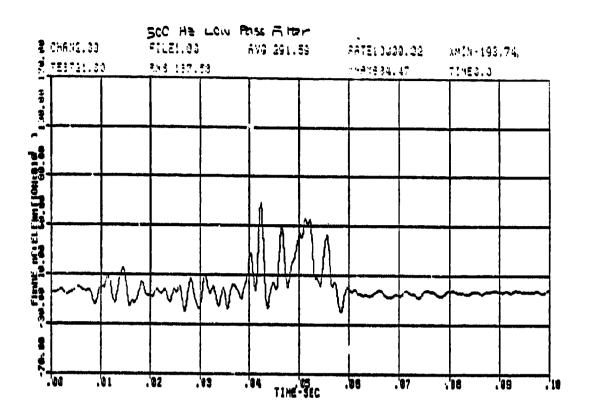




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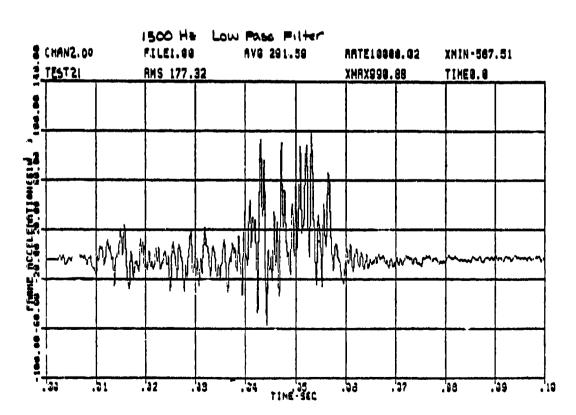


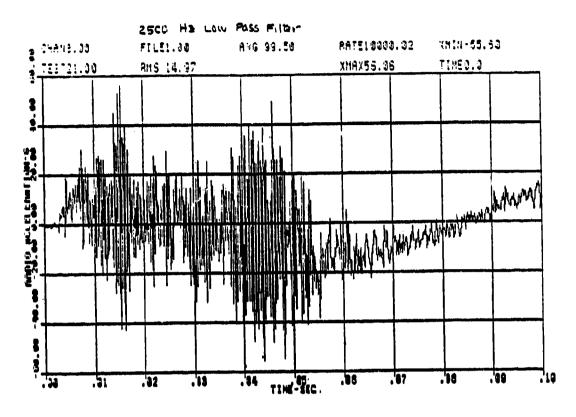


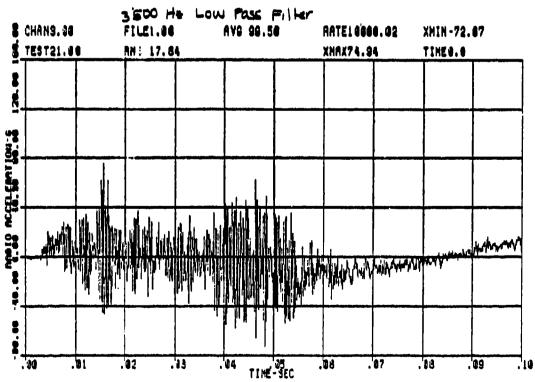
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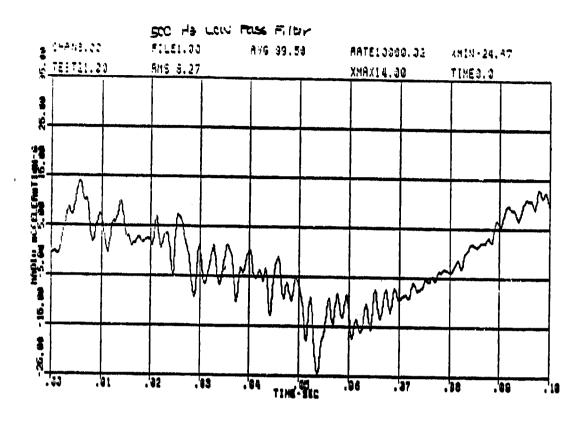
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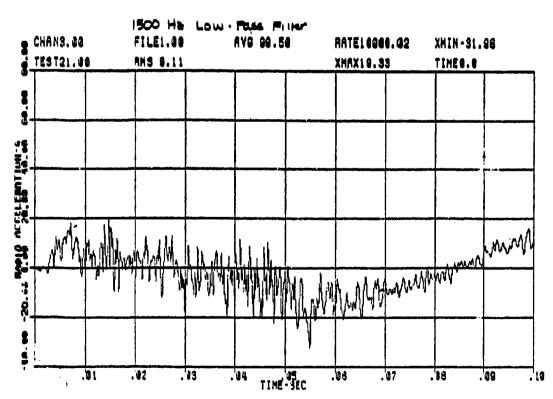




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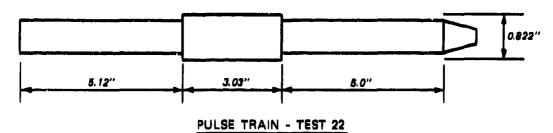
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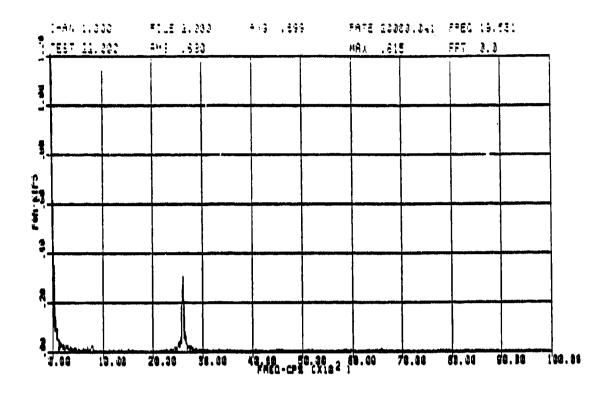


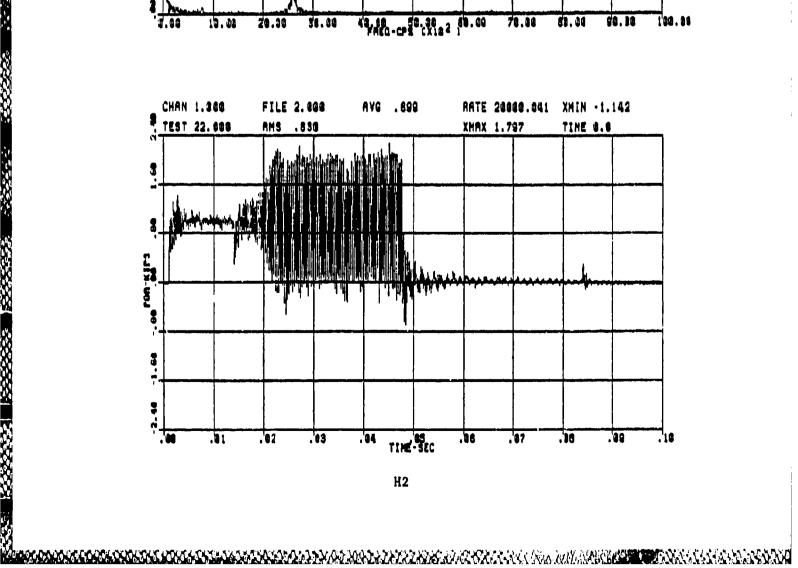
## APPENDIX H: TEST 22 RESULTS

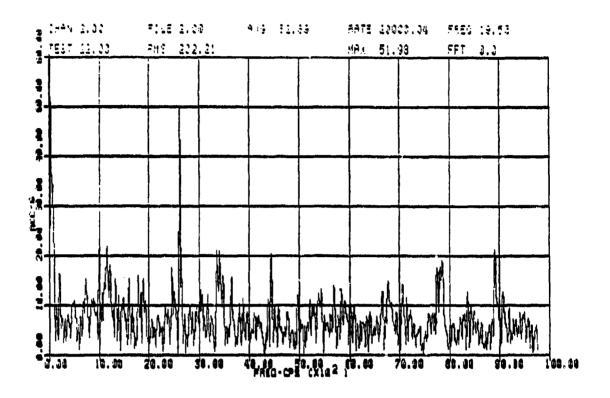
Test 22 Equipment Rack Soft-Mounted AN/GRC-103 in Rack, Off-Line

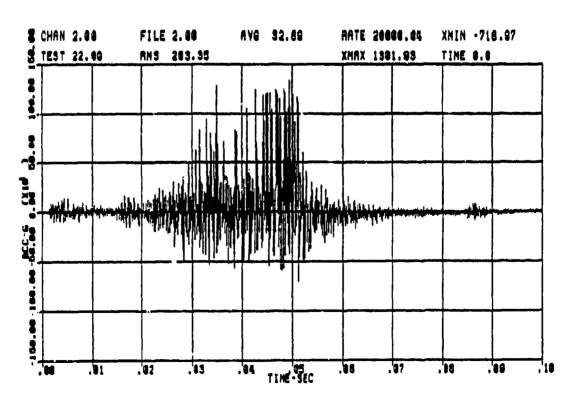


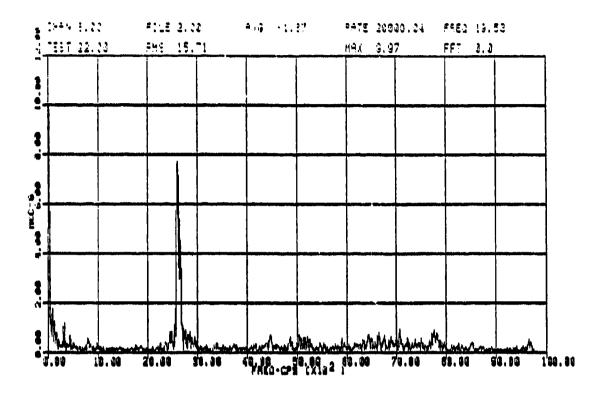
PRECHARGE PRESSURE = 1250 PSI SYSTEM PRESSURE = 2000 PSI FLOW CONTROL #61/2 AIR BAG PRESSURE = 30 PSI

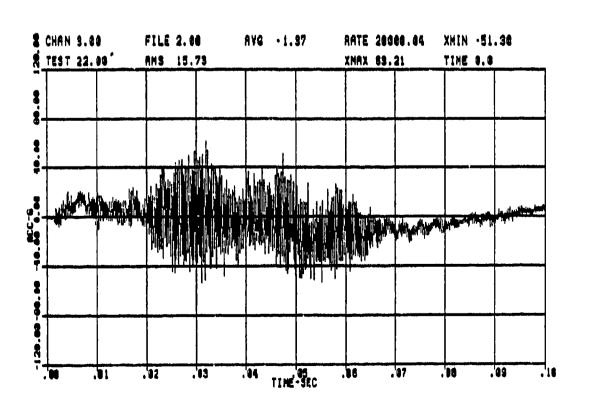


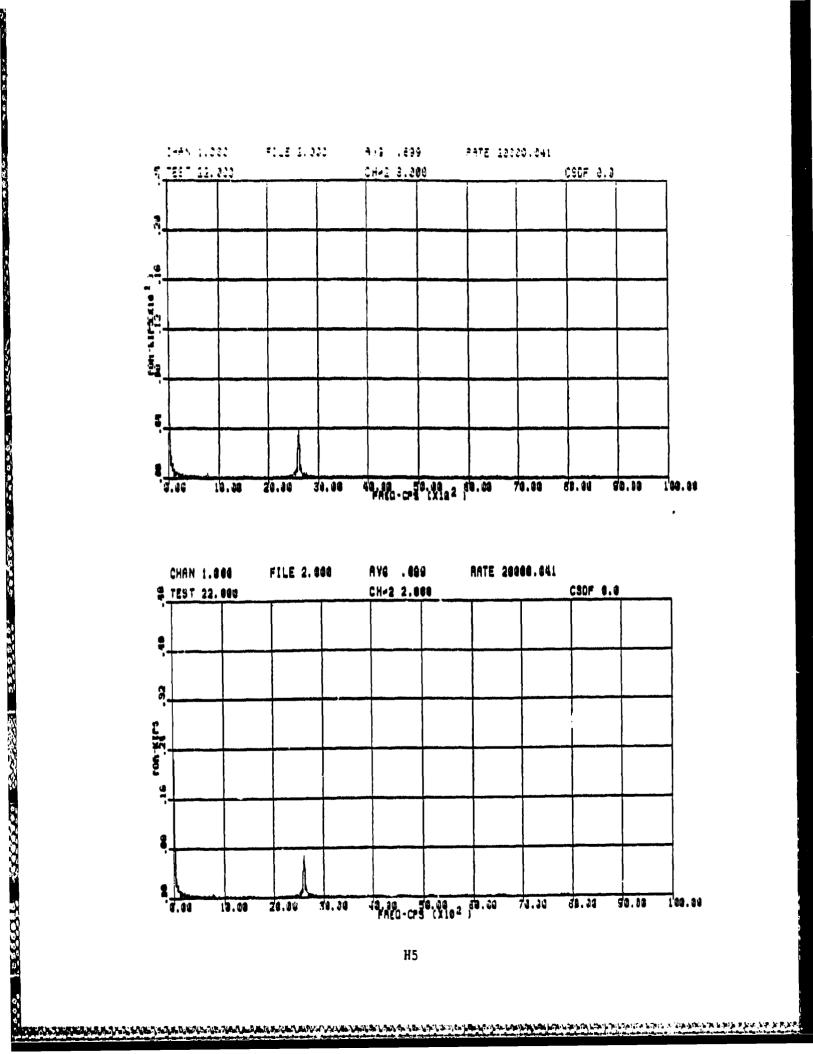


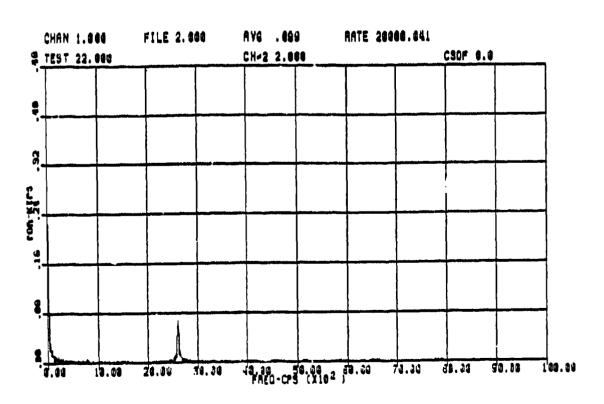


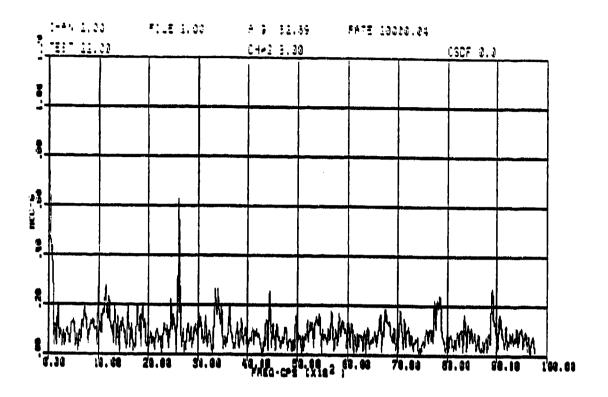








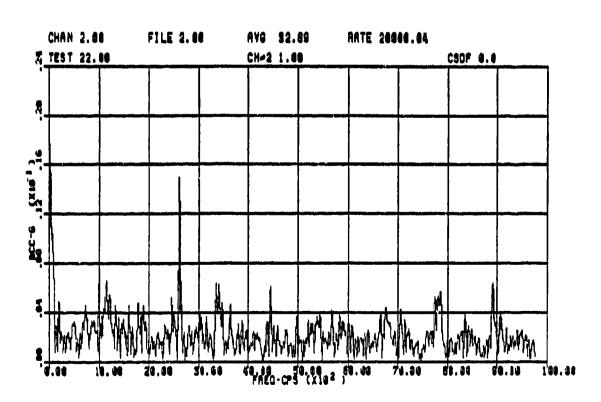


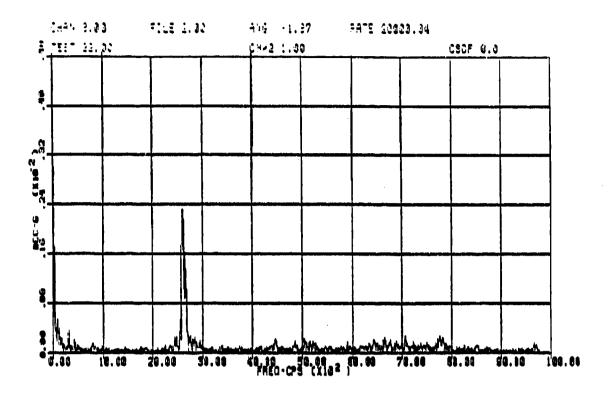


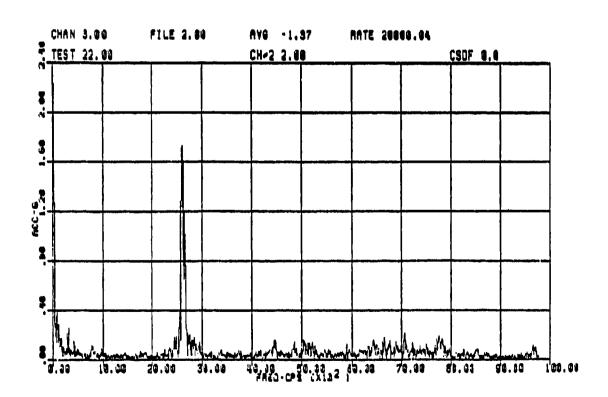
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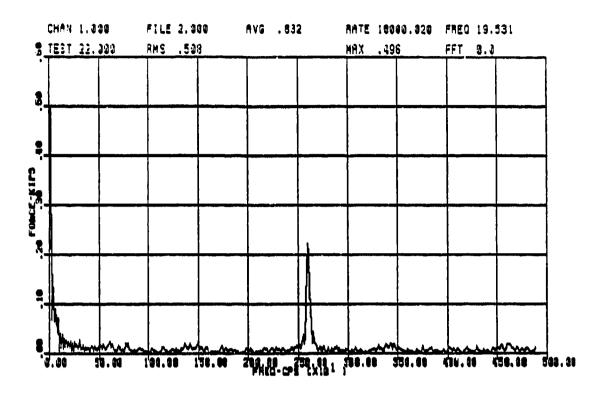


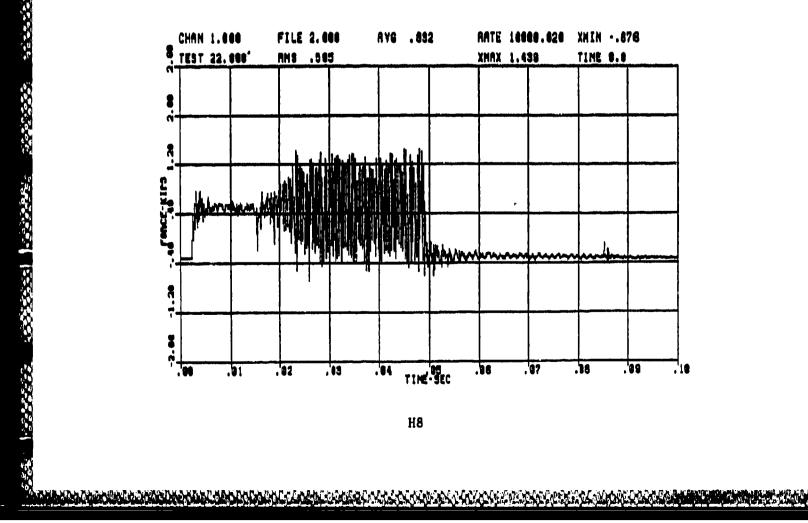


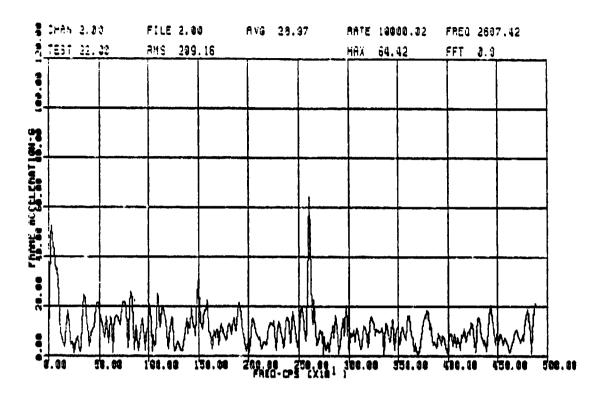


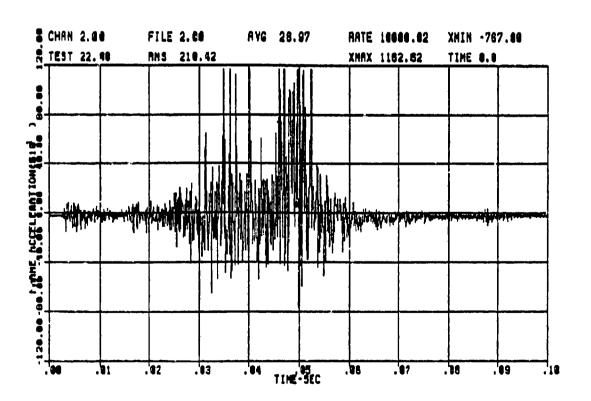
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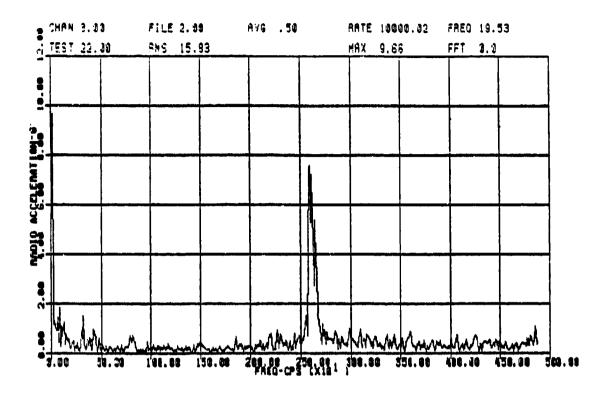
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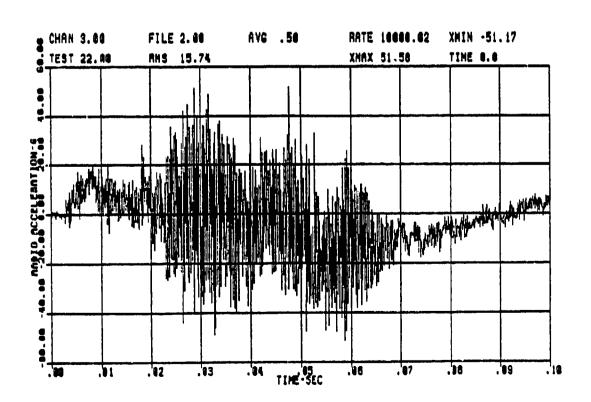


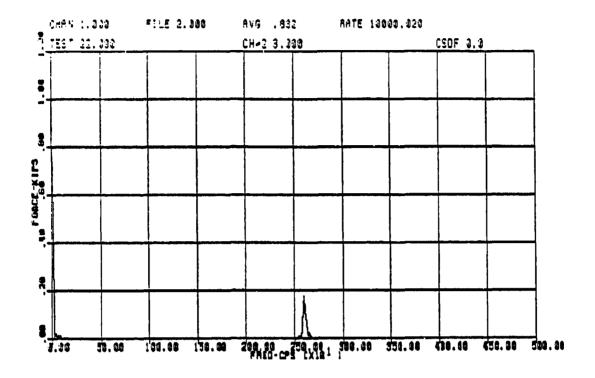


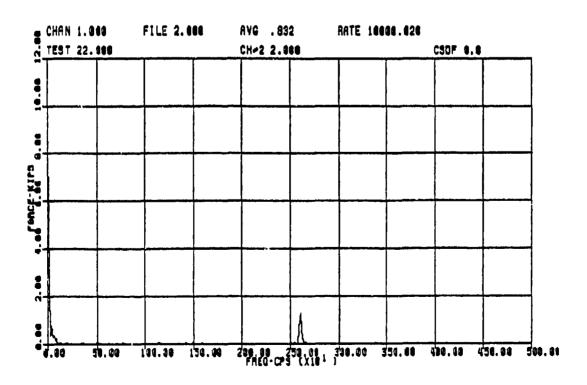






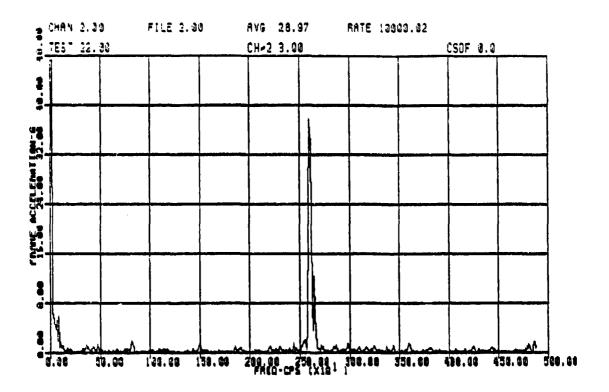






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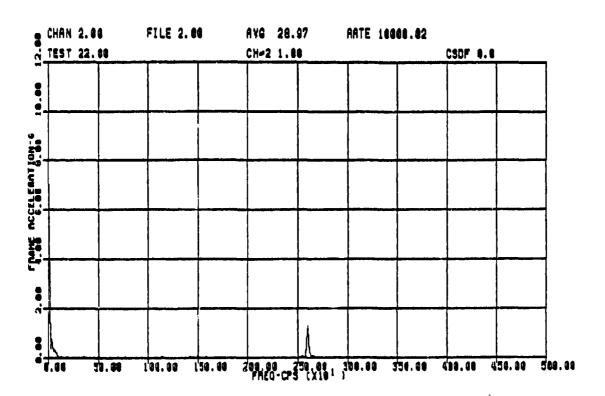
Proceeding

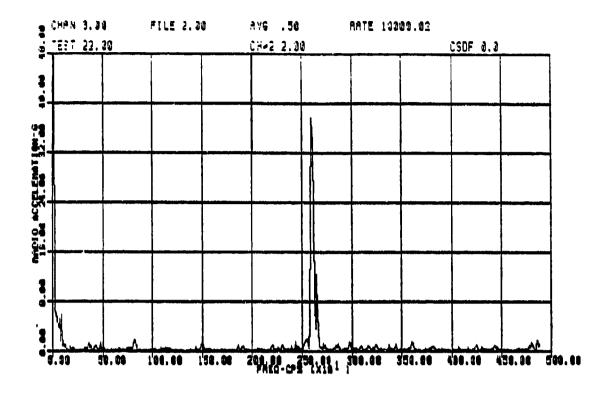


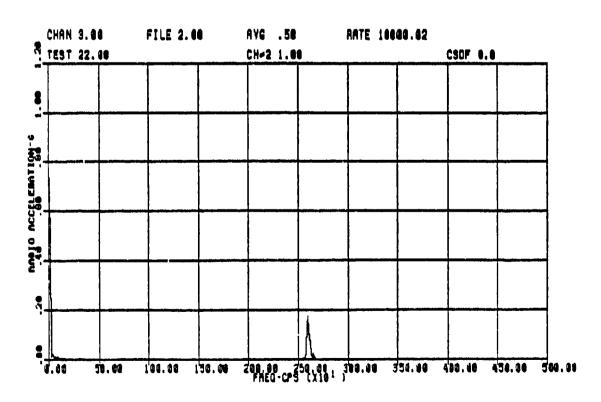
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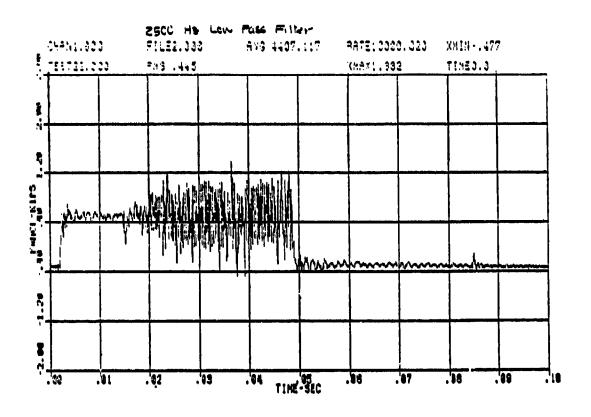
Secretary Secretary

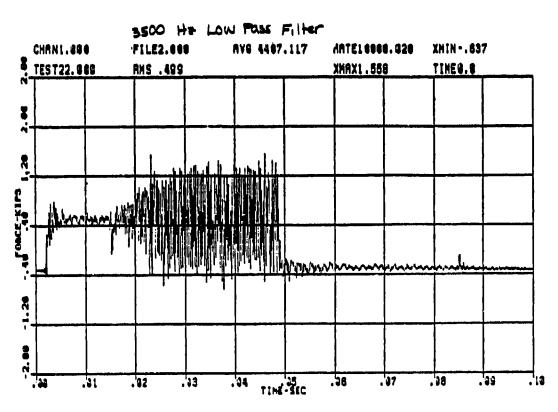
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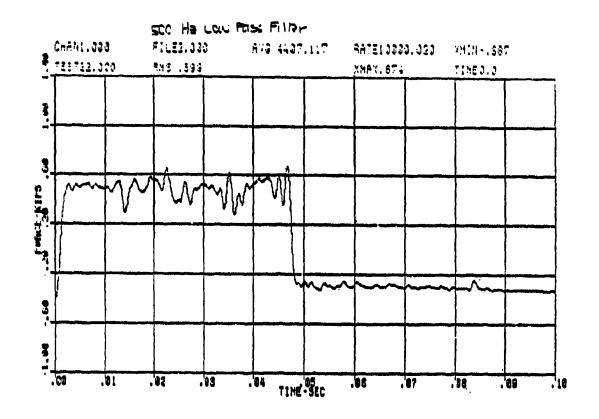


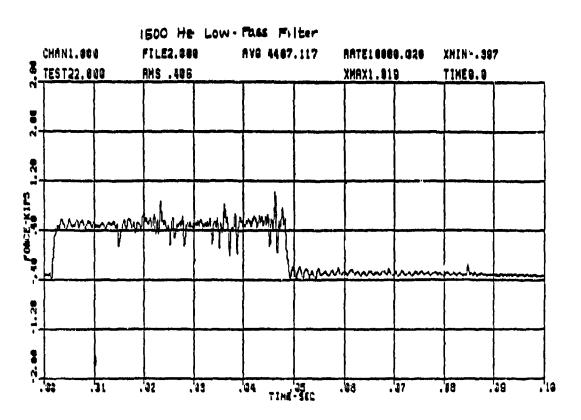




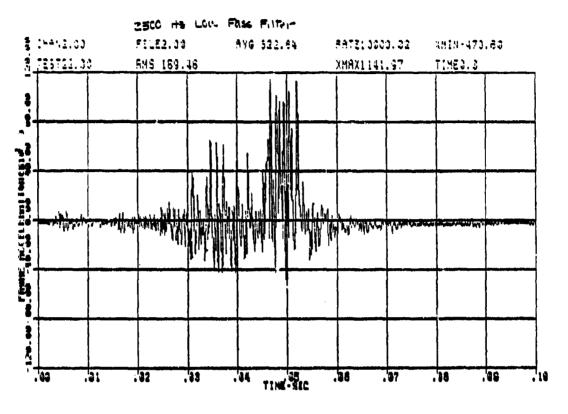




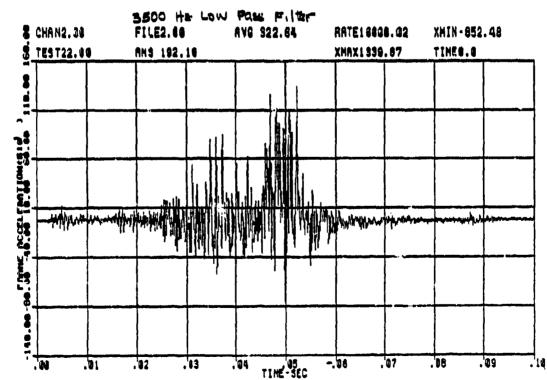


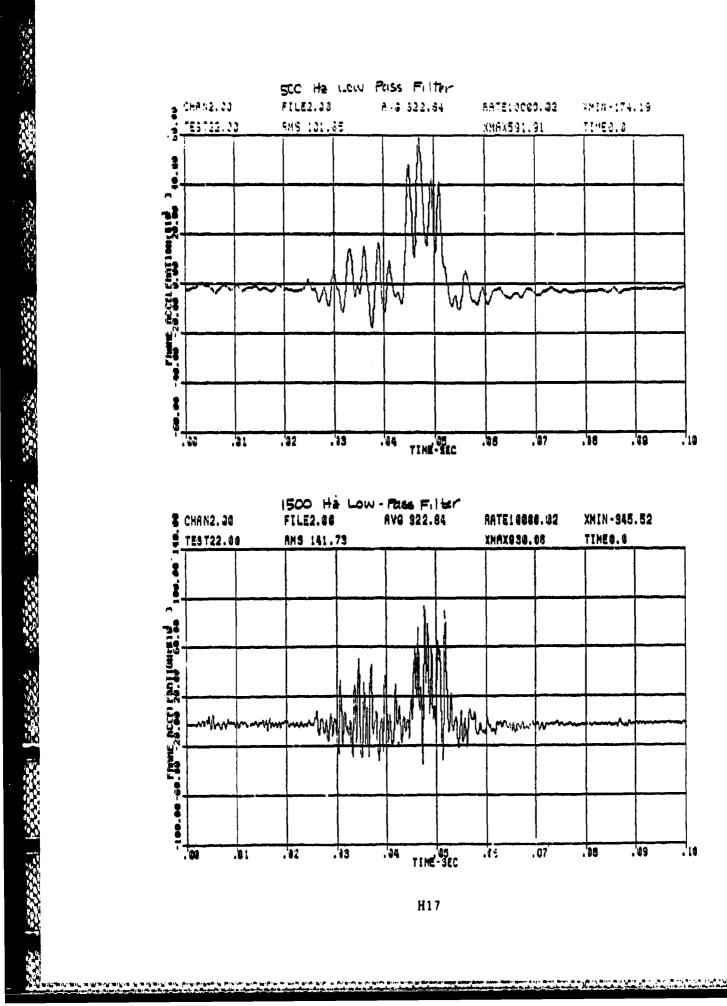


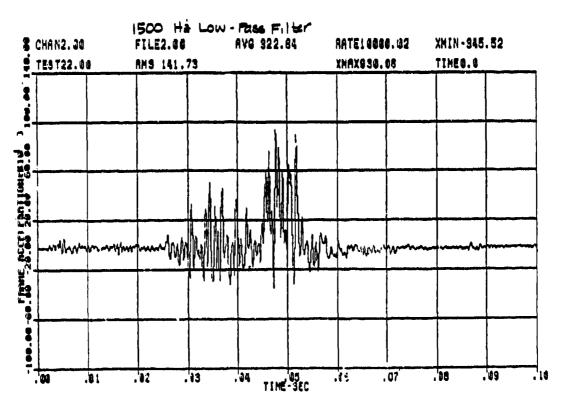
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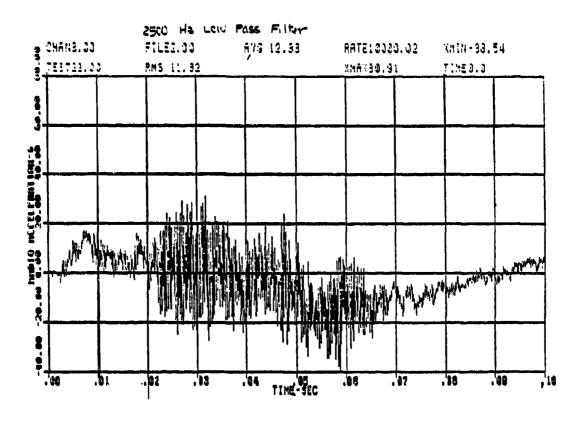
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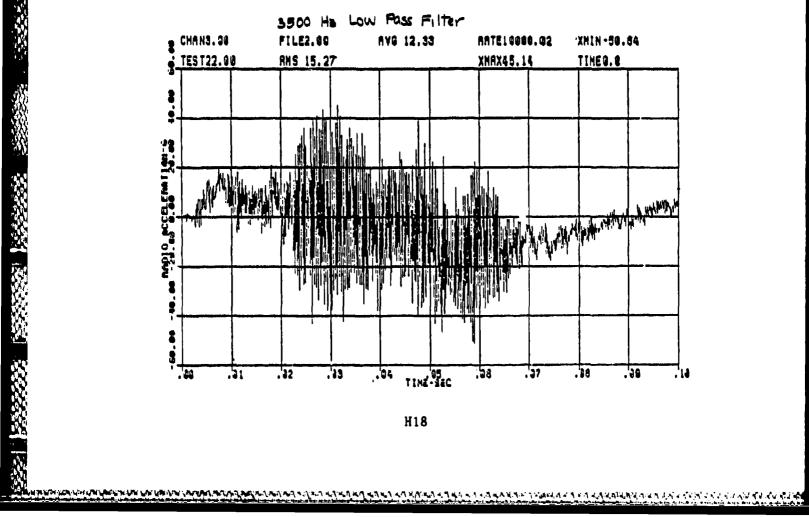


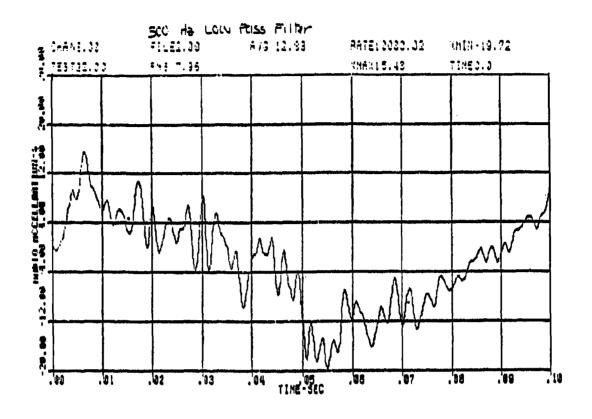


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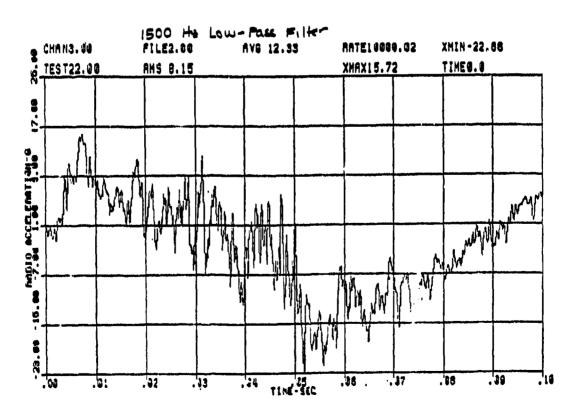


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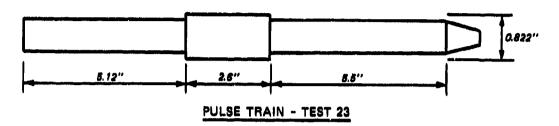
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## APPENDIX 1: TEST 23 RESULTS

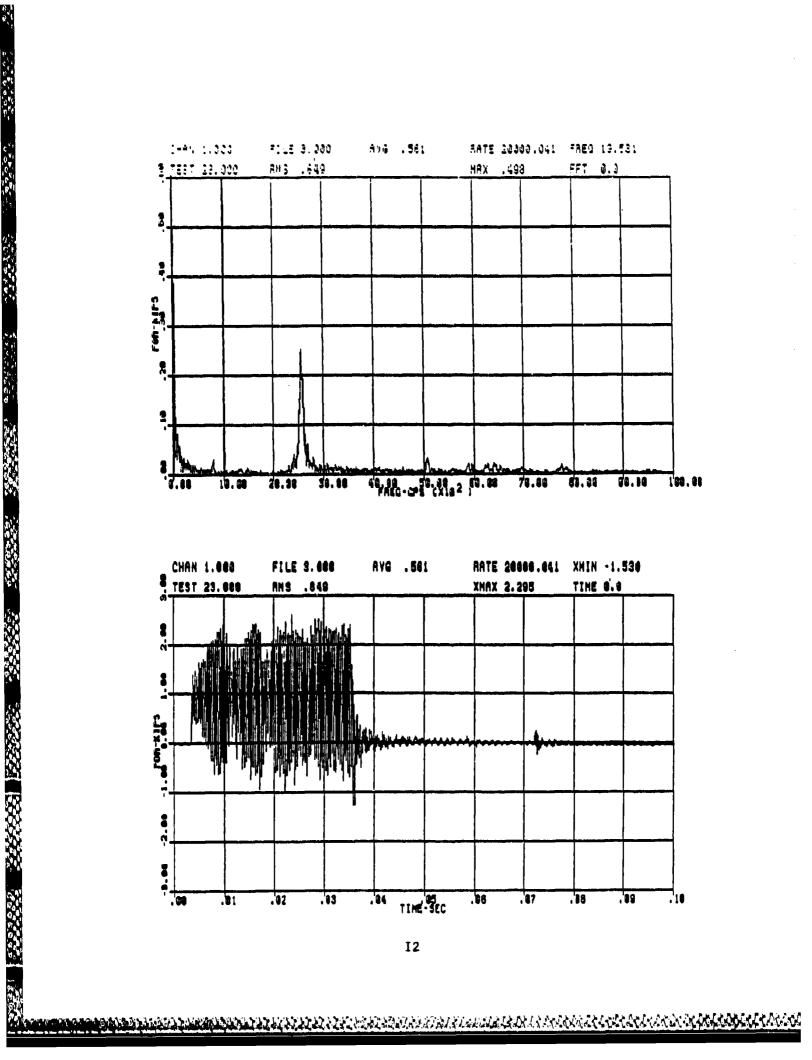
Test 23 Equipment Rack Soft-Mounted AN/GRC-103 in Rack, Off-Line

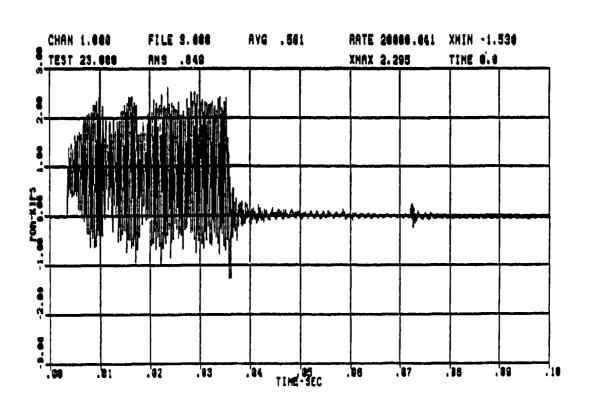


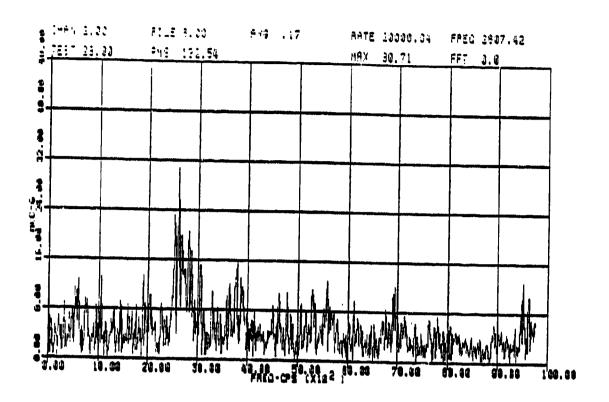
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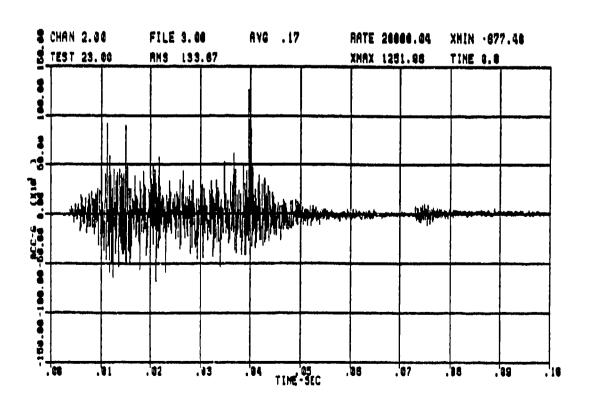
Sciences Carpeter.

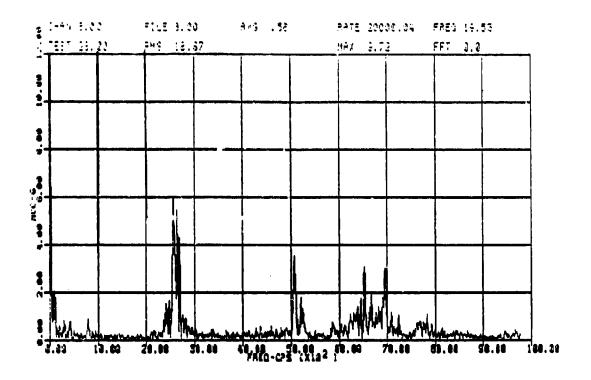
- 70 PSI

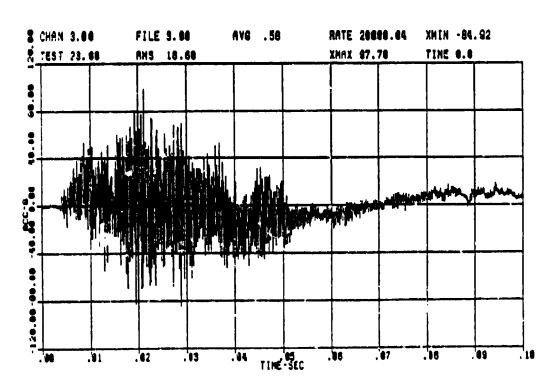


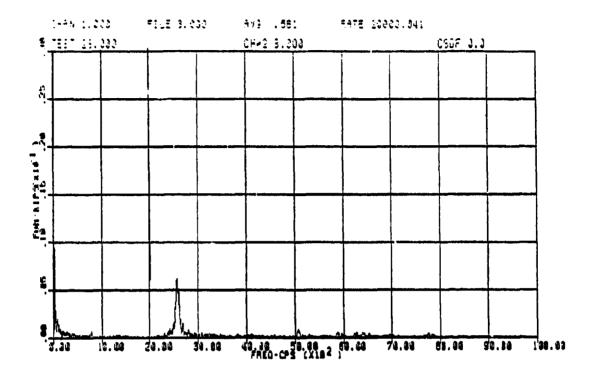


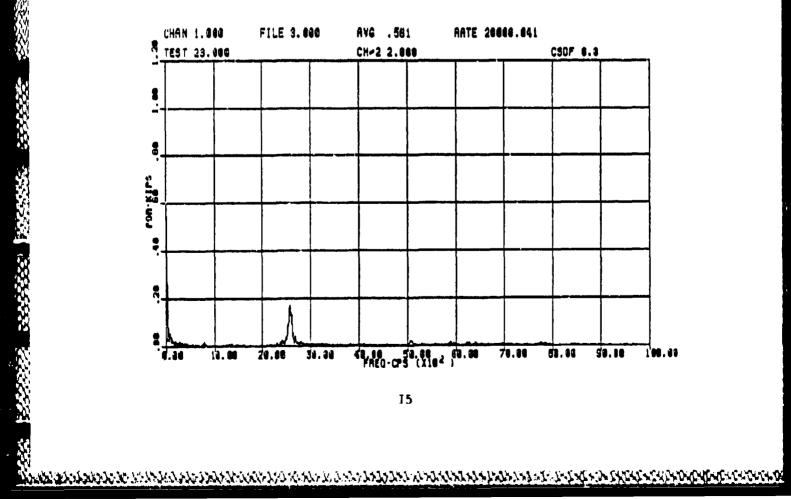


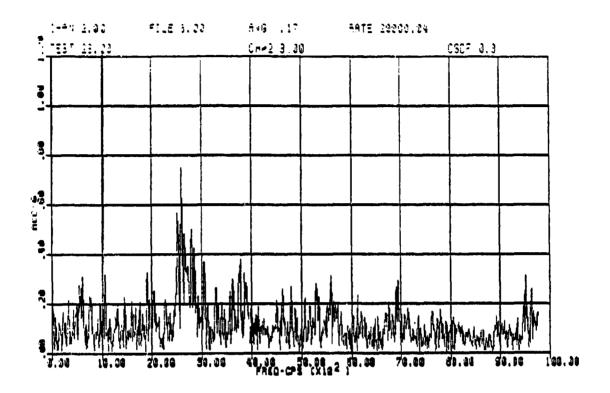


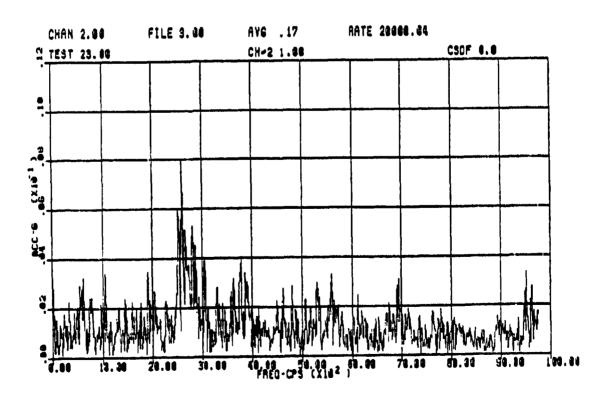


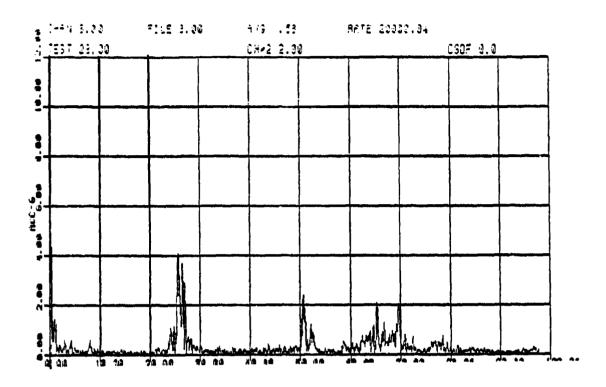


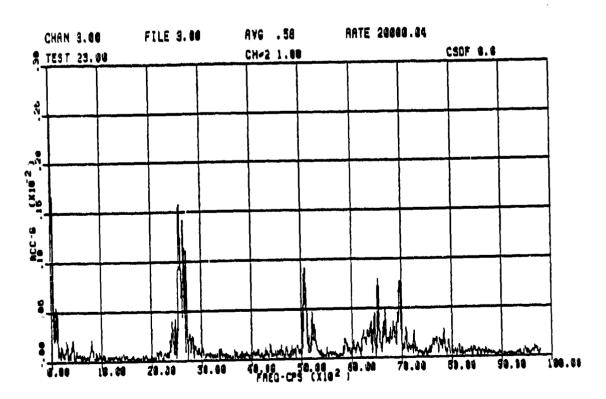


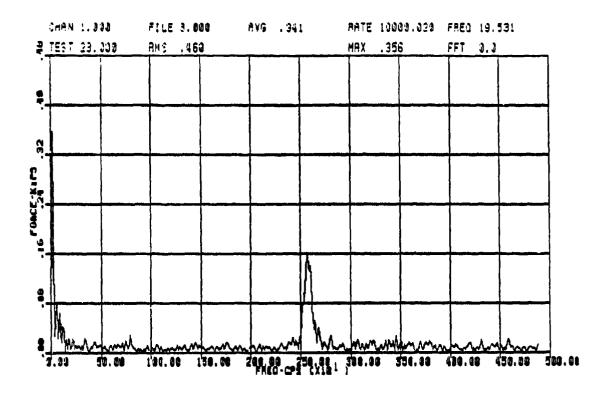




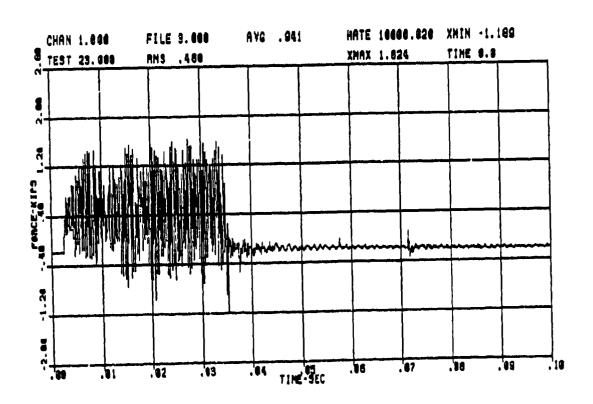


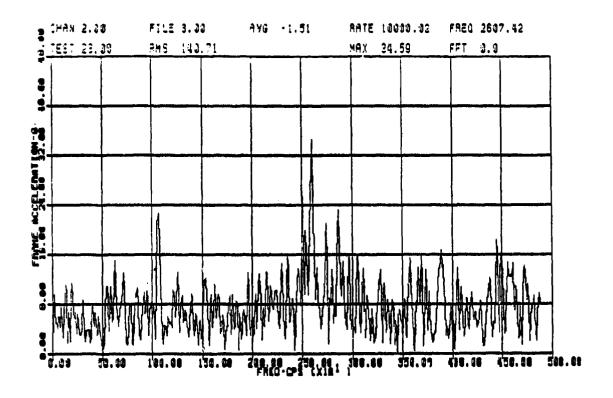




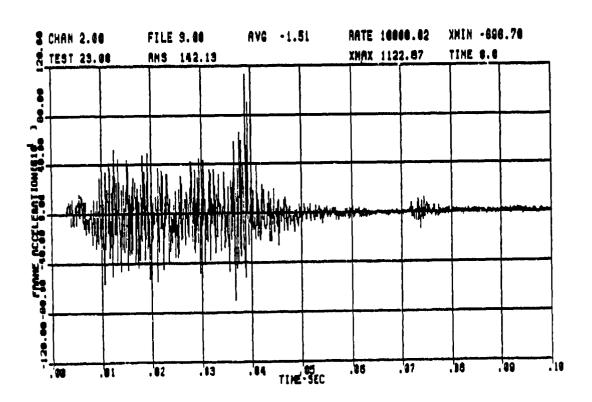


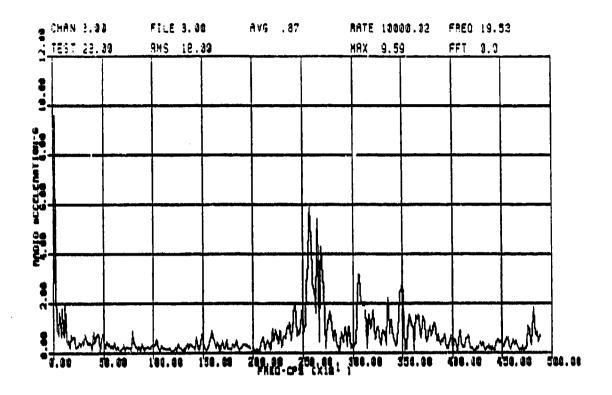
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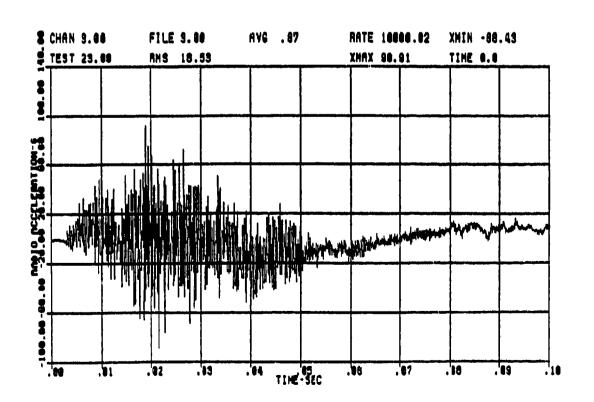




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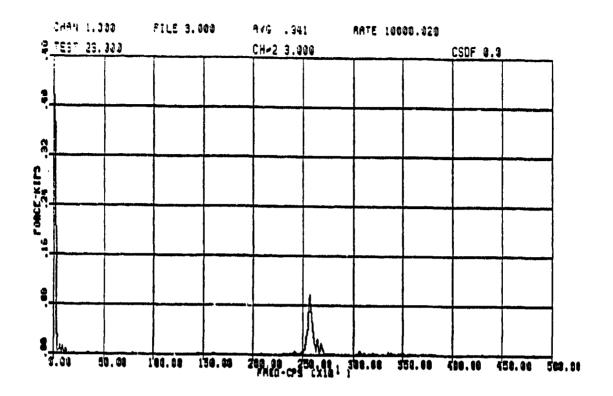


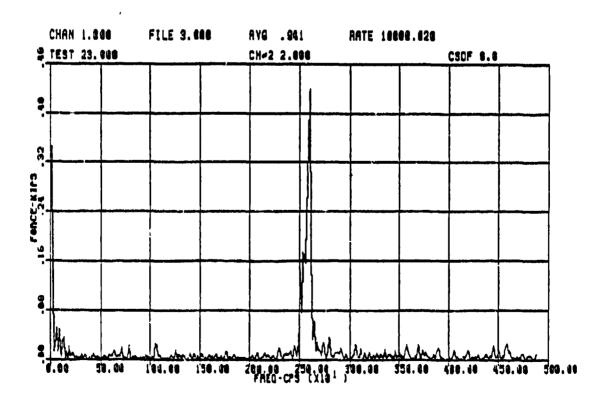


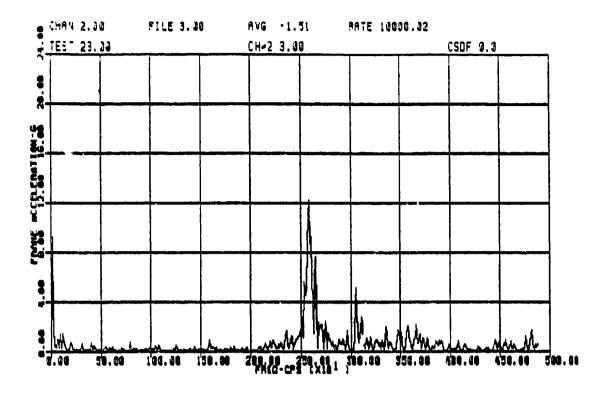


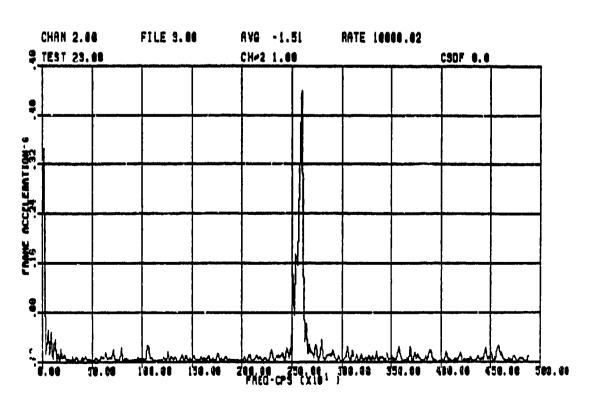
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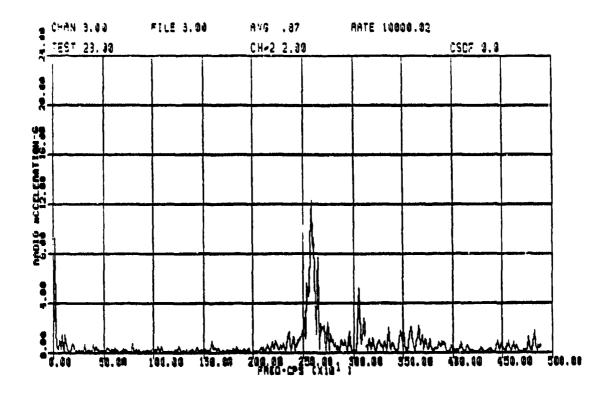
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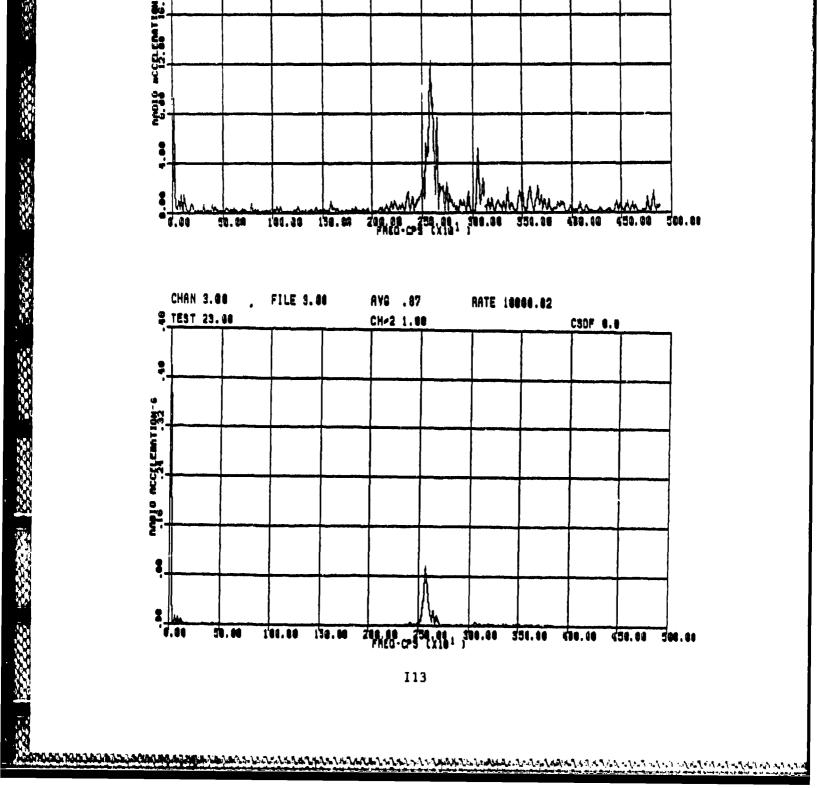


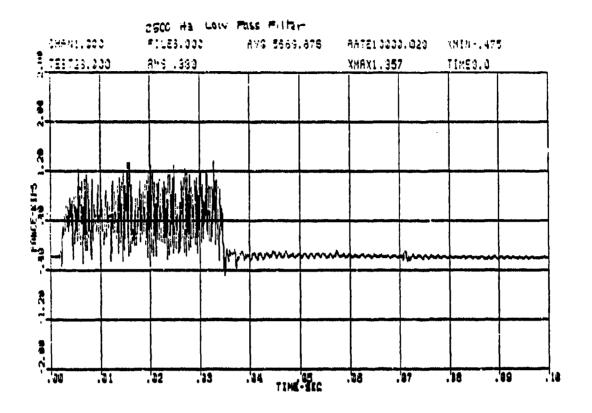




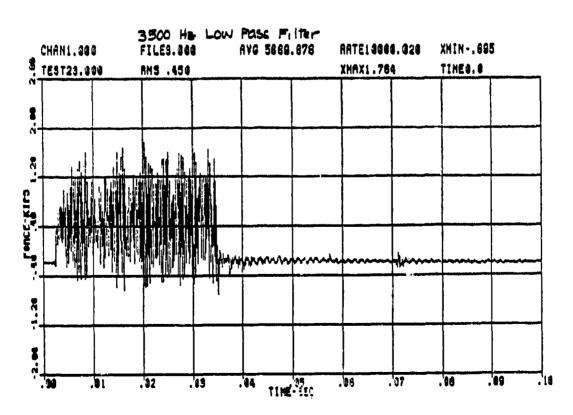




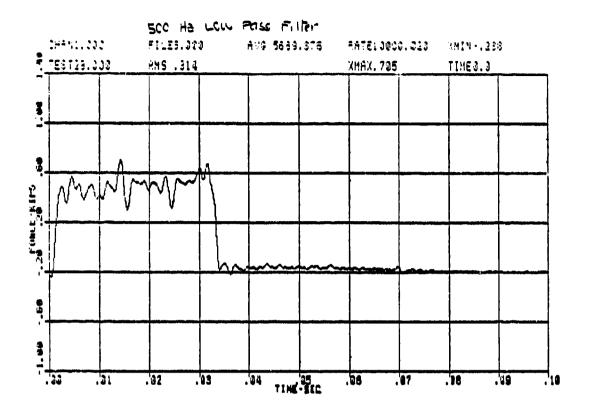


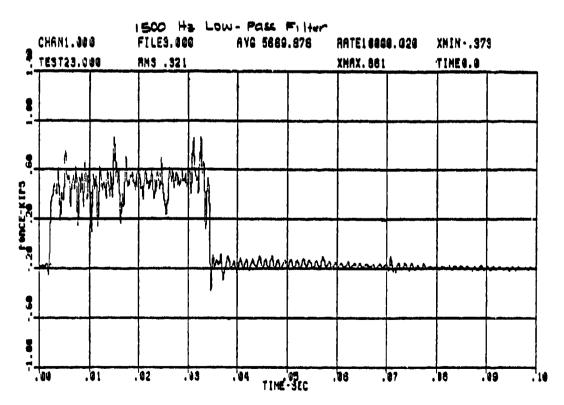


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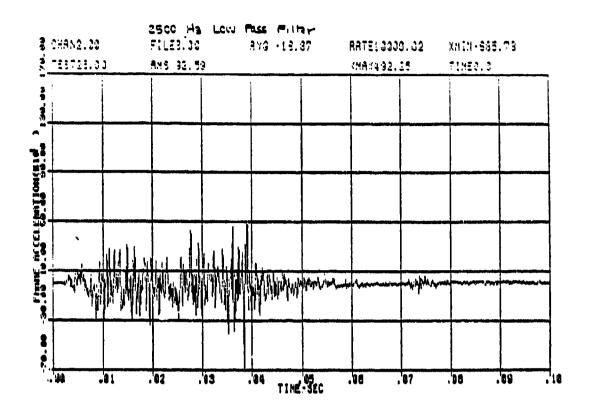


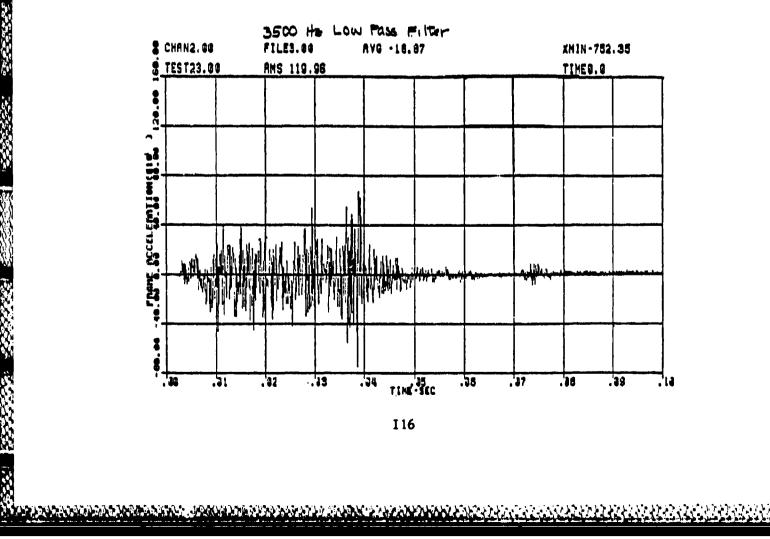
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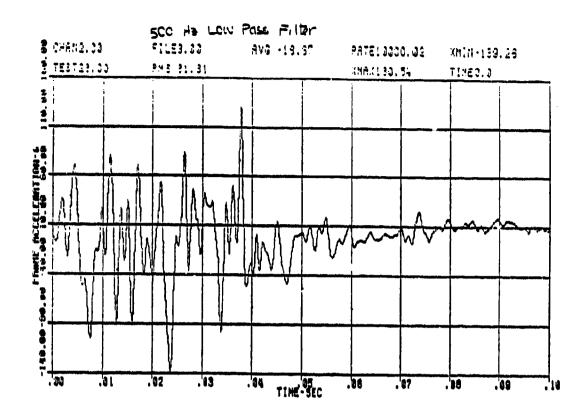




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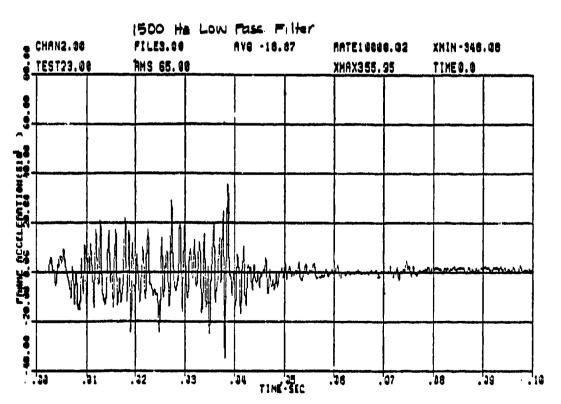


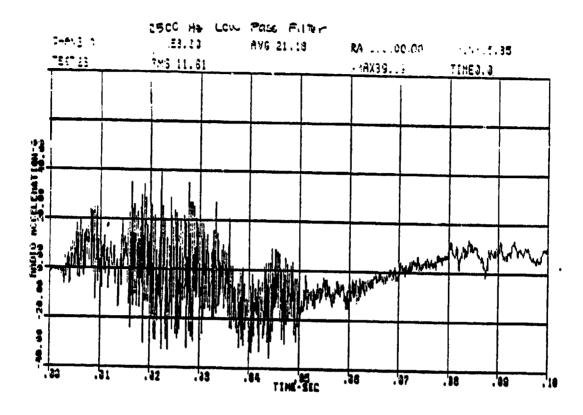


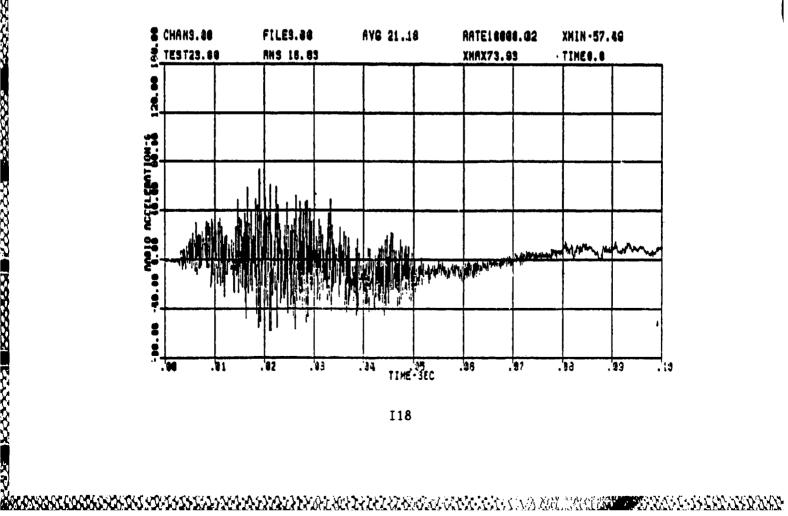
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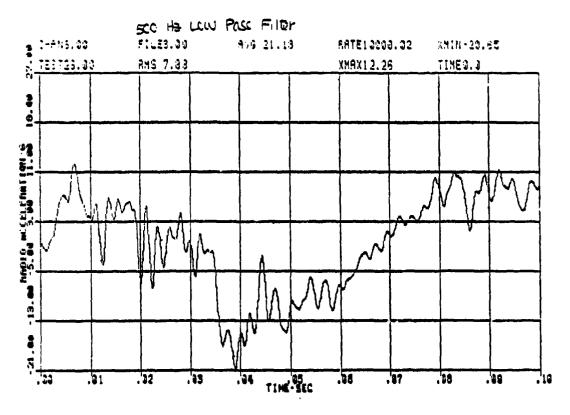
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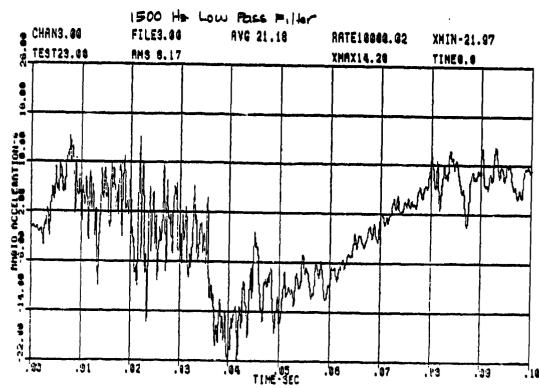




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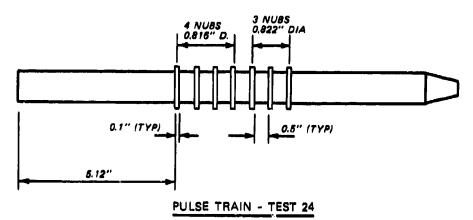
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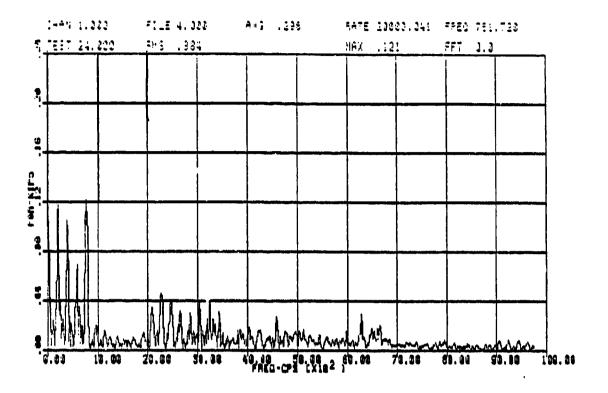


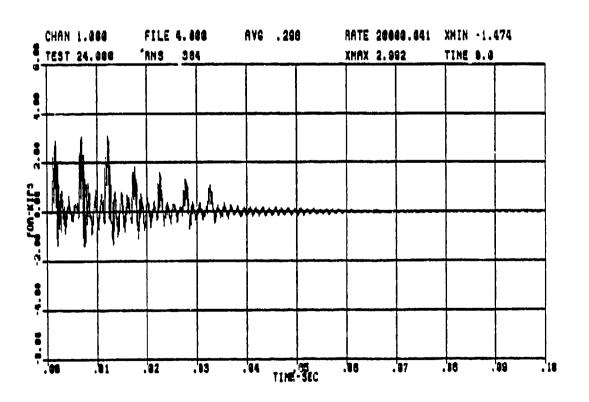
## APPENDIX J: TEST 24 RESULTS

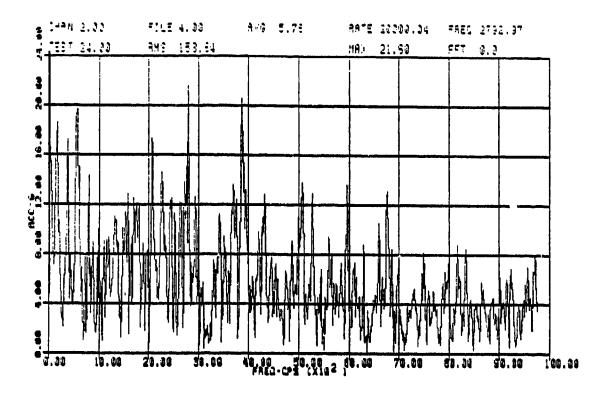
Test 24 Equipment Rack Soft-Mounted AN/GRC-103 in Rack, Off-Line

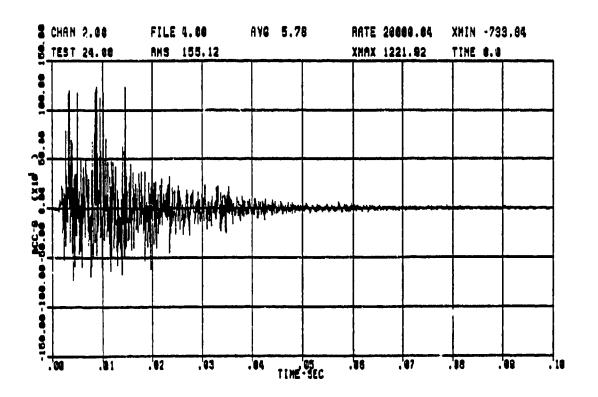


PRECHARGE PRESSURE = 1250 PSI SYSTEM PRESSURE = 2000 PSI FLOW CONTROL #6½ AIR BAG PRESSURE = 70 PSI

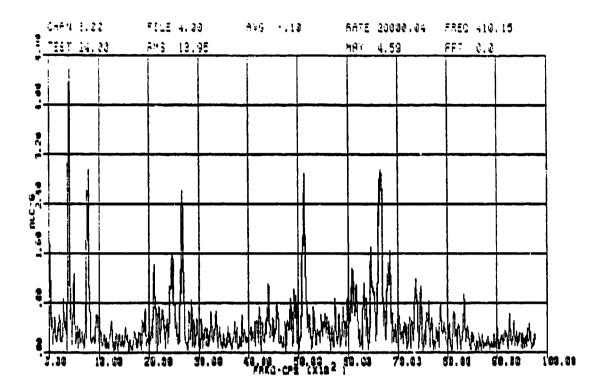


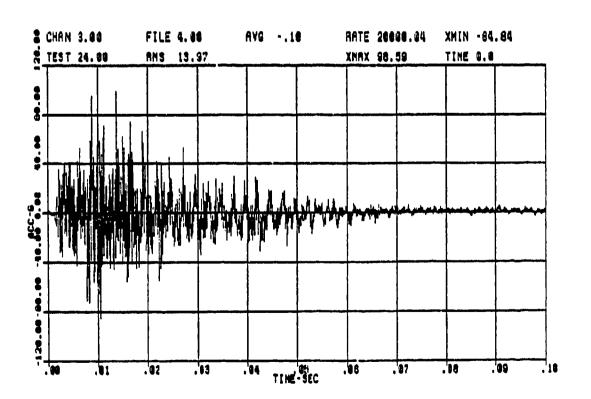


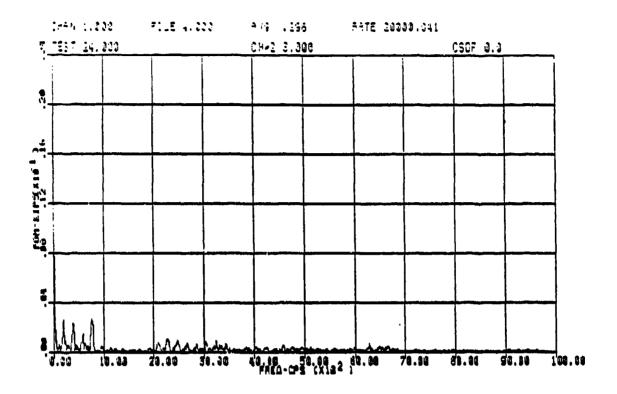


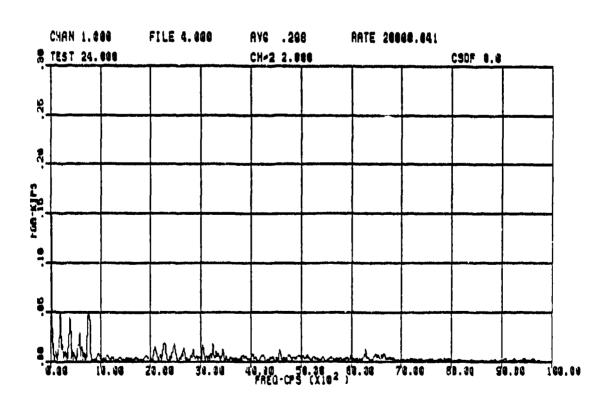


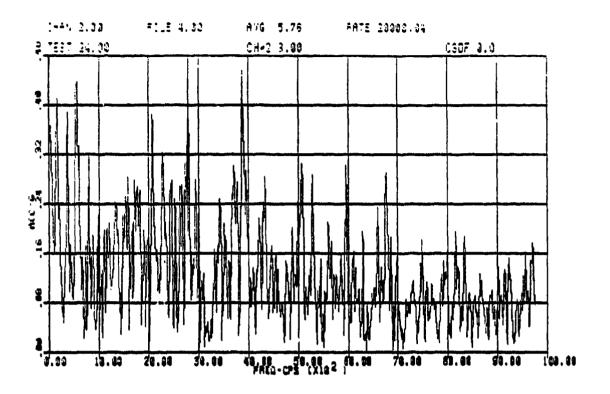
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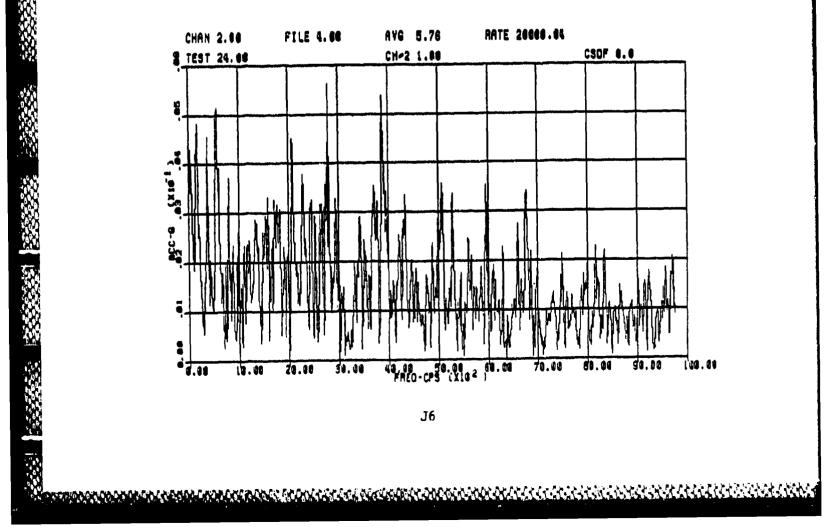


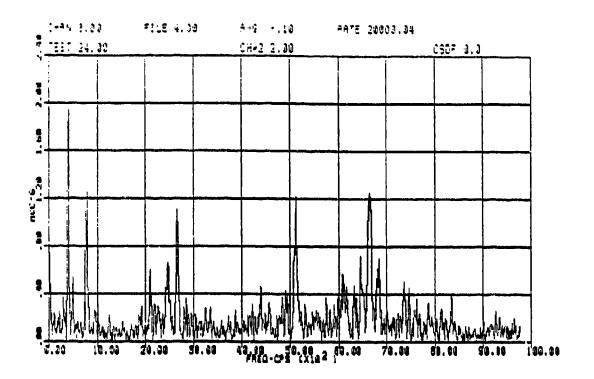






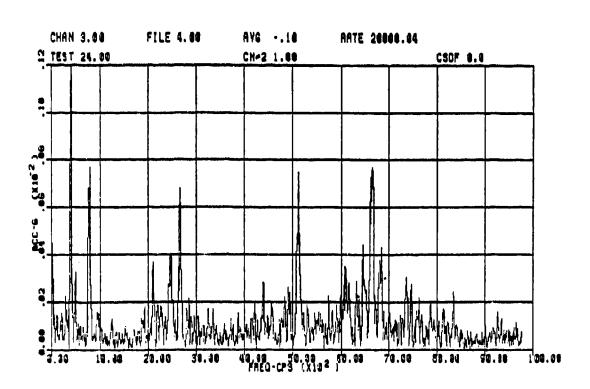


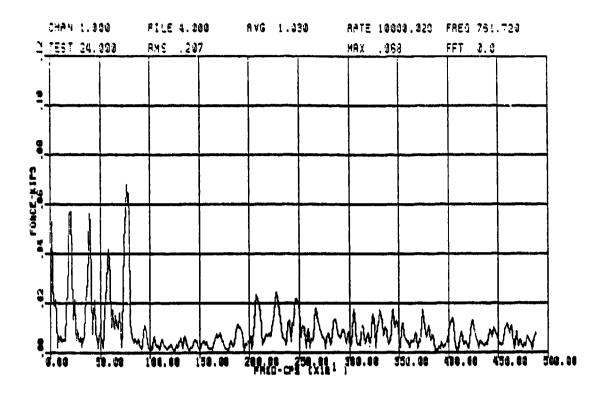


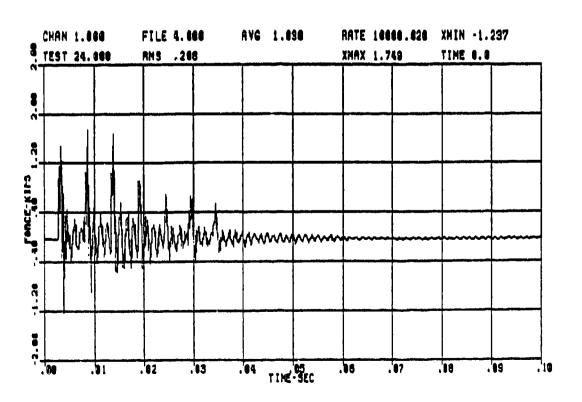


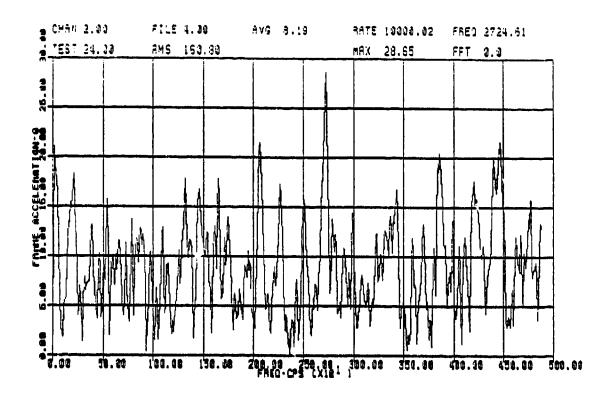
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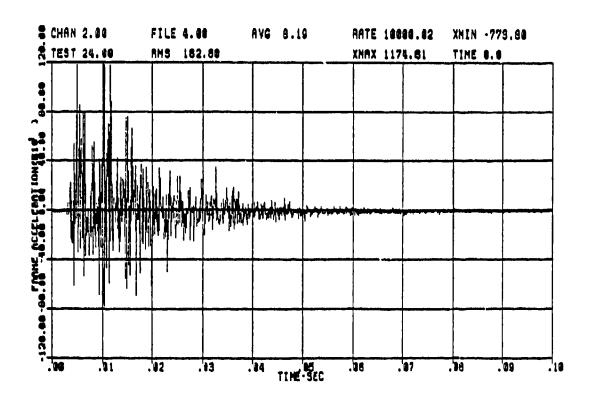
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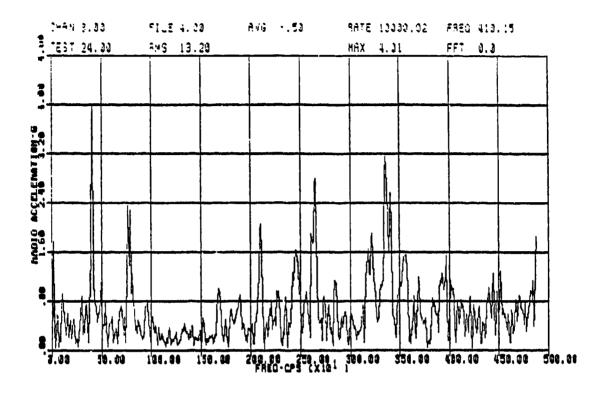


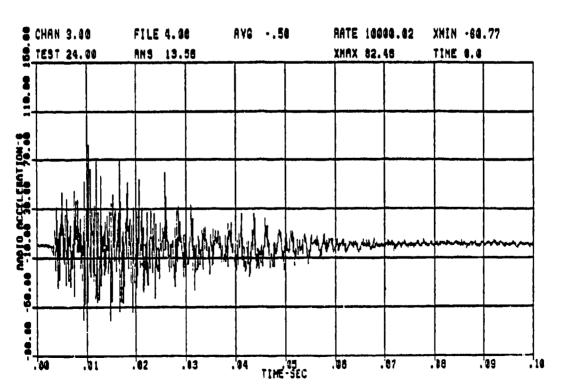


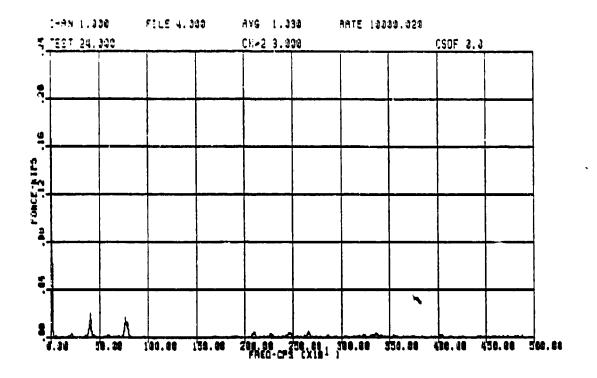


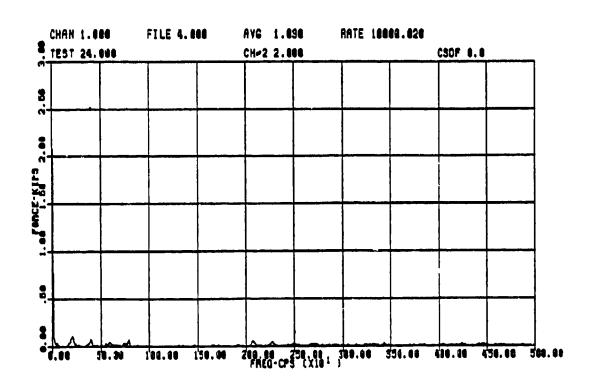
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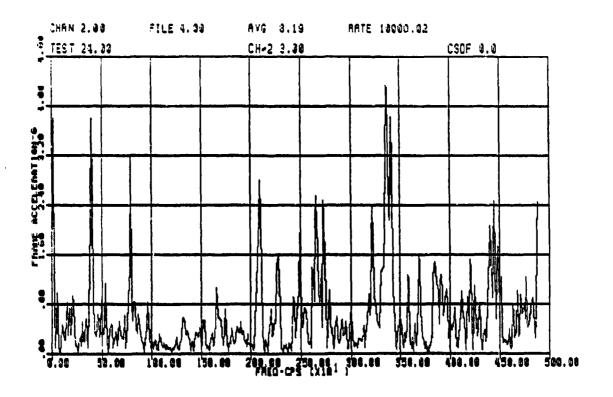
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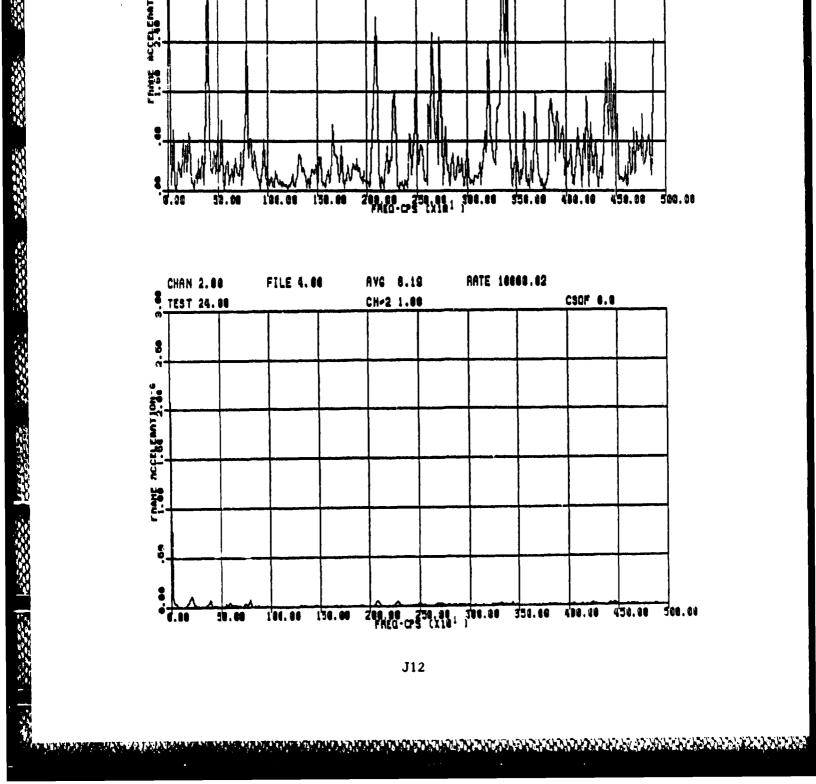


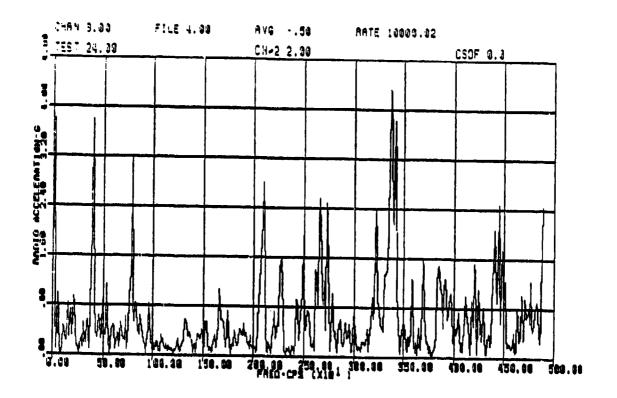




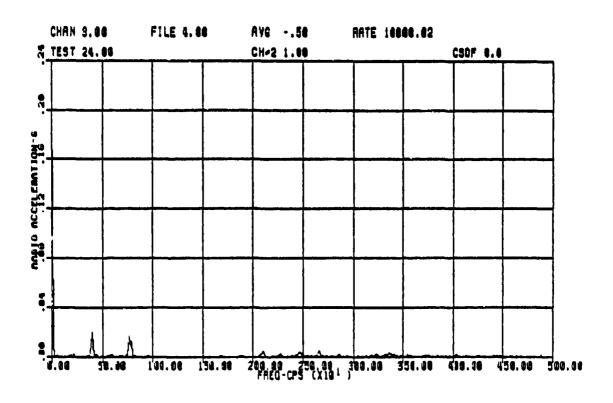


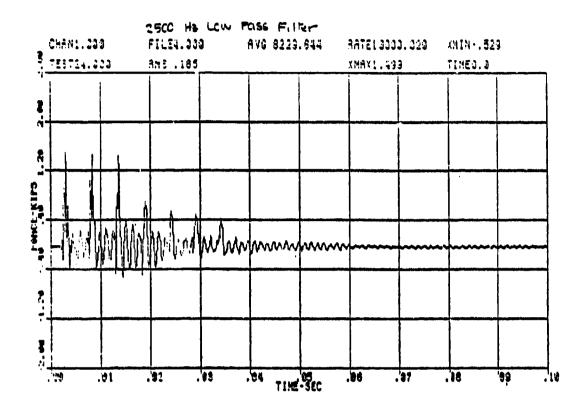


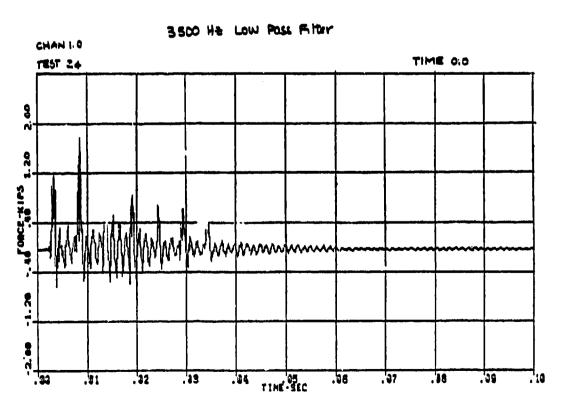




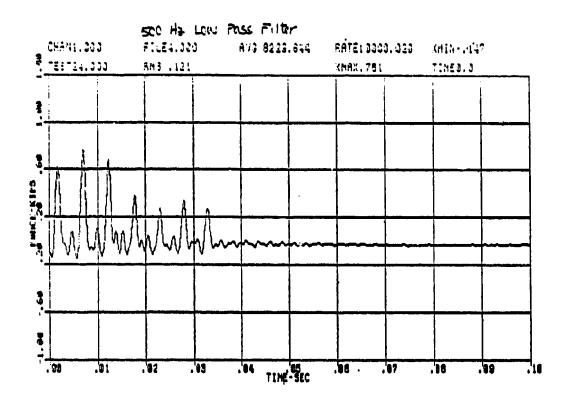
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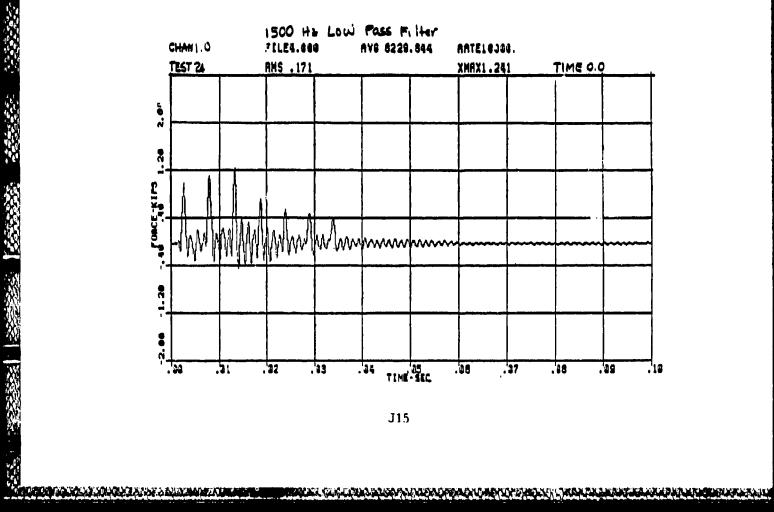


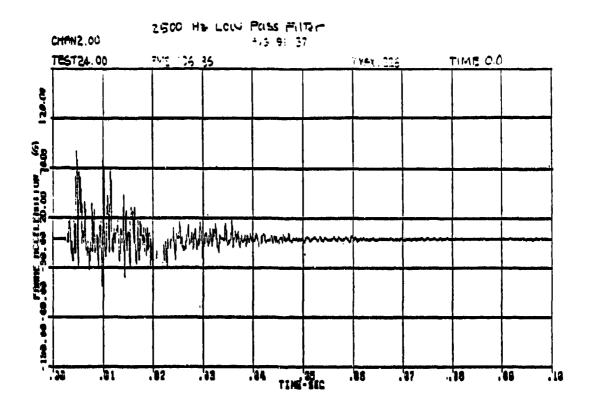


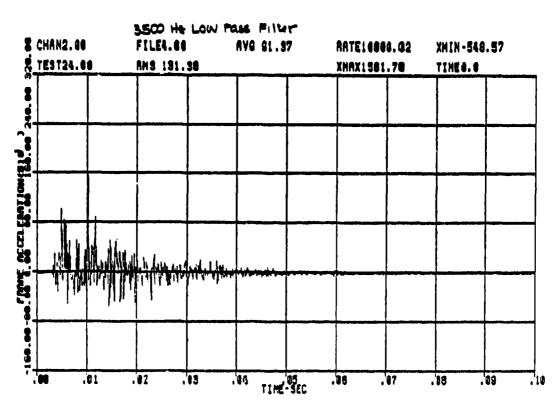


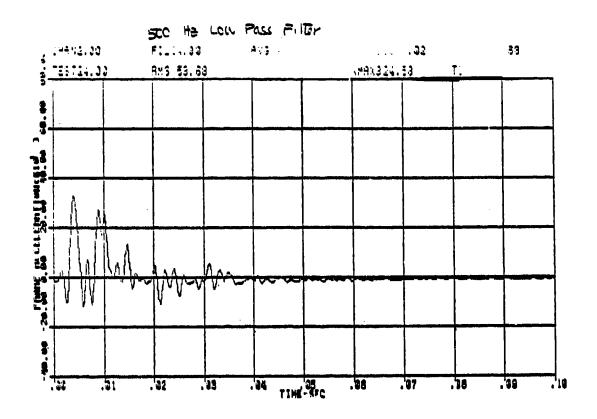
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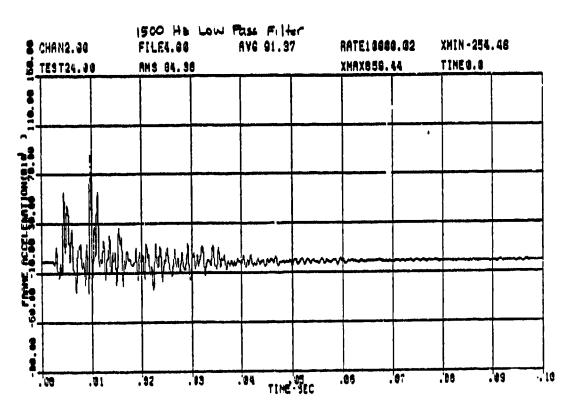




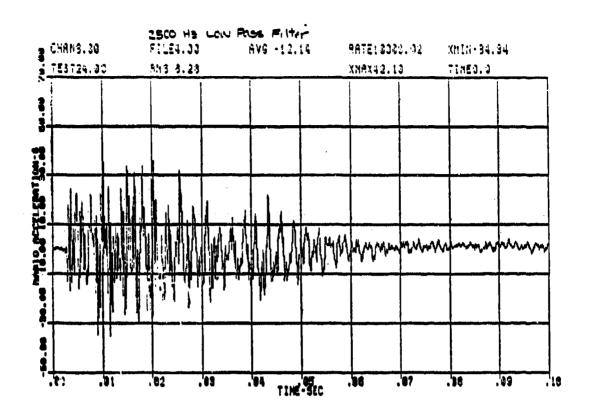


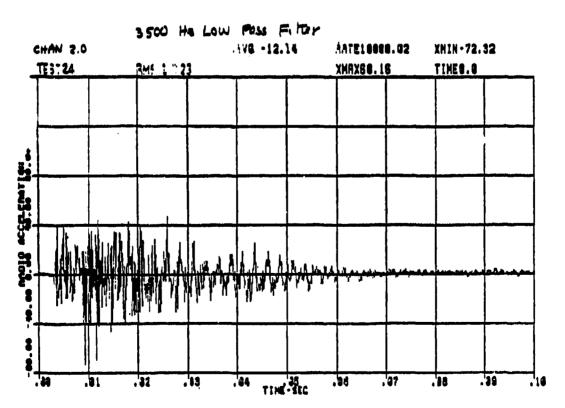


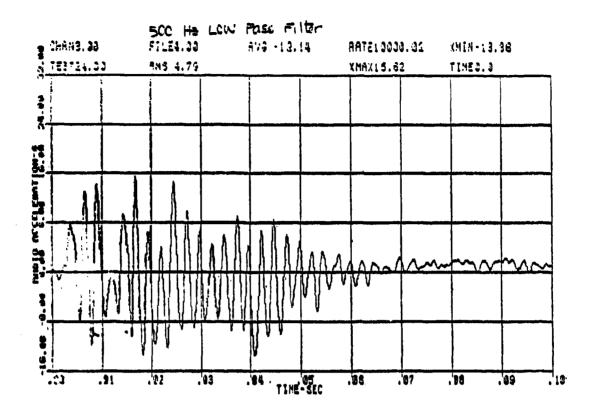




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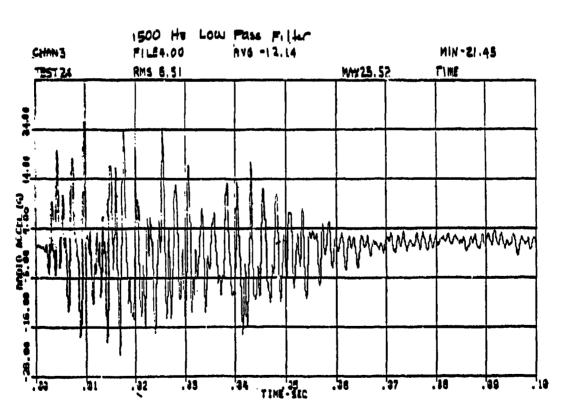




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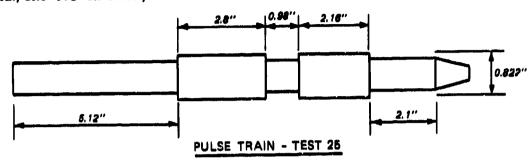
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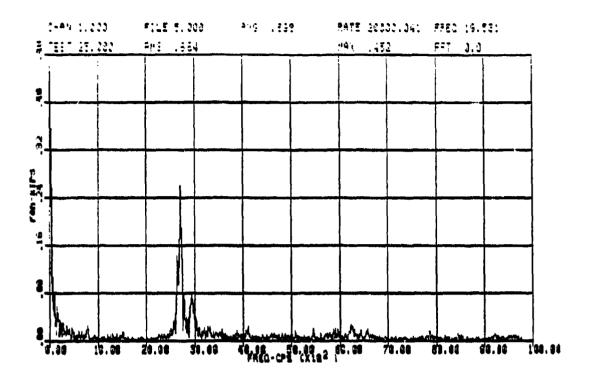
## APPENDIX K: TEST 25 RESULTS

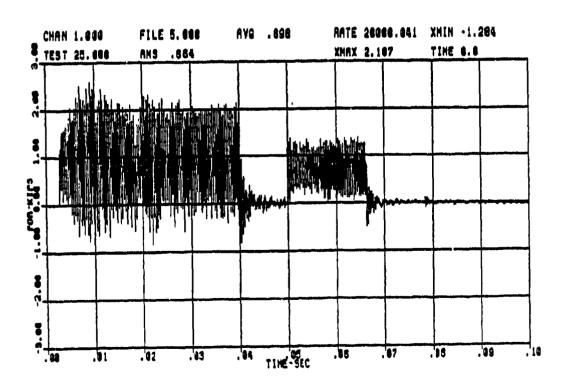
Test 25
Equipment Rack Soft-Mounted
AN/GRC-103 in Rack, Off-Line

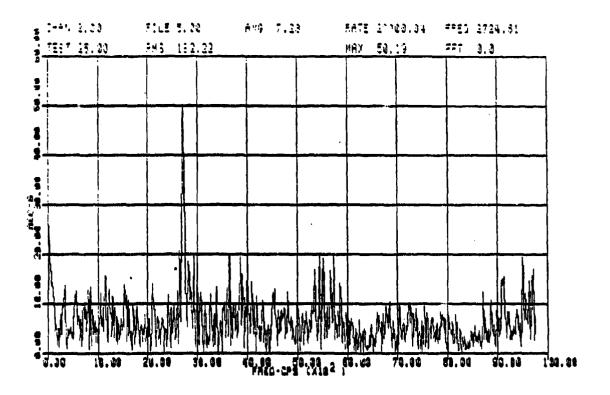


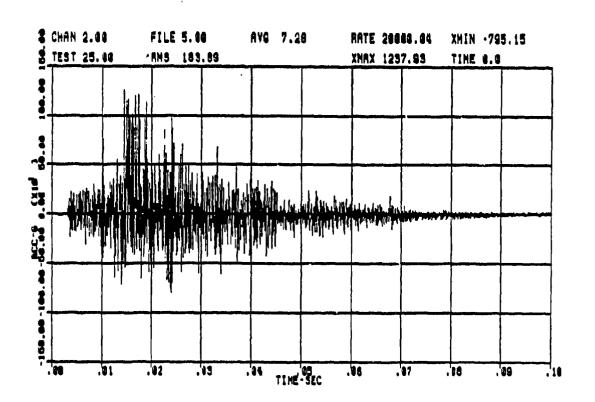
PRECHARGE PRESSURE = 1250 PSI SYSTEM PRESSURE = 2000 PSI FLOW CONTROL #8% AIR BAG PRESSURE = 50 PSI

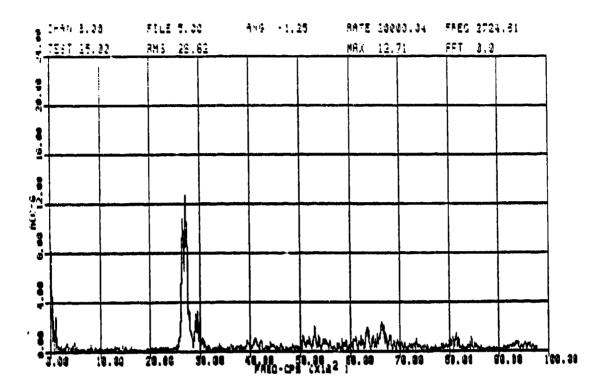
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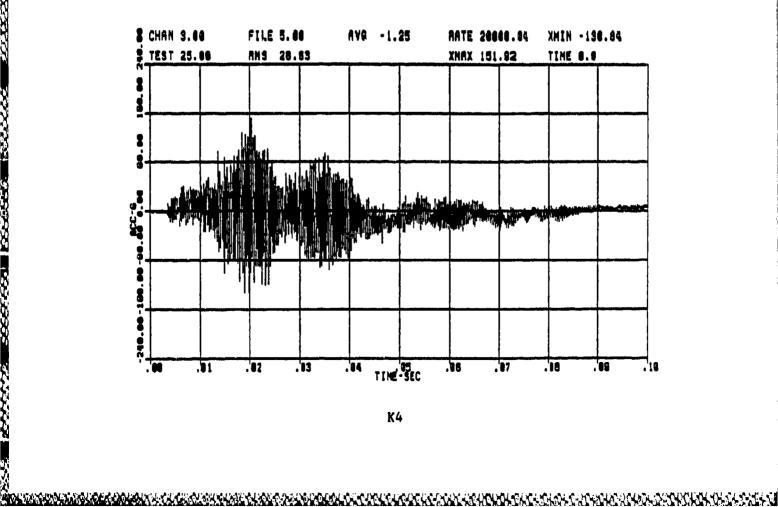


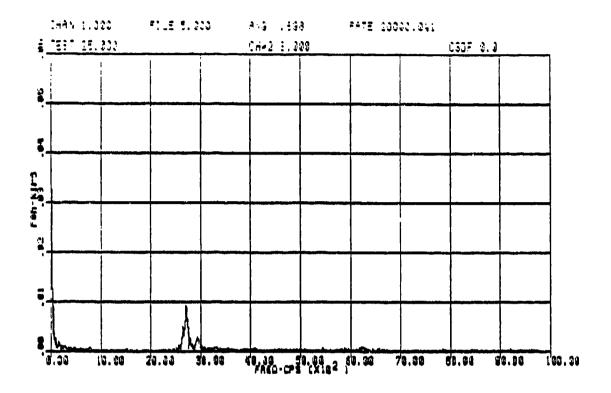




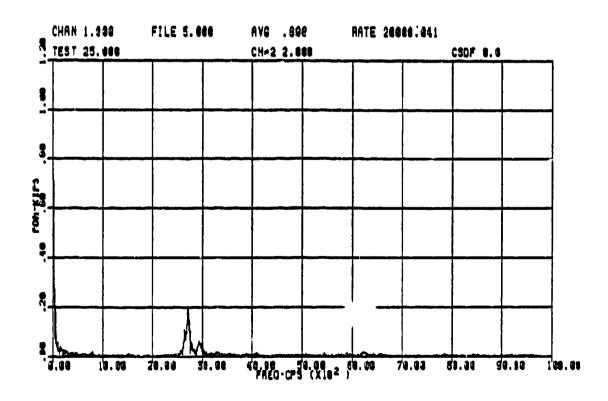


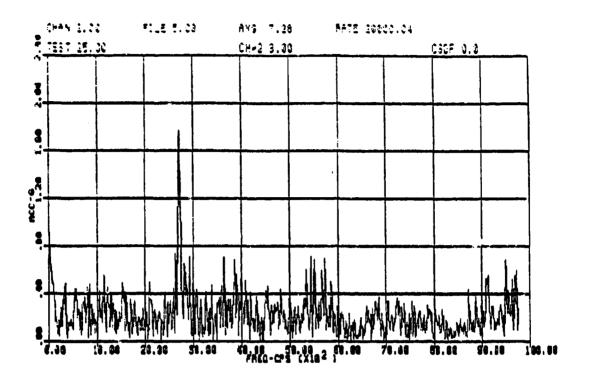


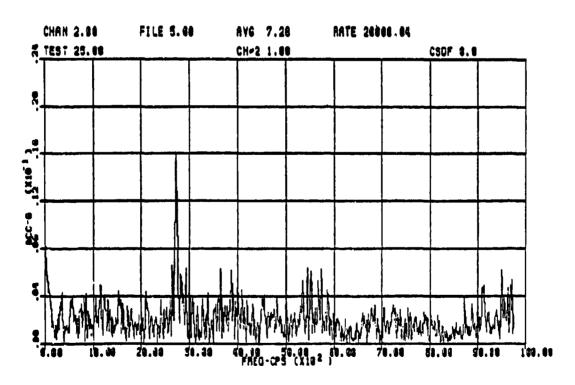




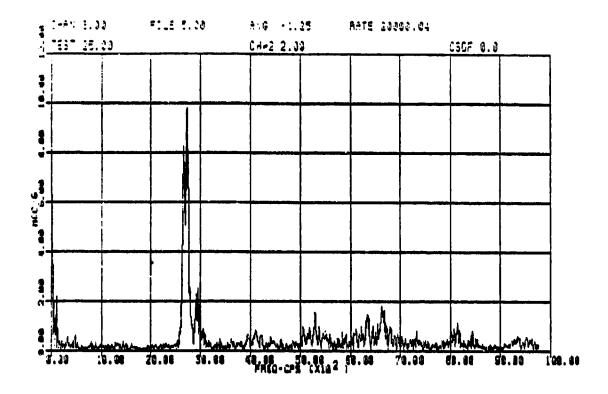
14.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 | 15.55.55 |

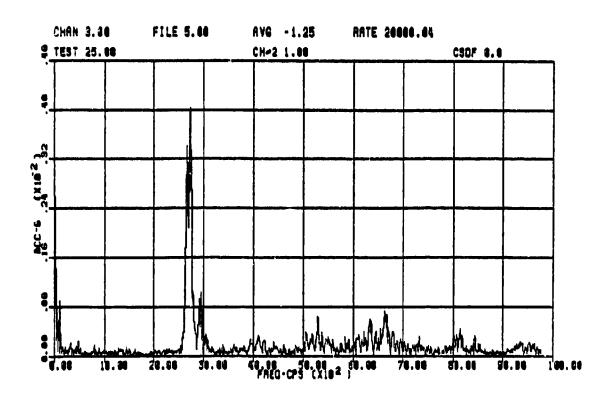


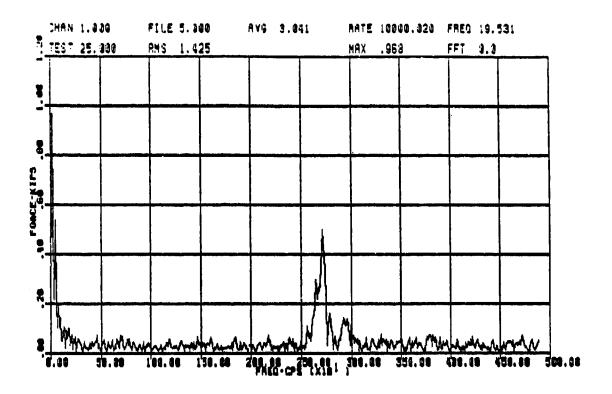


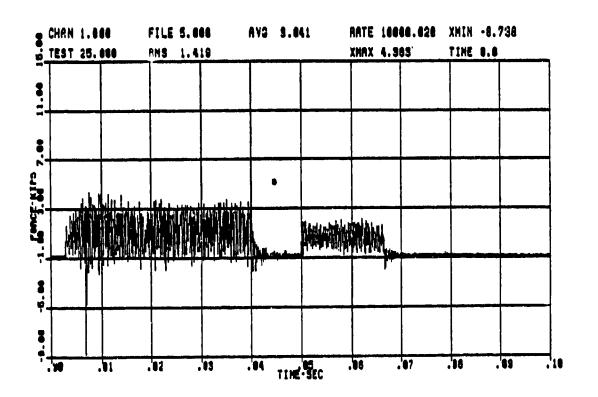


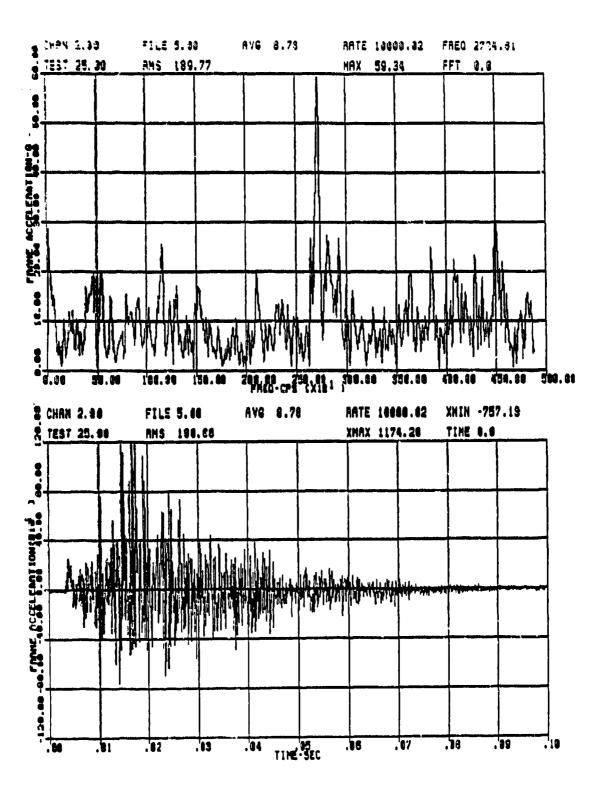
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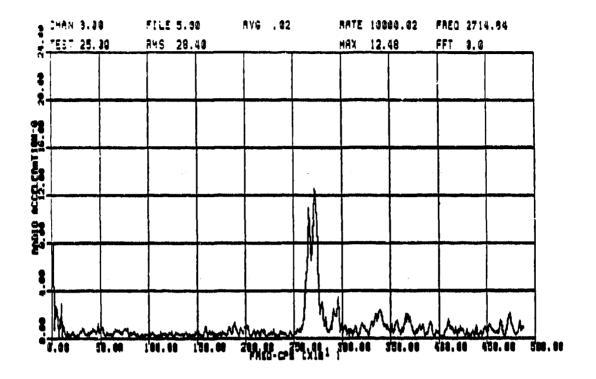


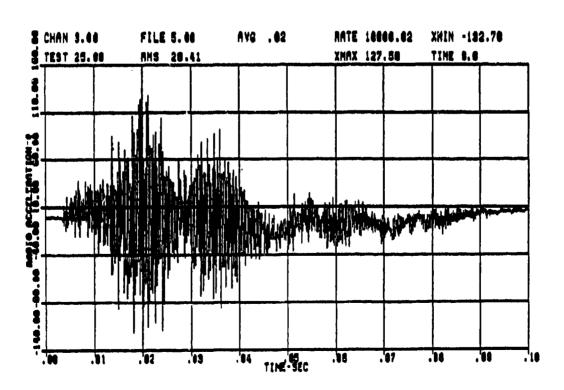


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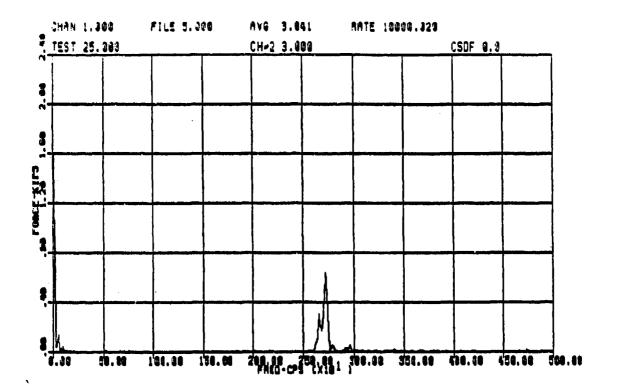
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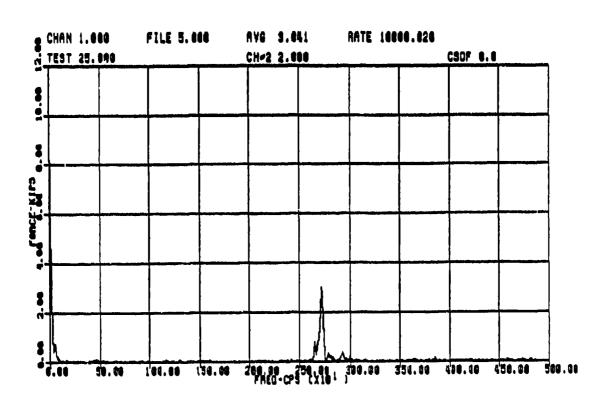
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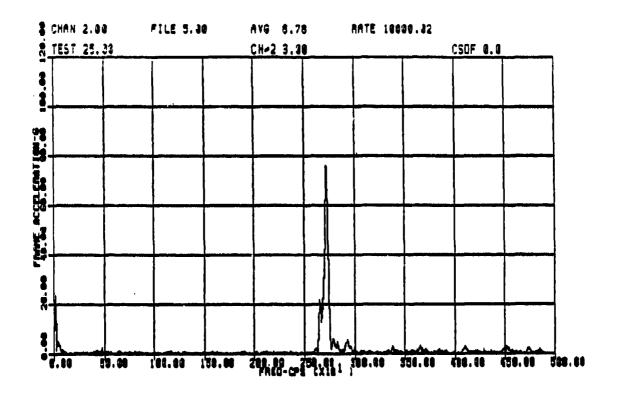


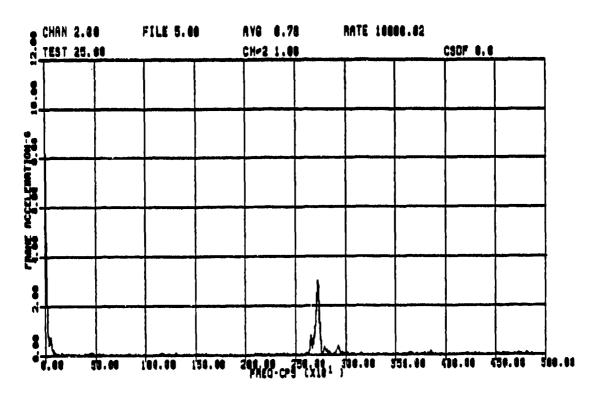


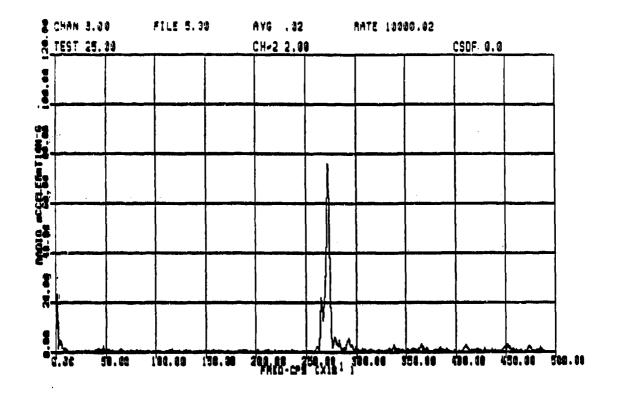
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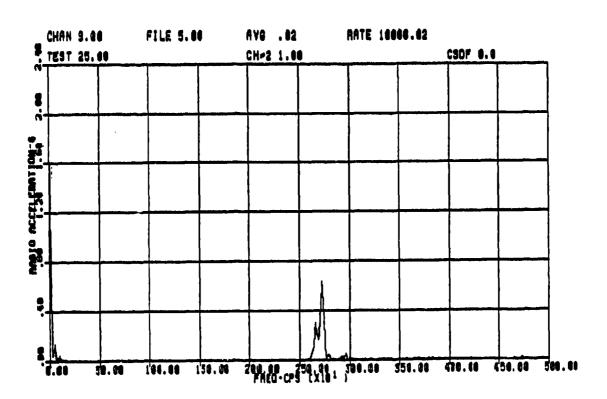


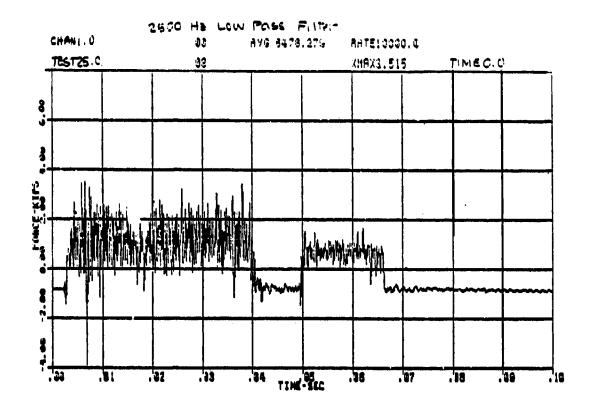


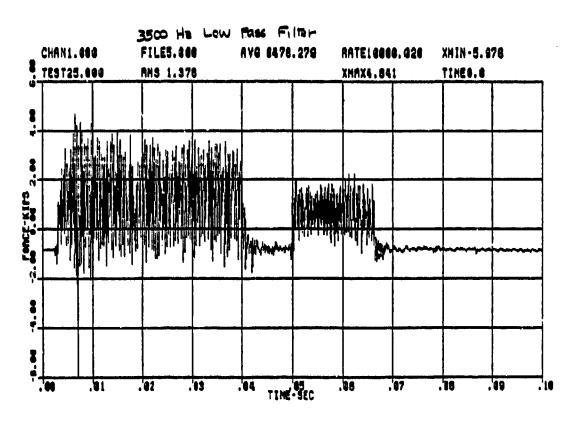


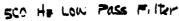


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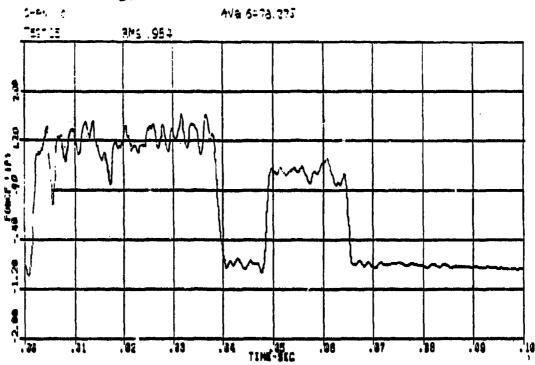


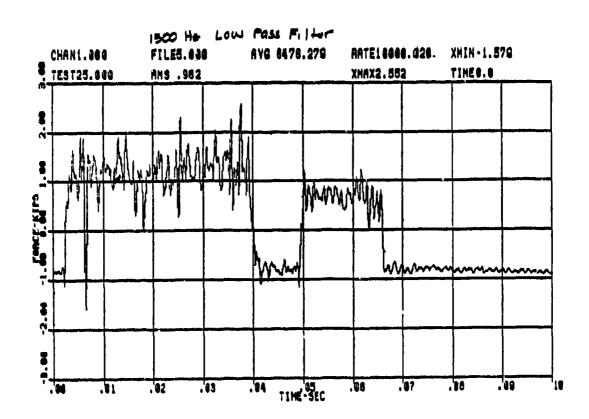


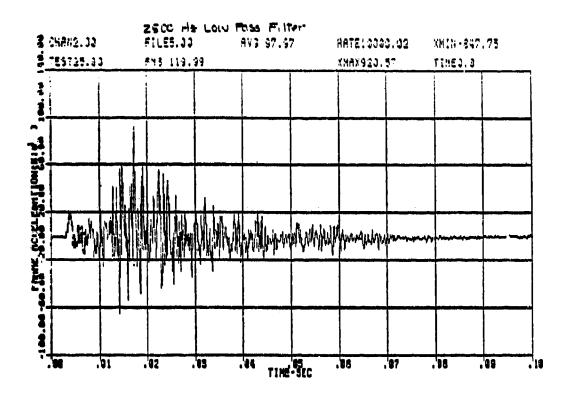


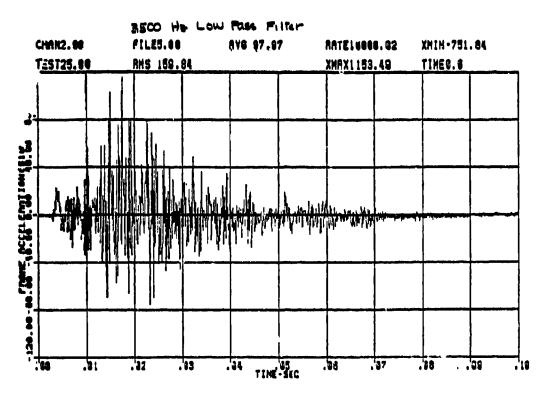


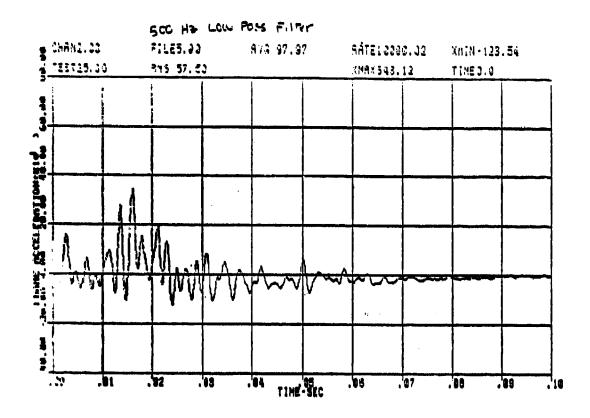
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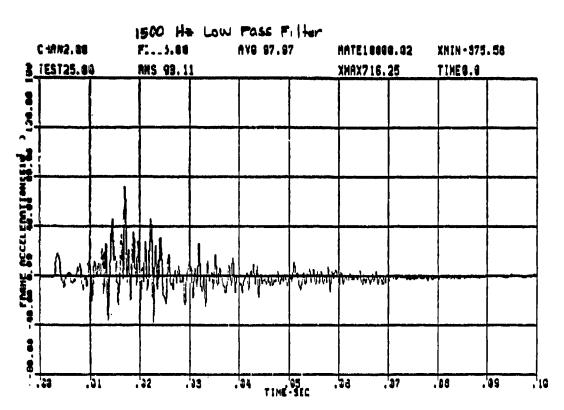




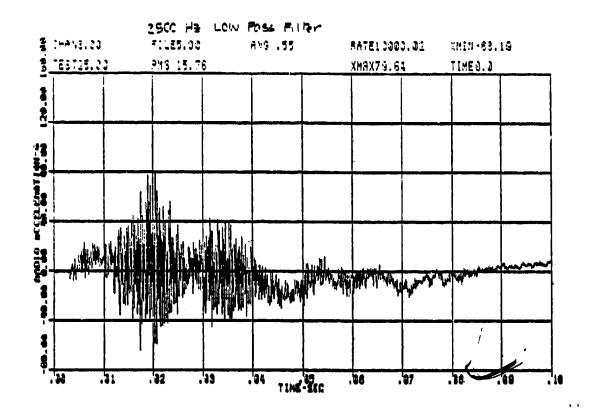








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Section 1

